A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIR GO. CALIFORNIA

SPEC NO.: ZD- 22-003	DATE:			
CUSTOMER: Delta Air Lines Inc.	MCL 10.31/4 DTD			
CHANGE NO: 200	MODEL: (880) 22-2			
Bilingual Signs, Instal	lation of			
Delta requested by Letter, F. REASON FOR CHANGE: Customer request	ile 142, dated 31 October 1958			
EFFECT ON WEIGHT	EFFECT ON BALANCE '			
GU.H. MI. EMPTY OPER. WT. EMPTY PROPOSAL "A" +6.0 Lb PROPOSAL "A" +6.1 Lb PROPOSAL "B" +9.0 lb PROPOSAL "B" +9.0 lb	U 15 PROPOSAL "A" +5,222 Inch-lo J 16 PROPOSAL "B" +7,717 INCH LB.			
EFFECT ON GUARANTEED PERFORMANCE: 4	e			
TOTALS REFLECTED IN A FUTURE CHANGE PROPO ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES	SAL.			
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-PECURRING: TOTAL:			
AGGERTED:	CONVAIR A DIVISION OF GENERAL DYNAMICS CORP.			

DATE:

CONVAIN: SD

Delta Air Lines Inc Change No. 200

Page 1 of 3

Title: Bilingual Signs, Installation of

Origin: Delta requested by Letter, File 142, dated 31 October 1958

Reason for Change: Customer request

Description of Change:

PROPOSAL "A"

Page 77, Paragraph 3.16.8.3.6 SIGNS:

Add the following to the end of the paragraph:

"The 'Fasten Seat Belt - No Smoking' and 'Occupied - Vacant' signs shall be bilingual, in Emglish and Spanish."

Page 107, Paragraph 3.19.2.2.2 OCCUPIED SIGHS:

Change the paragraph title and paragraph to the following:

"OCCUPIED-VACANT SIGNS: One bilingual (English and Spanish)
'Occupied-Vacant' sign shall be provided for each lavatory. These
signs shall be appropriately illuminated by lavatory door slide
bolt action. Indication of lavatory occupancy shall also be provided by the slide bolt over each door knob."

Page A-2, APPENDIX I-C, ELECTRICAL EQUIPMENT:

Delete the tenth and 12th items under "INTERIOR LIGHTS":

"(2 Fasten Seat Belts - No Smoking Sign and 3 Lavatory Occupied Sign)"

Page A-12, APPENDIX I-C, FURNISHINGS:

Change the 30th item in the Description List as follows:

From: "* Lavatory 'Occupied' Signs"

To: "3 Lavatory 'Occupied-Vacant' Signs (English and Spanish)"

CONVAIR: SD

Delta Air Lines Inc. Change No. 200

Page 2 of 3

1 40

Change the 31st item in the Description List as follows:

" No Smoking - Fasten Seat Belt Signs" From:

"4 "Fasten Seat Belts - No Smoking" Signs (English and Spanish)" To:

Effect on Weight Empty: +5.0 pounds Effect on Balance: +5,222 inch-pounds Effect on Performance: None

CONVAIA: SD

Delta Air Lines Inc. Change No. 200

Page 3 of 3

PROPOSAL "B" (Includes PROPOSAL "A")

Take at PROPOSAL "A" except includes the following:

Page 110, Paragraph 3.19.2.J.10 FLACARDS:

Change the first item in the placard list as follows:

From: "Emergency emits"

To: "mergency exit hatch operation (two, in English and Spanish)"

Add the following item to the placard list:

"Davatory signs (one for each lavatory, in English and Spanish)"

Page 115, Paragraph 3.19.0.3 EVACUATION PROVISIONS:

Add the fellowing sentence to the end of the paragraph:

"One bilingual, lighted 'ENIT' sign (English and Spanish) shall be installed at each main entrance door and service door; and one bilingual, lighted 'AMERGENCY EXIT' sign (English and Spanish) shall be installed at each of the two emergency exits."

Effect on Weight Empty: 49.9 pounds
Effect on Balance: 47,717 inch-pounds

Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-003		DATE:			
CUSTOMER: Delta Air Lines Inc.		MCL 10,313 DTD			
CHANGE NO: 199			MODEL: (880) 22-2 .		
TITLE Decal, (Clgare	otte Smoking Only) Bili	ngual, Installation of		
ORIGIN: Delta request	ed by Letter, File	e 142,	dated 31 October 1958		
REASON FOR CHANGE: Cu	stomer request				
			Rejectal		
EFFECT	N WEIGHT *		. EFFECT ON BALANCE *		
GUAR. WT. EMPTY	OPER. WT. EMPTY				
0	О		O INCH LB.		
EFFECT ON GUARANTEED P	ERFORMANCE: *				
	None				
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		SAL N			
			ENGINEERING APPROVAL		
LATEST DATE OF ACCEPTANCE:			ANES AFFECTED:		
SPECIAL, PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:			
		RECURRING:			
NON-KI			CURRING:		
CONVA			IR, A DIVISION OF GENERAL DYNAMICS CORP.		
BY:					
DATE:	THE STATE OF				
The state of the s	THE PERSON NAMED IN COLUMN TWO IS NOT THE PERSON NAMED IN COLUMN TWO IS NAMED IN COLUMN TW		7 May 10 W 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		

CONVAIR: SD

Title: Decal (Cigerette Smoking Only) Bilingual, Installation of

Origin: Delta requested by Letter, File 142, dated 31 October 1958.

Reason for Change: Customer request

Description of change:

Page 110, Paragraph 3.19.2.8.10 PLACARDS:

Add the following to the end of the paragraph:

"In addition to the above, a Buyer furnished bilingual decal (English and Spanish) shall be installed on each lower window frame in a position plainly visible to seat occupants."

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-003			DATE:	
CUSTOMER: Delta Air Lines Inc.			MCL 10,312 DTD	
CHANGE NO: 198			MODEL: (880) 22-2	
TITLE: Lavatory Placard, Installation of				
ORIGIN: Requested	d by Delta Letter.	File	142, dated 31 October 1958	
REASON FOR CHANGE:	Customer request.			
	ON WEIGHT .		EFFECT ON BALANCE *	
SUAR, WT. EMPTY	OPER. WT. EMPTY			
0 0			0 INCH LB.	
EFFECT ON GUARANTEED P				
	No	one		
" NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A				
ACCEPTANCE OF THIS CHA				
PRIOR ACCEPTANCE OF T	HE FOLLOWING CHANGES.			
			ENGINEERING APPROVAL	
LATEST DATE OF ACCEPTA	NCE:	AIRPLANES AFFECTED:		
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:		
		RECURRING: NON-RECURRING:		
TOTAL		TOTAL	·	
CONVA		IR, A DIVISION OF GENERAL DYNAMICS CORP.		
RV.				
BY:				
DATE			**************************************	

CONVAIR: SD

Delta Air Lines Inc. Change No. 198

Title: Lavatory Placard, Installation of

Origin: Requested by Delta Letter, File 142, dated 31 October 1958

Reason for Change: Customer request.

Description of Change:

Page 110, Paragraph 3.19.2.8.10 PLACARDS:

Add the following to the end of the placard list:

"Lavatory Sign (one forward and one aft, over Occupied-Vacant signs)"

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-003		DATE:
CUSTOMER: Delta Air Lines Inc.	1 - 1 12	MCL 10,265 DTD 23 May 1958
CHANGE NO: 197		MODEL: (880) 22-2
TITLE: Anti-Skid System, Revision	to	
ORIGIN: Delta requested by Letter,	File 142	2, dated 16 May 1958
REASON FOR CHANGE: Customer request		Rijected
EFFECT ON WEIGHT *		EFFECT ON BALANCE *
GUAR. WT. EMPTY Proposal "A" - 26.0 1b Proposal "A" - 51.0 1b Proposal "B" - 51.0 1b		
EFFECT ON GUARANTEED PERFORMANCE: *	one	
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED TOTALS REFLECTED IN A FUTURE CHANGE PRO ACCEPTANCE OF THIS CHANGE IS DEPENDENT PRIOR ACCEPTANCE OF THE FOLLOWING CHAN	OPOSAL UPON	
		ENGINEERING APPROVAL
LATEST DATE OF ACCEPTANCE:	AIRPLA	NES AFFECTED:
SPECIAL PROVISIONS:	RECURR NON-REC	ON PRICE PER AIRPLANE; ING:
ACCEPTED:	CONVAIR	R, A DIVISION OF GENERAL DYNAMICS CORP.
BY:	•	
DATE:		204V41745.D. 6-1217A

CONVAIR: SD

Delta Air Lines Inc. Change No. 197

Page 1 of 2

Title: Anti-Skid System, Revision to

Origin: Delta requested by Letter, File 142, dated 16 May 1958.

Reason for Change: Customer request

Description of Change:

PROPOSAL "A"

(Transistor automatic anti-skid system with skid indicator)

Page 34, Paragraph 3.8.2.2 WHEFLS, BRAKES AND BRAKE CONTROL SYSTEMS:

Add the following after the second sentence:

"The brake pedal shall be equipped with a Goodyear "foot thumper" unit to indicate wheel skidding to the pilot".

Page A-10, APPENDIX I-C, HYDRAULIC EQUIPMENT:

Change the below two items in the Description List as follows:

From: "1 Anti-Skid Control Valve (Dual) Hydro-Aire 38-103
4 Anti-Skid Control Valve (Dual) Hydro-Aire 38-099"

To: "9 Anti-Skid Control Valve Goodyear SE4-1653"

Effect on Weight Empty: -26.0 lbs

Effect on Balance: -25,469 Inch-lbs

Effect on Performance: None

CONVAIR: SD

Delta Air Lines Inc. Change No. 197

Page 2 of 2

PROPOSAL "B"

(Skid warning only (foot thumper)

Page 34, Paragraph 3.8.2.2 WHEELS, BRAKES AND BRAKE CONTROL SYSTEMS:

Add the following after the second sentence:

"The brake pedal shall be equipped with a Goodyear "foot thumper" unit to indicate wheel skidding to the pilot".

Page A-10, APPENDIX I-C, HYDRAULIC EQUIPMENT:

Delete the following two items from the Description List:

38-103 Hydro-Alre "1 Anti-Skid Control Valve 4 Anti-Skid Control Valve (Dual) Hydro-Aire 38-099"

Effect on Weight Empty: -51.0 lbs Effect on Balance: -46,663 Inch-1bs Effect on Balance:

Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-003			DATE:	
CUSTOMER: Delta Air Lines Inc.			MCL 10,311 DTD	
CHANGE NO: 196			MODEL: (880) 22-2	
TITLE: Specifica from insta	tion Administrativ	e Char ions)	nge (ATC Transponder, Change	
ORIGIN: Convair i	nitiated.			
REASON FOR CHANGE: T	o correct and clar	ify t	he intent of the Specification.	
EFFECT (ON WEIGHT *		EFFECT ON BALANCE *	
GUAR. WT. EMPTY	OPER. WT. EMPTY			
0	0		O INCH LB.	
EFFECT ON GUARANTEED F	PERFORMANCE: *	ne		
ACCEPTANCE OF THIS CH	LL BE ACCUMULATED AND FUTURE CHANGE PROPOSE ANGE IS DEPENDENT UPOF THE FOLLOWING CHANGES:	SAL N		
			ENGINEERING APPROVAL	
LATEST DATE OF ACCEPTA	ANCE:	AIRPL	ANES AFFECTED:	
RECU NON-F		RECUR NON-RE	FECT ON PRICE PER AIRPLANE: CURRING: N-RECURRING: TAL:	
ACCEPTED:		CONVA	NR, A DIVISION OF GENERAL DYNAMICS CORP.	
BY:				
DATE:			CONVARR, S.D. 6-1217A	

Delta Air Lines Inc. Change No. 196

CONVAIR: SD

Title: Specification Administrative Change (ATC Transponder, Change

from installation to provisions)

Origin: Convair initiated.

Reason for Change: To correct and clarify the intent of the Specification.

Description of Change:

Page 102, Paragraph 3.17.4.4 RADAR SAFETY BEACON:

Revise paragraph to read as follows:

"RADAR SAFETY BEACON (PROVISIONS): Provisions shall be made for the installation of a dual-ATC transponder beacon system."

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

The following shall not appear in the Specification language:

The subject paragraph was not revised in Delta CCP No. 49 which changed ATC transponder system from installation to provisions for a dual system. The intent of this proposal is to correct this oversight.

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-003			DATE:		
CUSTOMER: Delta Air Lines Inc.		MCL 10,309 DTD			
CHANGE NO: 194			MODEL: (880) 22-2		
TITLE: Muslin Sub-Co	overs, Cabin Passe	enger	Seats, Installation of		
ORIGIN: Joint Convai Monica on 22	er and National See July 1958.	ating	meeting with Delta at Santa		
REASON FOR CHANGE: Cus	stomer request.		Pagectal		
EFFECT	N WEIGHT *		EFFECT ON BALANCE '		
GUAR. WT. EMPTY /36.0 1b	OPER. WT. EMPTY		/31,536 INCH LB.		
EFFECT ON GUARANTEED P	ERFORMANCE: *				
None None Negligible Changes will be accumulated and totals reflected in a future change proposal acceptance of this change is dependent upon prior acceptance of the following changes:					
			ENGINEERING APPROVAL		
LATEST DATE OF ACCEPTAN	NCE:	AIRPL	ANES AFFECTED:		
NON-RECURRING:		T ON PRICE PER AIRPLANE: RING: CURRING: :			
GCEPTED:		CONVA	IR, A DIVISION OF GENERAL DYNAMICS CORP.		
DATE:			-0-1V 57-1,5 9-1417 A		

CONVAIR: SD

Delta Air Lines Inc. Change No. 194

Title: Muslin Sub-Covers, Cabin Passenger Seats, Installation of

Joint Convair and National Seating meeting with Delta at Origin:

Santa Monica on 22 July 1958

Reason for Change: Customer request.

Description of Change:

Page 103, Paragraph 3.19.1.1.1 SEATS:

Add the following sentence after the first sentence:

"A muslin sub-cover shall be installed in each cabin passenger seat between the cushioning material and the upholstering material".

Effect on Weight Empty: /36.0 1b

Effect on Balance:

/31,536 inch-1b

Effect on Performance:

None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003			DATE:		
CUSTOMER: Delta Air Lines Inc.			MCL 10,310 DTD		
CHANGE NO: 195 MODEL: (880) 22-2					
TITLE: Adjustabl	e Head Rest, Cabin	Pass	enger Seats, Installa	tion of	
ORIGIN: Convair i	nitiated.				
REASON FOR CHANGE:	To provide an opt cabin passenger s	ional seats.	l adjustable head rest	on	
EFFECT	ON WEIGHT *		EFFECT ON BALAN	ICE *	
GUAR. WT. EMPTY ≠162.0 1b	OPER. WT. EMPTY		≠141,91 2	INCH LB.	
EFFECT ON GUARANTEED F	PERFORMANCE: * None	9			
ACCEPTANCE OF THIS CH	L BE ACCUMULATED AND FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPON THE FOLLOWING CHANGES:	٧	ENGINEERING	APPROVAL	
LATEST DATE OF ACCEPTANCE:		AIRPL	ANES AFFECTED:		
RECU NON-F		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED: BY:		CONVA	AIR, A DIVISION OF GENERAL DYNAM	MICS CORP.	
DATE				PUZALICA CE GADICA	

Delta Air Lines Inc. Change No. 195

CONVAIR: SD

Title: Adjustable Head Rest, Cabin Passenger Seats, Installation of

Origin: Convair initiated.

Reason for Change: To provide an optional adjustable head rest on

cabin passenger seats.

Description of Change:

Page 104, Paragraph 3.19.1.1.6 PASSENGER SEATS:

Add the following sentence to the end of the paragraph:

"An adjustable head rest shall be installed on each seat in the main passenger compartment."

Effect on Weight Empty: /162.0 pounds

Effect on Balance: /141,912 inch-pounds

Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003	DATE:		
CUSTOMER: Delta Air Lines Inc.	MCL 10,308 DTD		
CHANGE NO: 193	MODEL: (880) 22-2		
TITLE: Crash Energy Absorber, Stand Installation of	lard Cabin Passenger Seats,		
ORIGIN: TWA requested and Convair pr	roposed for Delta		
REASON FOR CHANGE: To provide addition	nal crash protection		
EFFECT ON WEIGHT *	EFFECT ON BALANCE *		
GUAR, WT. EMPTY OPER, WT. EMPTY			
+44.0 lb +44.0 lb	+38,544 INCH LB.		
EFFECT ON GUARANTEED PERFORMANCE: *			
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSED ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPOPRIOR ACCEPTANCE OF THE FOLLOWING CHANGES	SAL DN		
	ENGINEERING APPROVAL		
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:		
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:		
motion at this material and all manufacturing, reproduction, use and all controls portioning to such about rather are controls a control of controls that are section to control this material areas that this material will not be used, copied or improduced to while or in	NON-RECURRING:		
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.		
BY:			

Delta Air Lines Inc. Change No. 193

CONVAIR: SD

Title: Crash Energy Absorber, Standard Cabin Passenger Seats, Installation of

Origin: TWA requested and Convair proposed for Delta

Reason for Change: To provide additional crash protection

Description of Change:

Page 105, Add the following new paragraph to the page, after paragraph 3.19.1.1.7:

"3.19.1.1.7.1 CRASH ENERGY ADSORBER: All cabin passenger seats shall be equipped with crash energy absorbers installed at each safety belt attachment point".

Page A-12, APPENDIX I-C, FURNISHINGS:

Revise the 12th item in the Description List as follows:

From: "84 Passenger Safety Belts"

To: "72 Passenger Safety Belts, and Crash Energy Absorbers 12 Passenger Safety Belts (Club Area)"

Effect on Weight Empty: +44.0 lb Effect on Balance: +38,544 inch-lb Effect on Performance: None

> PROTEINTAIN RIGHTS of Conver one increase in the subjectment of the material and all increase up, reproduction, use god soles notice posture of the succession to the respect of an expect of an expect of a subject in the angle of the succession of the succession

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003			DATE:		
CUSTOMER: Delta Air Lines Inc.			MCL 10,307 DTD		
CHANGE NO: 192			MODEL: (880) 22-2		
TITLE: Two-Position	Foot Rest, Capin I	issenger (Seats, Installati	on of	
ORIGIN: Joint Convai	r and National Sea July 1958	ting meet:	ing with Delta at	Santa	
REASON FOR CHANGE: Cu	stomer request.		Rejected		
EFFECT	ON WEIGHT *		EFFECT ON BALAN	ICE *	
GUAR, WT. EMPTY	OPER. WT. EMPTY #82.0 1b		<i>4</i> 70,27 ⁴	INCH LB.	
			710941	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	
EFFECT ON GUARANTEED F					
None Negligible Changes will be accumulated and totals reflected in a future change proposal Acceptance of this change is dependent upon prior acceptance of the following changes:					
			ENGINEERING	APPROVAL	
LATEST DATE OF ACCEPTA	NCE:	AIRPLANES A	FFECTED:		
RECUI NON-R			FFECT ON PRICE PER AIRPLANE: ECURRING: ON-RECURRING: OTAL:		
ACCEPTED: BY:		CONVAIR, A DI	VISION OF GENERAL DYN AM		
DATE:	Marie Control of the			H V 41R. 5. 3. 6-1217 A	

Delta Air Lines Inc. Change No. 192

Title: Two-Position Foot Rest, Cabin Passenger Seats, Installation of

Origin: Joint Convair and National Seating meeting with Delta at Santa

Monica on 22 July 1958

Reason for Change: Customer request.

Description of Change:

Page 104, Paragraph 3.19.1.1.6 PASSENGER SEATS:

Add the following sentence to the end of the paragraph:

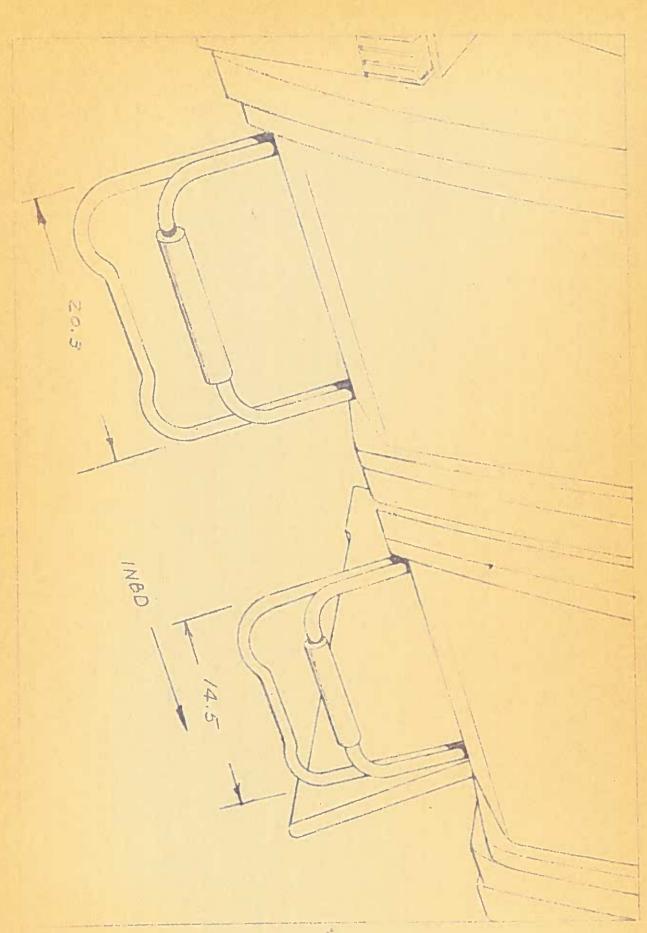
"A two-position foot rest, attached to the rear beam of each cabin passenger seat structure, except last seat row, shall be installed for use of each main cabin passenger except those occupying the front row seats and those immediately aft of the movable coat dividers."

Enclosure: (A) One copy of sketch "CONVAIR 880 FOOT REST" (for information only)

Effect on Weight Empty: 182.0 1b

Effect on Balance: #70.274 inch-lb

Effect on Performance: None



CONVAIR 880 - FOOT REST

A DIVISION OF STRENAL DYNAMICS CORPORATION SAN DIR GO, CALIFORNIA

SPEC NO.: ZD- 22-003		DATE:			
CUSTOMER: Delta Air	clines, Inc.	MCL		_DTD	
CHANGE NO: 19I		MODEL:	22-2	(Convair	"880")
TITLE Manufacti	arer"s Name and/or	Part Number			
	approved specific equipment.	designations	for i	tems of C	onvair
REASON FOR CHANGE: TO	o identify equipme	ent items lis	ted in	Appendix	I-C.
EFFECT	ON WEIGHT .		EFFEC	T ON BALANC	EE •
GUAR, WT. EMPTY	OPER, WT, EMPTY	~			
0	0 , ,			0	INCH L.B.
EFFECT ON GUARANTEED F	ERFORMANCE: ' None				
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF THE PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPO ANGE IS DEPENDENT UPO	SAL	hope	NGINEERING A	APPROVAL
LATEST DATE OF ACCEPTA	NCE:	AIRPLANES AFFI			
SPECIAL PROVISIONS:		EFFECT ON PRICE RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED		CONVAIR, A DIVISI	ON OF GE	NERAL DYNAMI	CS CORP.
BY.	<u> </u>				
EATE:					

Page 1 of 12

Title: Manufacturer's Name and/or Part Number

Origin: Customer approved specific designations for items of Convair

furnished equipment.

Reason for Change: To identify equipment items listed in Appendix I-C.

Description of Change:

APPENDIX I-C: Change or add manufacturer's name and/or part number to

items so as to read as follows:

Page No.	Quan Reqd	Description	Manufacturer	Part or Spec. Number
A-1	POWER	PLANT EQUIPMENT		
From:	4 4 2 4	Tachometer Generators (Furnished with engine) Starter Shutoff Valves Ice Warning Detectors Pressure Ratio Transmitter	AiResearch Goodyear (CARL) Kollsman	105492-325-1 Type T260-MK-120 033041-50-433
To:	#4 ##4 **4 #2	Tachometer Generators (Furnished with engines) Starter Shutoff Valves Starter Shutoff Valves Pressure Ratio Transmitter Ice Warning Detectors Ice Warning Detectors	Gen. Electric AiResearch AiResearch Kollsman Goodyear (CARL)	2CM9AAH4 105492-325-1 105492-4 C33041-50-433 123-00142 Type T260-MK12A) 123-00142 Type T260-MK19A)

Change the **item at bottom of page to read as follows:

**Effective Ships 8 and on.

Add the following to bottom of page:

#Effective Ships 1 through 10. ##Effective Ships 11 and on.

A-la FUEL SYSTEM EQUIPMENT

From: 4 Cap - Pressure Fuel Servic- Parker ing (Flush-Type) Aircraft 1327-575806

Page 2 of 12

Page No.	Quan Reqd	Description	Manufacturer	Part or Spec. Number	
A-la	FUEL	SYSTEM EQUIPMENT (Cont)			
To:	*4	Cap, Pressure Fuel Servicing (Flush Type) Cap, Pressure Fuel Servicing (Flush Type)	Parker Aircraft Parker Aircraft	1327-575806 1327-575806M1	
	OIL S	YSTEM EQUIPMENT			
From:	4	Engine Oil Temperature Indicator	Lewis Eng.	162C23A	
To:	#4 ##4	Engine Oil Temperature Indicator Engine Oil Temperature	Lewis Eng.	162C23A	
		Indicator	Lewis Eng.	162C23B	
Add t	he fol	lowing to bottom of page:			
*Effective Ships 1 through 13. **Effective Ships 14 and on. #Effective Ships 1 through 10. ##Effective Ships 11 and on.					
A-2	ELECT	RICAL EQUIPMENT			
From:	2 1 29	Light, Retractable Landing Light, Dome Wheel Well Cockpit Lights	Grimes Grimes Grimes	40285A-4616 31595-23D 30750 30780	
<u>To</u> :	*2 **2 1 29	Light, Retractable Landing Light, Retractable Landing Light, Dome Wheel Well Cockpit Lights	Grimes Grimes Grimes Grimes Grimes	40285A-4616 40295A-4559 31595A-23D 30750-1 30780-2	

Add the following to bottom of page:

^{*}Effective Ships 1 through 13.
**Effective Ships 14 and on.

Page 3 of 12

Page No.	Quan Reqd	Description	Manufacturer	Part or Spec. Number
A-3	ELECT	RICAL EQUIPMENT		
From:	4 1 3 1 4 1	Ignition Switches External Power Contractor Shaver Cutlet Receptacles Utility Receptacle Line and Bus Tie Contactor Warning Horn Repeat Cycle Timer	Micro Gen. Electric ABC Elec. Hubbell Gen. Electric Std.Elec.Prod.	V3-1 729C574P2 1097 7332 729C572
	4	Constant Speed Drive (Including Disconnect) Constant Speed Drive Load	Gen. Electric	2CLKH4OB1
	1	Controller Wheel Control Motor	Gen. Electric	7TAR10A01
		Stabilizer Trim	Sperry	2582028-2
To:	*4 #1 ##1 31 #4 ##4	Ignition Switches External Power Contactor External Power Contactor Shaver Outlet Receptacles Utility Receptacle Line and Bus Tie Contactor Line and Bus Tie Contactor Warning Horn Repeat Cycle	Micro Gen. Electric Hartman ADC Elec. Hubbell Gen. Electric Hartman	AN3234-1 729C574P2 A874KF 1097-3 10108 729C572 B124A
	*4	Timer Constant Speed Drive	Haydon	A31379
	**1	(Including Disconnect) Constant Speed Drive	Gen. Electric	2CLKH4OB1
	***4	(Including Disconnect) Constant Speed Drive	Gen. Electric	2CLKH40B5
	4	(Including Disconnect) Constant Speed Drive Load Controller Wheel Control Motor	Gen. Electric Gen. Electric Honeywell	2CLKH40B6 7TAR10A01 or 31357E
	1.	Stabilizer Trim	Sperry	2582028-02

Add the following to bottom of page:

*Effective Ships 1 through 10 only.

**Effective Ships 11 through 13.

***Effective Ships 14 and on.

#Effective Ships 1 through 13.

##Effective Ships 14 and on.

A-4 ELECTRONIC EQUIPMENT

From: 2 VOR/LOC Instrumentation Unit Collins

Page 4 of 12

Page No.	Quan Reqd	Description	Manufacturer	Part or Spec. Number
A-4	ELECT	RONIC EQUIPMENT (Cont)		
	*2 2	Glide Slope Receivers (a-c Power Source) Antenna (Low Drag)	Collins Microwave	51V-3 Type 522-0671-044 22-31006-1
To:	2	VOR/LOC Instrumentation	Collins	344B-1 Type, 522-0887-055
	ø*2	Glide Slope Receivers (a-c Power Source)	Collins	51V-3 Type, 522-0671-044
3	dø*2	Glide Slope Receivers (a-c Power Source)	Collins	51V-3 Type, 522-0671-314
	2	Antenna (Low Drag)	Microwave	22-31006-3
Add ti	he fol	lowing to bottom of page:		
*Indicates miniaturized equipment. øEffective Ships 1 through 13. øØEffective Ships 14 and on.				
A-5	ELECT	RONIC EQUIPMENT		
From:	2	Antenna	Convair	LB-147
To:	2	Antenna	Electronic Specialty	LB-147
A-7	INSTR	UMENTS AND RELATED EQUIPMENT		
From:	1.	Fuel Quantity Totalizer Indicator Fuel Quantity Ind. (Counter-	Simmonds Aerocess.	393012-01642
		Pointer-Type) Tanks #1 and #4	Simmonds Aerocess.	393022-01581
	2	Fuel Quantity Ind. (Counter- Pointer-Type) Tanks #2 and #3	Simmonds Aerocess.	393022-01582
	2 2 4	Flap Position Indicator - Dual Flap Position Transmitter Pressure Indicator(Hydraulic) Tachometer Indicator	Gen. Electric Gen. Electric U. S. Gauge Gen. Electric	8DJ91LAB-Y1 8TJ39AAH-1 SRL-07J 8DJ81-LAW-1

Page 5 of 12

Page No.	Quan Reqd	Description	Manufacturer	Part or Spec. Number
A-7	INSTR	UMENTS AND RELATED EQUIPMENT	(Cont)	
To:	1 2	Fuel Quantity Totalizer Indicator Fuel Quantity Ind. (Counter-	Simmonds Aerocess.	393012-08642
		Pointer-Type) Tanks #1 and #4	Simmonds Aerocess.	393022-05581
	2	Fuel Quantity Ind. (Counter-Pointer-Type) Tanks #2 and #3	Simmonds	
	*1	Flap Position Indicator -	Aerocess.	393022-05582
	**1	Dual Flap Position Indicator -	Gen. Electric	8DJ91LAB-1
	\$2 \$2 \$2 \$2 *4 *4	Dual Flap Position Transmitter Flap Position Transmitter Pressure Indicator (Hydraulic) Pressure Indicator (Hydraulic) Tachometer Indicator Tachometer Indicator	Gen. Electric Gen. Electric U. S. Gauge U. S. Gauge Gen. Electric Gen. Electric	8DJ91LAE-1 8TJ39AAH-1 8TJ39AAH-2 SRL-07J SRL-07K 8DJ81LAW-1 8DJ81LAW-2

Add the following to bottom of page:

*Effective Ships 1 through 10.
**Effective Ships 11 and on.

©Effective Ships 1 through 13.

©Effective Ships 14 and on.

A-7a INSTRUMENTS AND RELATED EQUIPMENT

From:	1 4	Warning Bell Indicator, Engine Pressure	Edward	168-2c
	2	Ratio Indicator, Cabin Compressor	Kollsman	B29187-10-001
		Bearing Temperature	Ham. Standard	527732
To:	1 ø4	Warning Bell Indicator, Engine Pressure	Sperti-Farady	80364
	øø4	Ratio Indicator, Engine Pressure	Kollsman	B29187-10-001
	#2	Ratio Indicator, Cabin Compressor	Kollsman	C29187-10-001
	##2	Bearing Temperature Indicator, Cabin Compressor Bearing Temperature	Ham. Standard Ham. Standard Fenwal	527732 562623 or 80062-5

Page 6 of 12

	Regd Description	Manufacturer	Part or Spec. Number
A-7a	INSTRUMENTS AND RELATED EQUIPMENT	(Cont)	
Add t	he following to bottom of page:		
	øEffective Ships 1 through 13. øØEffective Ships 14 and on. #Effective Ships 1 through 10. ##Effective Ships 11 and on.		
A-7a-	INSTRUMENTS AND RELATED EQUIPMEN	T	
From:	2 Indicator, Turbine Tachomete Cabin Compressor (RPM Comp Ind.)		535461
To:	ø2 Indicator, Turbine Tachom- eter Cabin Compressor (RPM Comp. Ind.) øø2 Indicator, Turbine Tachom-	Ham.Standard	535461
	eter Cabin Compressor (RPM Comp. Ind.)	Ham. Standard	562208
Add t	he following to bottom of page:		
	øEffective Ships 1 through 10. øØEffective Ships 11 and on.		
A-8	INSTRUMENTS AND RELATED EQUIPMENT		
From:	1 Servo Amplifier	Sperry	618379
To:	*1 Servo Amplifier **1 Servo Amplifier	Sperry Sperry	618379 1783867-1
Add t	he following to bottom of page:		
	*Effective Ships 1 through 13. **Effective Ships 14 and on.		
A-9	INSTRUMENTS AND RELATED EQUIPMENT		
From:	3 Servo Amplifier 1 Indicator, Master True Air- speed	Sperry Kollsman	618379 A29277-10-001 or B29277-10-001

Page 7 of 12

Page No.	Quan Reqd Description	Manufacturer	Part or Spec. Number
A-9	INSTRUMENTS AND RELATED EQUIPMENT	(Cont)	
To:	\$\omega_3\$ Servo Amplifier \$\omega_3\$ Servo Amplifier #1 Indicator, Master True Air-	Sperry Sperry	618379 1783867-1
	speed ##1 Indicator, Master True Air-	Kollsman	A29277-10-001
	speed speed	Kollsman	B29277-10-001

Change the following at bottom of page:

From: ***Applicable to Airplanes 10 through 13.

To: ***Applicable to Airplanes 10 and on.

Add the following to bottom of page:

øEffective Ships 1 through 13.
øØEffective Ships 14 and on.
#Effective Ships 1 through 10.
##Effective Ships 11 and on.

A-10 HYDRAULIC EQUIPMENT

From:	1 2 1 1	Nose Gear Steering Cylinder and Valve Assembly Gage, Emergency Air Brake Flap Actuating Motors NLG Selector Valve Flap Selector Valve	Cleveland Pneumatic U. S. Gauge Vickers Peacock Bertea	9818-A AW1826ADO1 MF016B007A 51200-3 38400-303
	*1 **1	Nose Gear Steering Cylinder and Valve Assembly Nose Gear Steering Cylinder and Valve Assembly Gage, Emergency Air Brake Gage, Emergency Air Brake Flap Actuating Motors Flap Actuating Motors NLG Selector Valve NLG Selector Valve Flap Selector Valve Flap Selector Valve	Cleveland Pneumatic Cleveland Pneumatic U. S. Gauge U. S. Gauge Vickers Vickers Peacock Peacock Bertea Bertea	9818-A 9818-B AW1826AD01 AW1826AD02 MF-016B007A MF-016B012A 51200-3 51200-5 38400-303 38400-5003

Page 8 of 12

Page No.	Quan Reqd	Description	Manufacturer	Part or Spec. Number
A-10	HYDRAU	ULIC EQUIPMENT (Cont)		
Add t	he foll	lowing to bottom of page:		
	**Effe	ective Ships 1 through 13. ective Ships 14 and on. ective Ships 1 through 10. ective Ships 11 and on.		
A-10a	HYDRAU	ULIC EQUIPMENT		
From:	1 4 2	Anti-Skid Control Valve Anti-Skid Control Valve(Dual) Hyd. Booster Pump and Motor	Hydro-Aire Hydro-Aire Vickers	38-103A 38-099A AD-155-6-C
<u>To</u> :	1 4 2	Anti-Skid Control Valve, NLG Anti-Skid Control Valve, MLG Hyd. Booster Pump and Motor	Hydro-Aire Hydro-Aire Vickers	38-289 38-299 AS-15506-C
A-12	FURNI	SHINGS		
From:	1 1 4 1 3 84	Pilot Seat Copilot Seat Lounge Area Seats (Double) Lounge Area Seat (Double) Lounge Area Seat (Double) Stewardess Safety Belts and Shoulder Harnesses Passenger Seat Belts (with	Convair Convair Convair Convair Convair	22-98504-7 22-98504-8 22-93597-805 22-93552-1 22-93598-1 BN-1-1510-2
		Cummins and Saunders 3100A Gold Anodized Buckle)	Convair	BN-1-1510-1
<u>To</u> :	*1 **1 ***1 **1 ***1 ***1 ***1 **1 **1	Pilot Seat Pilot Seat Pilot Seat Copilot Seat Copilot Seat Copilot Seat Lounge Area Seats (Double) Lounge Area Seat (Double) Lounge Area Seat (Double) Lounge Area Seat (Double) Lounge Area Seat (Double)	Convair	22-98504-7 22-98504-805 30-98504-805 22-98504-806 30-98504-803 22-93597-805 22-93597-811 22-93552-1 22-93552-5

Page 9 of 12

Page Quan No. Reqd Description	Manufacturer	Part or Spec. Number		
A-12 FURNISHINGS (Cont)				
To: *1 Lounge Area Seat (Double Cont **1 Lounge Area Seat (Double t**1 Lounge Area Seat (Double Lounge Area Seat (Double Stewardess Safety Belts	c) Convair c) Convair	22-93598-1 22-93598-5 22-93598-807		
Shoulder Harnesses ##3 Stewardess Safety Belts Shoulder Harnesses	Autocrat and Cummins and Saunders	BN1-1510-2 CV1013-2-and -3		
#84 Passenger Seat Belts (wi Cummins-Saunders 3100A Gold Anodized Buckle) ##84 Passenger Seat Belts (wi Gold Anodized Buckle)	Autocrat	BN1-1510-1 CV1013-1		
Delete the following item at bottom	of page:			
1 Toilet Tissue Dispenser	Convair	22-93803 (Fwd Lav)		
Add the following to the bottom of p	page:			
*Effective Ships 1 through 10. **Effective Ships 11 through 13. ***Effective Ships 14 and on. pEffective Ships 1 through 10. pEffective Ships 11 and on. #Effective Ships 1 through 13. ##Effective Ships 14 and on.				
A-12a FURNISHINGS				
From: 2 Soap Dispensers - Cake 1 Soap Dispenser - Cake	Convair Convair	22-94806-3 22-94806-4		
To: 1 Soap Dispenser - Cake	Convair	22-94806-3 Aft Lav. L/H		
l Soap Dispenser - Cake	Convair	22-94806-4 At Lav. R/H		
A-13 FURNISHINGS				
From: 4 Buffet, No. 1 No. 2 No. 3 No. 4	Convair Convair Convair Convair	22-93002 22-93005-3 22-93005-1 22-93009		

Page 10 of 12

Page No.	Quan Reqd Descrip	otion	Manufacturer	Part or Spec. Number
A-13	FURNISHINGS	(Cont)		
<u>To:</u>	4 Buffet	, pno. 1 #no. 2 ##no. 3 #no. 3 #no. 4 \$no. 4 \$\$no. 4	Convair Convair Convair Convair Convair Convair Convair Convair	22-93002-1 22-93005-3 22-93005-801 22-93005-1 22-93005-5 22-93009 22-93009-1 22-93009-3

Change the following at the bottom of page:

From: *Effective Ships 1 through 13.
**Effective Ships 14 and on.

To:

*Effective Ships 1 through 10.

**Effective Ships 11 and on.

©Effective Ships 1 through 13.

#Effective Ships 1 through 10.

##Effective Ships 11 through 13.

\$Effective Ships 2 through 7.

\$\$Effective Ships 1, and 8 through 10.

\$\$\$Effective Ships 11 through 13.

A-13a FURNISHINGS

From:	3	Ash Trays, Lavatory Cabin Dispenser Aft Lav. Consisting of:	Adams Rite Convair	2140-3-1 94803-1
	1	Cabin Dispenser Aft Lav. Consisting of:	Convair	94804-1
To:	*3 **3	Ash Trays, Lavatory Ash Trays, Lavatory	Adams Rite Benbow	2140-3-1 10311-6
	1	Cabinet Dispenser Aft Lav. L/H, Consisting of:	Convair	94805-3
	1	Cabinet Dispenser Aft Lav. R/H, Consisting of	Convair	94805-4

Add the following to bottom of page:

*Effective Ships 1 through 10.

**Effective Ships 11 and on.

Page 11 of 12

Page No.	Quan Reqd	Description	Manufacturer	Part or Spec. Number
A-14	PRESS	URIZATION, ANTI-ICING AND AIR	CONDITIONING EQU	JIPMENT
From:	2 2 2	ATM Driven Cabin Compressors Freon Compressors Condenser Cooling Air	Ham. Standard Ham. Standard	550150 560333
		Shutoff Valves	Ham. Standard	548333
	6	Anti-Icing Pressure Regulators	AiResearch	108594-130-2
	1	Recirculating Blower (Cabin Air)	Ham. Standard	545751-C
To:	2 2	ATM Driven Cabin Compressors Freon Compressors Condenser Cooling Air	Ham. Standard	55 8650 561254
		Shutoff Valves	Ham. Standard	569952
	ø6	Anti-Icing Pressure Regulators	AiResearch	108594-130-2
	øø6	Anti-Icing Pressure Regulators	AiResearch	108594-130-3
	\$1	Recirculation Blower (Cabin Air)	Ham. Standard	545751-C
	\$\$1	Recirculation Blower		
	\$\$\$1	(Cabin Air) Recirculation Blower	Ham. Standard	568715
		(Cabin Air)	Ham. Standard	573972
Add t	he fol	lowing to bottom of page:		
øEffective Ships 1 through 13. øÆEffective Ships 14 and on. \$Effective Ships 1 through 10. \$\$Effective Ships 11 through 13. \$\$\$Effective Ships 14 and on.				
A-15	OXYGE	EN EQUIPMENT		

From:	1	Automatic Passenger Valve with Manual Override Line Valve	Alar Robbins	5870-100 0V-601
To:	2	Automatic Passenger Valve with Manual Override Line Valve	Cons. Cont. Co. Robbins	2620A1-1 0V-601-1P

Effect on Weight Empty: O
Effect on Balance: O
Effect on Performance: None

Page 12 of 12

Page No.	Quan Reqd	Descripti	on	Manufacturer	Part or Spec. Number		
A-11	LANDING GEAR EQUIPMENT						
From:	2	Main Oleo	Shock Strut	Cleveland Pneu	9729A - L/R or 9729C - L/R		
	1	Nose Oleo	Shock Strut	Cleveland Pneu	9772A		
To:	885 85		Shock Strut Shock Strut	Cleveland Pneu Cleveland Pneu	9729A - L/R 9729C - L/R or 9729C-00 - L/R		
	1	Nose Olec	Shock Strut	Cleveland Pneu	9772A or 9772-00A		

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003			DATE:				
CUSTOMER: Delta Air	Lines Inc.	MCL 10,304 DTD					
CHANGE NO: 190		MODEL: (880) 22-2					
TITLE: Specification Administrative Change (Change "tail cone" from interchangeable to replaceable)							
ORIGIN: Convair initated. REASON FOR CHANGE: To clarify and correct the intent of the Specification.							
EFFECT	ON WEIGHT *	-	EFFECT ON BALANCE *				
UAR, WT. EMPTY	OPER, WT. EMPTY						
0	0		O INCH LB.				
EFFECT ON GUARANTEED PERFORMANCE: * None							
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES: ENGINEERING APPROVAL							
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:				
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:					
BY:DATE:		CONVA	AR, A DIVISION OF GENERAL DYNAMICS CORP.				

Delta Air Lines Inc. Change No. 190

CONVAIR: SD

Title: Specification Administrative Change (Change "tail cone"

from interchangeable to replaceable)

Origin: Convair initiated.

Reason for Change: To clarify and correct the intent of the Specifi-

cation.

Description of Change:

Page 125, Paragraph 3.23.1 EQUIPMENT INTERCHANGEABILITY:

Delete "tail cone" from "1. Interchangeable Parts"

Page 126, Paragraph 3.23.1 EQUIPMENT INTERCHANGEABILITY:

Add the following item under "5. Replaceable Parts"

"Tail Cone"

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-003	DATE:			
CUSTOMER: Delta Air Lines Inc.	MCL 10,356 DTD			
CHANGE NO: 189	MODEL: 22-2 (Convair "880")			
TITLE: Specification Administrative Change (L.G. truck position indicator light)				
ORIGIN: Convair initiated.				
REASON FOR CHANGE: To incorporate the landing gear truck position indi- cater light into the Specification language.				
EFFECT ON WEIGHT *	EFFECT ON BALANCE *			
GUAR, WT. EMPTY OPER, WT. EMPTY				
0 0	O INCH LB.			
EFFECT ON GUARANTEED PERFORMANCE: *				
Non				
NEGLIGIBLE CHANGES WILL BE ACCUMULATED A TOTALS REFLECTED IN A FUTURE CHANGE PROF ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPPRIOR ACCEPTANCE OF THE FOLLOWING CHANGE	PON			
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:				
DATE:	10 V 414, 3.L. 3.1217 4			

Title: Specification Administrative Change (L.G. truck position indi-

cator light)

Origin: Convair initiated.

Reason for Change: To incorporate the landing gear truck position indi-

cator light into the Specification language.

Description of Change:

Page 33, Paragraph 3.8.1.3 WARNING SYSTEM:

Add the following sentence to the end of the paragraph:

"An indicator light shall be provided to indicate when either main landing gear truck is not in position for retraction."

Page 59, Paragraph 3.14.1.3 ENGINE INSTRUMENT PANEL:

Add the following item to the instrument panel list:

"One main landing gear truck position indicator light."

Page 81, Paragraph 3.16.11.3 INDICATOR LIGHTS:

Revise the following item under "PILOTS" INSTRUMENT PANEL"

	Quan Reqd	Description	Color	Cap Dimming	Elec- trical Dimming
From:	"1	L.G. Truck Position	Red		X#
To:	ид	L.G. Truck Position	Amber		XII

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

The following shall not appear in the Specification language:

The effect of CCP No. 210A, approved by Delta, is included in this proposal.

Figure 3.14-1 "PILOT'S AND COPILOT'S INSTRUMENT PANELS" shows the L.G. truck position indicator light as amber.

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003	DATE:				
CUSTOMER: Delta Air Lines Inc.	MCL 10,365 DTD				
CHANGE NO: 188	MODEL: (880) 22-2				
TITLE: Specification Administrati	ve Change				
ORIGIN: Convair initiated.					
REASON FOR CHANGE: To clarify the inte	nt of the Specification.				
EFFECT ON WEIGHT *	EFFECT ON BALANCE *				
GUAR. WT. EMPTY OPER. WT. EMPTY					
0	O INCH LB.				
EFFECT ON GUARANTEED PERFORMANCE: *	EFFECT ON GUARANTEED PERFORMANCE: * None				
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND					
ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO					
PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:					
	ENGINEERING APPROVAL				
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:				
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:				
	RECURRING: NON-RECURRING:				
	TOTAL:				
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.				
BY:					
DATE:	DD1: VAIN, 5, D. 0-1217A				

CONVAIR: SD

Title: Specification Administrative Change

Origin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Page 59, Paragraph 3.14.1.4 FLIGHT ENGINEER'S INSTRUMENTS:

Add the following item to the instrument list:

"One indicator light, cabin altitude warning"

Page 122, Paragraph 3.20.2.1 DESIGN DIFFERENTIAL PRESSURE:

Add the following sentence after the third sentence which ends in the ninth line:

"An indicator light shall be provided on the flight engineer's panel to indicate cabin altitude warning."

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: 20-22-003			DATE:			
CUSTOMER: Delta Air Lines. Inc.			MCL 10,295 DTD 22 September 1			
CHANGE NO: 187			MODEL: (880)22-2			
TITLE: Audible Altitude Warning Signal, Incorporation of						
ORIGIN: TWA request	ced and Convair pr	opose	d for Delta			
REASON FOR CHANGE: To	provide audible ressurization fail	warni ure	ng indication for cabin			
			Rejected			
EFFECT	ON WEIGHT *		EFFECT ON BALANCE *			
GUAR. WT. EMPTY	OPER. WT. EMPTY					
1.0 lb	1b 1.0 lb		Negligible INCH LB.			
EFFECT ON GUARANTEED F	PERFORMANCE: * None	walles -				
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A						
ACCEPTANCE OF THIS CH. PRIOR ACCEPTANCE OF T						
			ENGINEERING APPROVAL			
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:				
N		RECURRING:				
ACCEPTED:		CONVA	AIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:						
DATE:			CONValu. 3, D 0,1217 A			
POTENTIAL PROPERTY OF THE PROP			" D' T T T T T T T T T T T T T T T T T T			

Delta Air Lines, Inc. Change No. 187

CONVAIR: SD

Title: Audible Altitude Warning Signal, Incorporation of

Origin: TWA requested and Convair proposed for Delta

Reason fer Change: To provide audible warning indication for cabin

pressurization failure

Description of Change:

Page 80, Paragraph 3.16.11.1 - LANDING GEAR AND STABILIZER WARNING HORN:

Revise the paragraph title to read as follows:

"LANDING GEAR, STABILIZER AND CABIN ALTITUDE WARNING HORN:"

Add the following sentence to the end of the paragraph:

"The warning horn shall also be utilized to give a warning signal in event of a cabin pressurization failure. A selector switch shall be provided to isolate the horn from the other systems when the alarm is actuated for cabin altitude warning only."

Page 122, Paragraph 3.20.3.1 - DESIGN DIFFERENTIAL PRESSURE:

Add the following sentence to the end of the paragraph:

"An audible warning signal shall be provided to indicate a cabin pressurization failure, as defined in Paragraph 3.16.11.1."

Effect on Weight Empty: +1.0 lb Effect on Balance: Negligible . Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003		DATE:			
CUSTOMER: Delta Air Lines Inc.			MCL 10.302 DTD		
CHANGE NO: 186			MODEL: (880) 22-22		
TITLE: Nose Land	TITLE: Nose Landing Gear Steering Unit Rack, Revision to				
ORIGIN: Convair 1	nitiated				
REASON FOR CHANGE: Product improvement Rejected					
EFFECT O	N WEIGHT *		EFFECT ON BALANCE *		
JUAR, WT. EMPTY	OPER. WT. EMPTY				
0	0		O INCH LB.		
EFFECT ON GUARANTEED P	EFFECT ON GUARANTEED PERFORMANCE: * None				
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES: ENGINEERING APPROVAL					
LATEST DATE OF ACCEPTANCE:		AIRPL	ANES AFFECTED:		
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:			
CCEPTED:		CONVA	AIR, A DIVISION OF GENERAL DYNAMICS CORP.		
BY:					
DATE:			CONVAER.5,U. 042174		

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003			DATE:		
CUSTOMER: Delta Air Lines Inc.		MCL 10,301 DTD			
CHANGE NO: 185			MODEL: (880) 22-2		
TITLE: Recirculat	ion Fan, Air Condi	itioni	ing System, Replacement o	f	
ORIGIN: Convair in	itiated :				
REASON FOR CHANGE: To wh	install new recir ich will meet Conv	reulat vair d	tion fan redesigned by ve lesign requirements	ndor	
EFFECT	N WEIGHT *		EFFECT ON BALANCE	4	
	OPER. WT. EMPTY				
0	0 . 0		0	INCH LB.	
EFFECT ON GUARANTEED P	none			¥	
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF TI	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N			
			ENGINEERING APP	PROVAL	
LATEST DATE OF ACCEPTAN	NCE:	AIRPL	ANES AFFECTED:		
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:			
		RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED:			IR, A DIVISION OF GENERAL DYNAMICS	CORP.	
BY:					
DATE:				3.5.0 G1217A	
				Charles Service 17 th	

Delta Air Lines Inc. Change No. 185

Title: Recirculation Fan, Air Conditioning System, Replacement of

Origin: Convair initiated

Reason for Change: To install new recirculation fan redesigned by

vendor which will meet Convair design requirements

Description of Change:

Page A-14, APPENDIX I-C, PRESS, ANTI-ICING AND AIR COND. EQUIPT:

Change the sixth item in the Description List as follows:

From: "1 Recirculation Blower (cabin air) Ham.

Standard 522200"

To: "1 Recirculation Blower (cabin air) Ham.

Standard 545751"

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

The following shall not appear in the Specification language:

The original version of the subject recirculation fan did not meet all Convair design performance requirements when tested under airborne conditions.

This original fan was a single stage unit using one impeller. The proposed fan is identified as a two-stage fan employing two impellers, one at each end of the drive-motor shaft, and which fully met Convair design performance requirements for ground and airborne operation. The new fan will be four inches greater in length than the original fan.

A possibility exists whereby the new fans will not be available for installation in the earlier delivered airplanes. In such case, retrofit will be accomplished on these airplanes through field service action.

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003			DATE:		
CUSTOMER: Delta Air Lines, Inc.			MCL 10,303 DTD		
CHANGE NO: 184A			MODEL: (880)22-2		
TITLE: CAR Amendment 4b-9, Inclusion of in Specification					
ORIGIN: Convair ini	tiated				
REASON FOR CHANGE: To comply with paragraph (e) Article 5 of Purchase Agreement, and revision to GCP No. 184					
EFFECT	N WEIGHT *		EFFECT ON BALANCE *		
SUAR, WT. EMPTY	OPER. WT. EMPTY				
+6.0 lb	+6.0 lb	COLUMN TO SERVICE	+5,632	NCH LB.	
EFFECT ON GUARANTEED P	ERFORMANCE: * None				
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES: ENGINEERING APPROVAL				ROVAL	
LATEST DATE OF ACCEPTANCE:		AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:			
BY:		CONVA	IR, A DIVISION OF GENERAL DYNAMICS C	ORP.	

Delta Air Lines, Inc. Change No. 184A

CONVAIR: SD

Title: CAR Amendment 4b-9, Inclusion of in Specification

Origin: Convair initiated

Reason for Change: To comply with paragraph (e) Article 5 of Purchase

Agreement, and revision to CCP No. 184

Description of Change:

Page 7, Paragraph 2.1, APPLICABLE FEDERAL SPECIFICATIONS:

Delete the following from the end of the last item in the amendment list:

"; and"

Add the following items to the amendment list:

"4b-8 (not applicable)
4b-9 adopted 27 August 1958; and"

Page 114, Paragraph 3.19.5.4 INDIVIDUAL OUTLETS:

In the first sentence, delete the word "lavatories", and add the following sentence after the first sentence:

"Two oxygen outlets shall be provided in each lavatory."

Page 114, Paragraph 3.19.5.5 MASKS:

Delete the following sentence from the paragraph:

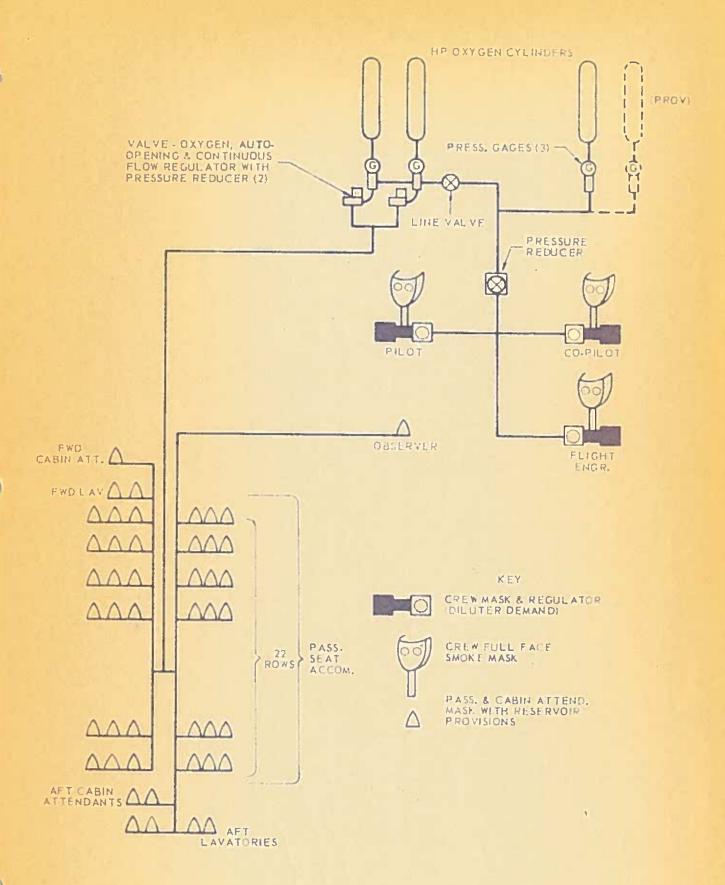
"One oxygen outlet shall be provided in each lavatory."

Enclosure: (A) One copy of sketch - GASEOUS OXYGEN SYSTEM (for preliminary use only, Figure 3.19-4 will be revised to reflect the above change)

Effect on Weight Empty: /6.0 pounds

Effect on Balance: /5,632 inch-pounds

Effect on Performance: None



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003	DATE:			
CUSTOMER: Delta Air Lines Inc.	MCL 10,299 DTD			
CHANGE NO: 183	MODEL: (880) 22-2			
TITLE: Specification Administrative Change				
ORIGIN: Convair initiated.				
REASON FOR CHANGE: To clarify the intent of the Specification.				
EFFECT ON WEIGHT *	EFFECT ON BALANCE *			
GUAR, WT. EMPTY OPER, WT. EMPTY O	O INCH LB.			
EFFECT ON GUARANTEED PERFORMANCE: * None				
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSED ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES	SAL N			
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:			
BY:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.			
DATE:	CONVAIN, SO, G-217A			

CONVAIR: SD

(e (i)

Delta Air Lines Inc. Change No. 183

Title: Specification Administrative Change

Origin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Page 5, Paragraph 1.2 GENERAL DESCRIPTION:

In line 7, change the words "vertical tail" to "fuselage".

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003		DATE			
CUSTOMER: Delta A	ir Lines Inc.	MCL	10,292	DTD	Revised 17 April 1959
CHANGE NO: 182B					ir "880")
TITLE: Weight Revi: Empty Items	sion, Designation	of Operat	ing Items	and V	Veight
Representat:	r, File 142, datedives and C. J. Mar r File 142, dated ustomer request for	y of Delta 10 June 19	on 15 Ap	ril 19	959 and
	N WEIGHT *		EFFE	ET ON BA	LANCE *
+119.0 lbs	OPER. WT. EMPTY -158.0 1bs		-96,395		INCH LB.
EFFECT ON GUARANTEED P	None				
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF TH	FUTURE CHANGE PROPOSITION OF THE PROPOSITION OF T	SAL N	Short	NGINEER	ING APPROVAL
LATEST DATE OF ACCEPTAN	NCE:	AIRPLANES A	FFECTED:		
SPECIAL PROVISIONS:		11011-116-00111111	NG:		
ACCEPTED:		CONVAIR, A DIV	ISION OF GEN	ERAL DY	NAMICS CORP.
BY:					
DATE:					CONVAIR, S. D. B-1217A

A DIVISION OF GENERAL DYNAMICS COMPORATION SAN DIE 10, CALIFORNIA

SPEC NO. 20-22-003		3	DATE:	
CUSTOMER Delta Air	Delta Air Lines Inc.		MCL 10,297 DTD	
CHANGE NO: 181			MODEL (880) 22-2	
TITLE Altimeter,	Revision to			
ORIGIN Delta reque	ested by Telecon t	o Conv	vair on 19 September 1958	
REASON FOR CHANGE:	Customer request.			
EFFECTO	N WEIGHT *		EFFECT ON BALANCE *	
O L			O INCH L.B.	
EFFECT ON GUARANTEED P	ERFORMANCE: * Nor	re		
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF TI	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	N N		
the transition of the second o			ENGINEERING APPROVAL	
LATEST DATE OF ACCEPTANCE: AIRP		AIRPLA	NES AFFECTED:	
RECUR NON-RE		RECURR NON-REC	ON PRICE PER AIRPLANE: RING: CURRING:	
CCEPTED:		CONVAIR	R, A DIVISION OF GENERAL DYNAMICS CORP.	

Delta Air Lines Inc. Change No. 181

CONVAIR: SD

Title: Altimeter, Revision to

Origin: Delta requested by Telecon to Convair on 19 September 1958

Reason for Change: Customer request.

Description of Change:

Page A-9, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Change the first item under "Kollsman Integrated Instrument System as follows:

From: "2 Altimeter Scale Error and Corrector Package Kollsman A30410-00-001"

20: "2 Altimeter Scale Error and Corrector Package Kollsman A32667-10-001"

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

A DESIGN OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003	DATE:			
CUSTOMER Delta Air Lines Inc.	MCL 10,288 DTD 13 August 1958			
CHANGE NO: 180	MODEL: (880) 22-2			
Flight Data Recorder, Chang	e in Vendor of			
Delta requested by TWX dated 11 August 1958				
REASON FOR CHANGE: Customer requested				
	Rejected			
EFFECT ON WEIGHT *	EFFECT ON BALANCE •			
Proposal 2 /65.0 lb Proposal 2 /65.0	lb Proposal 1 -5,639 inch lb Proposal 2 \$10,486 INCH LB.			
EFFECT ON GUARANTEED PERFORMANCE. No.	ne			
TOTALS REFLECTED IN A FUTURE CHANGE PROPOS	AL			
ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:				
	ENGINEERING APPROVAL			
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:			
	RECURRING: NON-RECURRING: TOTAL:			
	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:				
DATE				

Delta Air Lines Inc. Change No. 180

CONVAIR: SD

PROPOSAL No. 2

Page 62, Paragraph 3.14.3.8 FLIGHT DATA RECORDING PROVISIONS:

Delete the paragraph title and the paragraph and substitute the following:

"3.14.3.8 FLIGHT DATA RECORDER: One Minneapolis-Honeywell recording unit and one Minneapolis-Honeywell accessory unit shall be installed in the right hand electronic rack. Pitot and static lines shall be routed from the copilot's line to the recording equipment. The wiring shall include the installation of one circuit breaker. One Minneapolis-Honeywell acceleration sensor shall be installed on or near the center of gravity of the airplane. A power failure light shall be installed above the flight engineer's panel, adjacent to the door opening warning light, to indicate interruption of power to the recording equipment."

Page A9, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Add the following to the Description List:

"ILIGHT DATA RECORDER

l Recorder Minn-Honeywell
l Accessory Unit Minn-Honeywell
l Acceleration Sensor Minn-Honeywell

Effect on Weight Empty: 465.0 pounds

Effect on Balance: \$10,486 inch-pounds

Effect on Performance: None

NOTE: The effects of CCP 109A, Proposal No. 1 approved by Delta, are included in this Change.

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003			DATE:	
CUSTOMER. Delta Air	Lines Inc.		MCL 10,298	DTD 26 September 19
CHANGE NO: 1784			MODEL: 22-2	(Convair "880")
Specificat sure regul	ion Administrative ator P/N)	Chang	e (Revision	to bleed air pres-
ORIGIN: Convair in	itiated.			
	To furnish revised regulator, in lieu 178 and to show air	of pa	rt number su	
EFFECT	ON WEIGHT *		EFFE	ECT ON BALANCE .
GUAR. WT. EMPTY	OPER. WT. EMPTY			
0	0			O INCH LB.
EFFECT ON GUARANTEED				
	N	ne		
TOTALS REFLECTED IN ACCEPTANCE OF THIS C	ILL BE ACCUMULATED AND A FUTURE CHANGE PROPOS HANGE IS DEPENDENT UPO THE FOLLOWING CHANGES	SAL N		B/Limons
			*	ENGINEERING APPROVAL
LATEST DATE OF ACCEPT	ANCE:	AIRPL	ANES AFFECTED:	
RECU NON-		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:		
ACCEPTED:		CONVA	IR, A DIVISION OF G	ENERAL DYNAMICS CORP.
ВҮ:				
DATE				

Title: Specification Administrative Change (Revision to bleed air pres-

sure regulator P/N)

Origin: Convair initiated.

Reason for Change: To furnish revised part number for bleed air pressure

regulator, in lieu of part number submitted by CCP No.

178 and to show airplanes effected.

Description of Change:

This change has no effect on Specification language.

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

The following shall not appear in the Specification language:

The below is the change to the part number of subject regulator. All other items submitted by CCP No. 178 remain unchanged.

For Airplanes No. 1 and No. 2

From: "Press. Reg. 108972-400 (Convair P/N 22-02462 Bleed Air Pressure Regulator and Shut-Off Valve)"

To: "Press. Reg. 108972-400-1 (Convair P/N 22-02462-5 Bleed Air Pressure Regulator and Shut-Off Valve)"

For Airplanes No. 3 through No. 10

From: "Press. Regulator 108972-400 (Convair P/N 22-02462 Bleed Air Pressure Regulator and Shut-Off Valve)"

To: "Press. Regulator 108972-400-2 (Convair P/N 22-02462-7 Bleed Air Pressure Regulator and Shut-Off Valve)"

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003	DATE: Revised		
CUSTOMER: Delta Air Lines Inc.	MCL 10,290 DTD 17 December 1		
CHANGE NO: 177A	MODEL: 22-2 (Convair "880")		
TITLE VG Recorder, NACA Oil Damped,	Installation of		
ORIGIN: NACA request to install subje	ect equipment		
ONION. MACE LOGICODO OF SINSPERS DOLLO			
REASON FOR CHANGE: As above			
EFFECT ON WEIGHT *	EFFECT ON BALANCE *		
GUAR. WT. EMPTY OPER. WT. EMPTY			
See note on second sheet	INCH LB.		
EFFECT ON GUARANTEED PERFORMANCE: *			
None			
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL			
ACCEPTANCE OF THIS CHANGE IS DEPENDENT UP	PON		
PRIOR ACCEPTANCE OF THE FOLLOWING CHANGE	0,500		
	ENGINEERING APPROVAL		
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:		
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:		
	RECURRING:		
	:JATCT		
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.		
BY:			
DATE:	TON VALIS, STD. 642 (7A		
17 Abrah Colombia			

Oil Dawed, Installation of istall subject equipment

e .

aph 3.14.3.10 - VG RECORDER PROVISIONS:

paragraph title and paragraph to read as follows:

VG RECORDER: One VG recorder shall be installed on each of four airplanes at the approximate center of gravity of the airplane in bottom of the fusciage aft of the rear spar. The installation shall include mounting, and tubing from the putot and static lines connecting to the flight recorder (See Paragraph 3.14.3.8).

Not to be a part of the Specification language:

The following weight effect shall be subtracted from the as weighed figures when considering guaranteed weight of the affected airplanes and is not to be included in the accumulative specification weight figures.

Effect on Weight: +2.0 Lbs

Effect on Falance +1,670 Inch/Lbs

Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-003			DATE:	
CUSTOMER Delta Air Lines Inc.			MCL 10,291 DTD	
CHANGE NO: 176			MODEL: (880) 22-2	
TITLE: Specification	Administrative C	hange		
ORIGIN: Convair initia	ated.			
REASON FOR CHANGE: To	clarify the inte	ent of	the Specification.	
EFFECT ON WEIGHT *		EFFECT ON BALANCE *		
GUAR, WT. EMPTY	OPER. WT. EMPTY			
0	0		o INCH LB.	
EFFECT ON GUARANTEED PERFORMANCE: * None				
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:				
			ENGINEERING APPROVAL	
LATEST DATE OF ACCEPTAN		AIRPLA	ANES AFFECTED:	
SPECIAL PROVISIONS.		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:		
ACCEPTED:		CONVAI	R, A DIVISION OF GENERAL DYNAMICS CORP.	
BY:				
DATE:			ROHMAN ILLE CONSTRUCTOR	

Title: Specification Administrative Change.

Origin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Page 60, Paragraph 3.14.3.2.2 STATIC SELECTOR VALVES:

Change the following words in the first line:

From: "toggle-type"

To: "rotary-type"

Change the following word in the last line:

From: "instrument"

To: "auxiliary"

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003			DATE:	
CUSTOMER: Delta Air Lines Inc.			MCL 10,289 DTD	
CHANGE NO: 175			MODEL: (880) 22-2	
TITLE: Specification Administrative Change				
ORIGIN: Convair initiated, and as agreed by Delta Letter File 142, dated 29 July 1958. REASON FOR CHANGE: To delete the landing gear control throttle switches, and to make Specification compatible with the airplane.				
EFFECT ON WEIGHT *			EFFECT ON BALANCE *	
GUAR. WT. EMPTY	OPER, WT. EMPTY			
0	0		O INCH LB.	
EFFECT ON GUARANTEED PERFORMANCE: * None				
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES: ENGINEERING APPROVAL				
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:	
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:		
ACCEPTED: BY:	the state of the s	CONVA	IR, A DIVISION OF GENERAL DYNAMICS CORP.	
DATE:			CONVARIADO GARITA	

Title: Specification Administrative Change

Origin: Convair initiated, and as agreed by Delta Letter File 142,

dated 29 July 1958.

Reason for Change: To delete the landing gear control throttle switches

and to make Specification compatible with the airplane.

Description of Change:

Page 32, Paragraph 3.8.1.2 CONTROLS:

Revise the fifth sentence, starting in the 12th line to read as follows:

"The control lever shall remain locked until the left hand shock strut is extended and both main landing gear trucks are in the level zone."

Page 80, Paragraph 3.16.11.1 LANDING GEAR WARNING HORN:

Mevise the paragraph title and paragraph to read as follows:

"LANDING GEAR AND STABILIZER WARNING HORN: A horn shall be provided to operate under the following conditions:

When any landing gear is not fully extended and locked with the aircraft airborne, the horn will operate if any power lever is retarded or if the flaps are in the range of approach to land. The power lever retard function will have a horn interrupter switch.

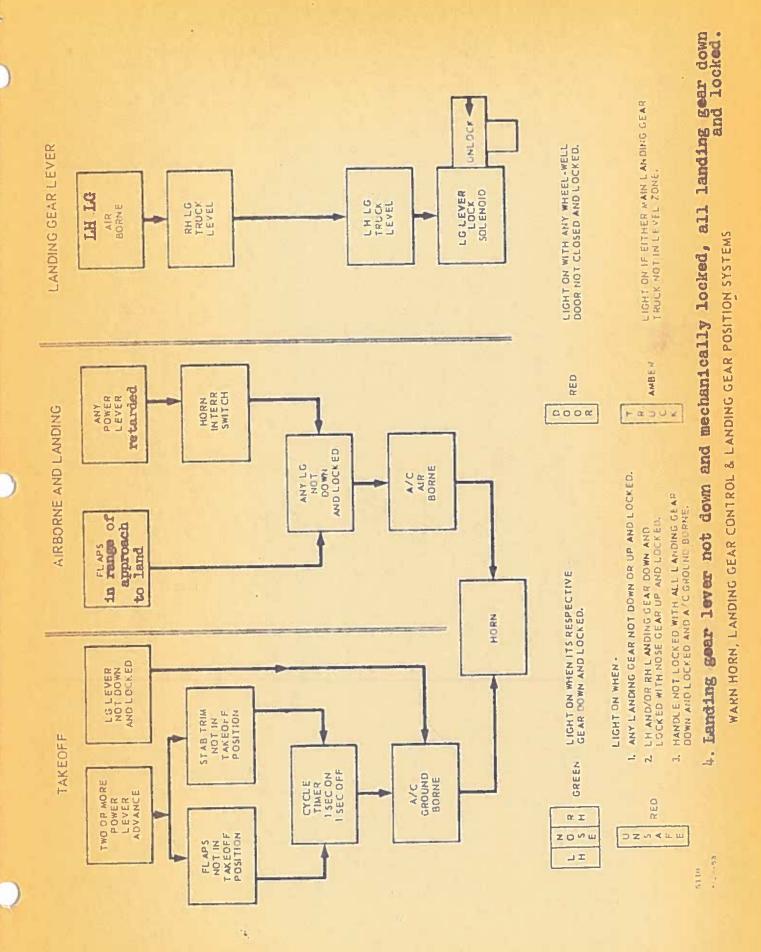
When the aircraft is ground borne, the horn will operate if the landing gear override lever is moved.

When the airplane is on the ground and any two or more of the power levers are advanced to take-off position, the horn will operate if the stabilizer and/or the flaps are not in the correct take-off position."

LANDING GLAR POSITION SYSTEMS (for information only)

Effect on Weight Empty: 0 Effect on Balance: 0

iffect on Performance: None



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO. ZD- 22-003		DATE:		
CUSTOMER: Delta Air Lines Inc.		MCL 10,287 DTD		
CHANGE NO. 174		MODEL: (880) 22-2		
TITLE: Specification Administrative Change				
ORIGIN: Convair initiated. REASON FOR CHANGE: To make landing gear component heat treat levels compatible with increased design loads.				
EFFECT	N WEIGHT *		EFFECT ON BALANCE *	
GUAR, WT EMPTY	OPER. WT. EMPTY			
0	0		O INCH LB.	
EFFECT ON GUARANTEED PERFORMANCE: * None				
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		ENGINEERING APPROVAL		
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:	
SPECIAL PROVISIONS:		RECUI NON-R	CT ON PRICE PER AIRPLANE: RRING: RECURRING: L:	
ACCEPTED:		CONV	AIR, A DIVISION OF GENERAL DYNAMICS CORP.	
BY:				
DATE:			009 VAIN, J. D. 1-1217 A	

Delta Air Lines Inc. Change No. 174

CONVAIR: SD

Specification Administrative Change Title:

Origin: Convair initiated.

Reason for Change: To make landing gear component heat treat levels compatible with increased design loads.

Description of Change:

Page 38. Paragraph 3.8.4.8 MATERIAL PROPERTIES:

Revise the first sentence to read as follows:

"The main landing gear shock strut outer cylinder and piston, main landing gear upper torque arm bolt and the wheel truck beam in addition to the nose gear side brace bolt, nose gear steering pinion and nose gear steering rack may be made from SAE 4340 steel, heat treated to 260,000 - 280,000 psi."

Effect on Weight Empty: Effect on Balance: 0 Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-003			DATE:
CUSTOMER: Delta Air Lines Inc.			MCL 10,277 DTD 10 July 1958
CHANGE NO: 173A			MODEL: (880) 22-2
TITLE: Ramp Weigh	t and Takeoff Wei	ght,	Increase of
ORIGIN: TWA reques	ted and Convair p	ropose	ed for Delta
REASON FOR CHANGE	As above, and rev	ision	to CCP No. 173.
The second secon	ON WEIGHT *		EFFECT ON BALANCE *
GUAR, WT. EMPTY	OPER. WT. EMPTY	- 2	
≠84.0 lb	≠84.0 lb		₹70,331 INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: *		
		noted.	
* NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A	L BE ACCUMULATED AND	SAL	
ACCEPTANCE OF THIS CHA			
PRIOR ACCEPTANCE OF TH	TE FOLLOWING CHANGES		
			ENGINEERING APPROVAL
LATEST DATE OF ACCEPTAN	NCE:	AIRPLA	ANES AFFECTED:
C C C C C C C C C C C C C C C C C C C			
		EFFECT ON PRICE PER AIRPLANE:	
		11011-110	RING:
		TOTAL:	
CCEPTED:		CONVAI	R, A DIVISION OF GENERAL DYNAMICS CORP.
BY:			
DATE:			
			ON VAIH, 5. 0. 5-1217A

Delta Air Lines Inc. Change No. 173A

CONVAIR: SD

Ramp Weight and Takeoff Weight, Increase of Title:

TWA requested and Convair proposed for Delta Origin:

Reason for Change: As above, and revision to CCP No. 173

Description of Change:

Page 17. Paragraph 3.4.1 STRENGTH:

Change the first two listed items as follows:

From:

To

Enclosure: (A) One copy of C.G. DESIGN LIMITS (for preliminary use only)

Effect on Weight Empty: /84.0 pounds

Effect on Balance: #70,331 inch-pounds Effect on Performance: As noted.

The following shall not appear in the Specification language:

"Weight increase of 84 lbs is based on C.G. Limits shown on Enclosure (A). If the airplane is operated at gross weight above 180,000 lbs, with center of gravity locations outside of the indicated limits, heavier tires and main landing gears will be required."

A NEUFFEL & ESSENCE

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003			DATE:
CUSTOMER: Delta Air Lines Inc.			MCL 10,284 DTD
CHANGE NO: 172			MODEL: (880) 22-2
TITLE: Specification Administrative Change			
ORIGIN: Convair initiated (Based on Delta Letter File 142, dated 23 July 1958)			
REASON FOR CHANGE: To waive Convair functional checkout of the Selcal and DMET electronic equipment.			
EFFECT (N WEIGHT *		EFFECT ON BALANCE *
GUAR, WT. EMPTY			
0	0		O INCH LB.
EFFECT ON GUARANTEED PERFORMANCE: * None			
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			
			ENGINEERING APPROVAL
LATEST DATE OF ACCEPTA	NCE:	AIRPLA	ANES AFFECTED:
SPECIAL PROVISIONS:		RECUR NON-RE	T ON PRICE PER AIRPLANE: RING: CURRING:
ACCEPTED:		CONVA	R, A DIVISION OF GENERAL DYNAMICS CORP.
BY:			
DATE:	nousy		CONVAIR, 3.0, 6-1217A

Delta Air Lines Inc. Change No. 172

CONVAIR: SD

Title: Specification Administrative Change

Origin: Convair initiated (Based on Delta Letter File 142, dated 23

July 1958)

Reason for Change: To waive Convair functional checkout of the Selcal

and DIET electronic equipment.

Description of Change:

Page 90, Paragraph 3.17.1 EQUIPMENT:

Add an asterisk before the first two items listed under "Complete provisions for the following systems shall be made:"

*Dual Distance Measuring System
*Dual Channel Airborne Selective Calling System

Add the following note to the bottom of Page 90:

"*The last sentence of paragraph 6.1.1.1 shall not apply for these items."

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None

CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003			DATE:		
CUSTOMER: Delta Air Lines Inc.			MCL 10,282	DTD 23	July 1958
CHANGE NO: 171		MODEL: (880) 22-2			
TITLE: Exterior Markings and Color Schemes, Delta Selected					
ORIGIN: Letter of Agreement No. 16 to Purchase Agreement, Convair to Delta Letter File 6-6976, dated 6 November 1956, Delta to Convair Letter File 142, dated 18 July 1958 and Delta Letter to Convair, File 142, dated 22 August 1958. REASON FOR CHANGE: Customer requested.				ir to to Con- er to	
EFFECT	N WEIGHT *		EFFEC	T ON BALAN	CE *
GUAR. WT. EMPTY	OPER. WT. EMPTY		<i>¥</i> 5	6,573	INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: *				
	None				
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			E	NGINE ERING A	ADDDOVAL
				TOINE ERING /	ALLKOVAL
LATEST DATE OF ACCEPTANCE:		AIRPL	ANES AFFECTED:		
SPECIAL PROVISIONS:		RECUR NON-RI	T ON PRICE PER AI RING: ECURRING:		
ACCEPTED:		CONVA	IR, A DIVISION OF GEN	NERAL DYNAMI	CS CORP.
BY:					
DATE:	DATE:				

CONVAIR: SD

Title: Exterior Markings and Color Schemes, Delta Selected

Origin: Letter of Agreement No. 16 to Purchase Agreement, Convair to Delta Letter File 6-6976, dated 6 November 1956, Delta to Convair Letter File 142, dated 18 July 1958 and Delta Letter to Convair, File 142, dated 22 August 1958.

Reason for Change: Customer requested.

Description of Change:

This change is only for weight accountability, and has no effect on Specification language. The exterior markings will be in accordance with Delta Air Lines Drawing "Markings Installation - CV 880" Number 110-190R, Revision "A" dated 18 August 1958.

Effect on Weight Empty: /61.0 pounds

Effect on Balance: \(\frac{1}{256}\), 573 inch-pounds

Effect on Performance: None

CONVAIA A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-00	3	DA'l'E:			
CUSTOMER: Delta A	irlines Inc.	MCIDtd_			
CHANGE NO: 169		MODEL: (880) 22-2			
TITLE: Specification Administrative Change					
ORIGIN: DAL Buffe	ORIGIN: DAL Buffet Mockup Review Comments dated 26 February 1958				
REASON FOR CHANGE: Customer request as contained in mockup review com- ments.					
	EFFECT ON WEIGH				
Guar. Wt. Empty	Oper. Wt. Empt	У			
0	0 ,	o Inch Lb.			
EFFECT ON GUARANTEED PERFORMANCE: * None					
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal					
Acceptance of the Prior Acceptance	is Change is Deper of the Following	ndent Upon Changes:			
		Engineering Approval			
LATEST DATE OF ACCI	EPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS		EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:			
ACCEPTAD:		CONVAIR, A Div. of Gon. Dyn. Corp.			
BY:					
DATE:					

CONVAIR: SD

Title: Specification Administrative Change

Origin: DAL Buffet Mockup Review Comments dated 26 February 1958

Reason for Change: Customer request as contained in mockup review com-

ments.

Description of Change:

Page 106, Paragraph 3.19.2.1 BUFFET:

Revise item (b) in the second line as follows:

"(b) a single door for the three inboard tray"

Page 106, Paragraph 3.19.2.1.2 BUFFET EQUIPMENT:

Revise the second item in the equipment list under unit No. 2 and No. 3 as follows:

From: "5 5"

To: "6 6"

Delete the fourteenth item in the equipment list as follows:

"*Drawers (4) 0 1 1 0"

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None

The following not to be a part of the Specification language:

This change modifies but does not supercede CCP No. 35B.

CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003	DATE: Revised
CUSTOMER: Delta Air Lines Inc.	MCL 10,274 DTD 2 September 195
CHANGE NO: 168A	MODEL: (880) 22-2
for	ts, Installation of/or Provisions
ORIGIN: Delta Letter, File 142, dat August 1958 between C. J. M tives.	sed 26 March 1958 and meeting of 29 lay of Delta and Convair representa-
REASON FOR CHANGE: Customer request,	
	Rejected
EFFECT ON WEIGHT *	EFFECT ON BALANCE
GUAR, WT. EMPTY OPER, WT. EMPTY	
See Attached Sheet	INCH LB.
EFFECT ON GUARANTEED PERFORMANCE: *	
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSE	
ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:	
	ENGINEERING APPROVAL
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:
	RECURRING:
	TOTAL:
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.
BY:	
DATE:	DONIVATION, OF GREETA

WEIGHT DATA

PROPOSAL "A"

Effect on Guaranteed Weight Empty: #16.0 pounds Effect on Operating Weight Empty: /16.0 pounds

Effect on Balance: #3,260 inch-pounds Effect on Performance: None

PROPOSAL "B"

Effect on Guaranteed Weight Empty: /3.0 pounds Effect on Operating Weight Empty: \(\frac{2}{3.0}\) pounds

Effect on Balance: \(\frac{2}{752}\) inch-pounds

Effect on Performance: None

PROPOSAL "C"

Effect on Guaranteed Weight Empty: /16.0 pounds

Effect on Operating Weight Empty: /16.0 pounds Effect on Balance: /3,260 inch-pounds

Effect on Performance:

PROPOSAL "D"

Effect on Guaranteed Weight Empty: 450.0 pounds Effect on Operating Weight Empty: /50.0 pounds

Effect on Balance: #13,183 inch-pounds Effect on Performance: None

PROPOSAL "E"

Effect on Guaranteed Weight Empty: ≠16.0 pounds Effect on Operating Weight Empty: \$16.0 pounds Effect on Balance: +3,925 inch-pounds

None

Effect on Performance:

PROPOSAL "F"

Effect on Guaranteed Weight Empty: #51.0 pounds

Effect on Operating Weight Empty: Effect on Balance: #13,735 inch-pounds

Effect on Performance: None CONVAIR: SD

Delta Air Lines Inc. Change No. 168A

Page 1 of 9

Title: John Oster Flight Instruments, Installation of/or Provisions

for

Origin: Delta Letter, File 142, dated 26 March 1958 and meeting of

29 August 1958 between C. J. May of Delta and Convair repre-

sentatives.

Reason for Change: Customer request, and revision to CCP No. 168.

Description of Change:

PROPOSAL "A"

(Climb, Cruise and Approach System, Provisions)

Add the following paragraph to a continued page 62:

"3.14.3.11 CLIMB, CRUISE AND APPROACH SYSTEM PROVISIONS: Complete provisions shall be made for the later installation of a climb, cruise and approach system consisting of the following:

a. 1 Altitude Transducer (Oster AX-112-0000-000)

b. 1 Control Panel (Oster 9950-02)

c. 1 Computer (Oster 9823-06)

d. 2 Indicator, Airspeed (Kollsman)
(incorporating command
speed index)

e. l Fuel Quantity (Simmonds 393012-04642)"
Totalizer Indicator

Effect on Weight Empty: /16.0 pounds

Effect on Balance: /3,260 inch-pounds

Effect on Performance: None

CONVAIR: SD

Delta Air Lines Inc. Change No. 168A

Page 2 of 9

PROPOSAL "B"

(Safe Take-off - Provisions)

Add the following paragraph to a continued page 62:

- "3.14.3.11 SAFE TAKE-OFF INDICATION SYSTEM, PROVISIONS: Complete provisions shall be made for the later installation of a safe take-off indication system consisting of the following:
 - a. 2 Indicator, Airspeed (Kollsman) (incorporating command speed index)
 - b. 1 Control Panel (Oster)

Effect on Weight Empty: /3.0 pounds

Effect on Balance: #752 inch-pounds

Effect on Performance: None

Page 3 of 9

PROPOSAL "C"

(Climb, Cruise and Approach System Provisions, and Safe Take-Off Provisions)

Add the following paragraphs to a continued page 62:

- "3.14.3.11 CLIMB, CRUISE AND APPROACH SYSTEM, PROVISIONS: Complete provisions shall be made for the later installation of a climb, cruise and approach system consisting of the following:
 - (Oster AX-112-0000-000) a. 1 Altitude Transducer
 - b. 1 Control Panel (Oster 9950-02) (Oster 9823-06) c. 1 Computer
 - d. 1 Fuel Quantity Totalizer (Simmonds 393012-04642)" Indicator
- "3.14.3.12 SAFE TAKE-OFF INDICATION SYSTEM, PROVISIONS: Complete provisions shall be made for the later installation of a safe take-off indication system consisting of the following:
 - Indicator, Airspeed (Kollsman) (incorporating command speed index)
 - b. 1 Control Panel (Oster)

Effect on Weight Empty: /16.0 pounds

/3,260 inch-pounds Effect on Balance:

Effect on Performance:

Page 4 of 9

PROPOSAL "D"

(Climb, Cruise and Approach System)

Page 58, Paragraph 3.14.1.1 PILOT'S INSTRUMENT PANEL:

Revise the first item in the instrument list as follows:

From: "*One indicator, airspeed angle of attack"

"*One indicator, airspeed (incorporating command speed To:

index)"

Page 58, Paragraph 3.14.1.2 COPILOT'S INSTRUMENT PANEL:

Revise the second item in the instrument list as follows:

"*One indicator, airspeed angle of attack" From:

"*One indicator, airspeed (incorporating command speed To:

index)"

Page 58A, Paragrah 3.14.1.5 MISCELLANEOUS INSTRUMENTS:

Add the following item to the miscellaneous instrument list:

"One computer (climb, cruise and approach system)"

Add the following paragraph to a continued page 62:

"3.14.3.11 CLIMB, CRUISE AND APPROACH SYSTEM: A climb, cruise and approach system shall be installed to indicate to the pilot, the optimum climb and approach speeds; and in the cruise mode, to indicate range and flight time remaining for the various types of cruise con-

ditions."

Page 5 of 9

PROPOSAL "D" (Cont)

(Climb, Cruise and Approach System)

Page A-7, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the first item in the description list as follows:

From: "1 Fuel Quantity Totalizer Indicator, Simmonds 393006-20585"

To: "1 Fuel Quantity Totalizer Indicator, Simmonds 393012-04642"

Page A-9, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the second item under "Kollsman Integrated Instrument System" as follows:

From "2 Indicator, Airspeed Angle of Attack, Kollsman A29297-10 plus range markings"

To: "2 Indicator, Airspeed (incorporating Kollsman (P/N to be command speed index) supplied)"

Add the following 1tems under "Kollsman Integrated Instrument System":

"1 Altitude Transducer Oster AX-112-0000-000

l Control Panel (climb, cruise Oster 9950-02 and approach system)

1 Computer Oster 9823-06"

Revise the "PILOT AND COPILOT INSTRUMENT PANEL" illustration to reflect the above changes.

Revise the "PILOT AND COPILOT CONSOLE" illustruation to show the climb, cruise and approach computer.

Effect on Weight Empty: /50.0 pounds

Effect on Balance: #13,183 inch-pounds

Effect on Performance: None

Page 6 of 9

PROPOSAL "E"

(Safe Take-Off)

Page 58, Paragraph 3.14.1.1 PILOT'S INSTRUMENT PANEL:

Revise the first item in the instrument list as follows:

From: "*One indicator, airspeed angle of attack"

To: "*One indicator, airspeed (incorporating command speed index)"

Page 58, Paragraph 3.14.1.2 COPILOT'S INSTRUMENT PANEL:

Revise the second item in the instrument list as follows:

From: "*One indicator, airspeed angle of attack"

To: "*One indicator, airspeed (incorporating command speed index)"

Add the following paragraph to a continued page 62:

"3.14.3.11 SAFE TAKE-OFF INDICATOR SYSTEM: A safe take-off indication system shall be installed to indicate take-off progress to the pilots, and to indicate the existence of a malfunction which could prevent the airplane from achieving a safe take-off speed."

Page A-9, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the second item under "Kollsman Integrated Instrument System" as follows:

From: "2 Indicator, Airspeed Angle of Attack, Kollsman A29297-10 plus range markings"

To: "2 Indicator, Airspeed (incorporating Kollsman (P/N to be command speed index) supplied)"

Add the following item under "Kollsman Integrated Instrument System":

"1 Control Panel (safe take-off system) Oster (P/N to be supplied)"

CONVAIR: SD

Delta Air Lines Inc. Change No. 168A

Page 7 of 9

PROPOSAL "E" (Cont)

(Safe Take-Off)

Revise "PILOT AND COPILOT INSTRUMENT PANEL" illustration to reflect the above change.

Effect on Weight Empty: \$\(\frac{1}{2}\)6.0 pounds

Effect on Balance: \$\(\frac{4}{3}\),925 inch-pounds

Effect on Performance: None

Page 8 of 9

PROPOSAL "F"

(Climb, Cruise and Approach System, and Safe Take-Off System)

Page 58, Paragraph 3.14.1.1 PILOT'S INSTRUMENT PANEL:

Revise the first item in the instrument list as follows:

From: "*One indicator, airspeed angle of attack"

To: "*One indicator, airspeed (incorporating command speed index)"

Page 58, Paragraph 3.14.1.2 COPILOT'S INSTRUMENT PANEL:

Revise the second item in the instrument list as follows:

From: "*One indicator, airspeed angle of attack"

To: "*One indicator, airspeed (incorporating command speed index)"

Add the following paragraphs to a continued page 62:

- "3.14.3.11 CLIMB, CRUISE AND APPROACH SYSTEM: A climb, cruise and approach system shall be installed to indicate to the pilot, the optimum climb and approach speeds; and in the cruise mode to indicate range and flight time remaining for the various types of cruise conditions.
 - 3.14.3.12 SAFE TAKE-OFF INDICATION SYSTEM: A safe take-off indication system shall be installed to indicate take-off progress to the pilots and to indicate the existence of a malfunction which could prevent the airplane from achieving a safe take-off speed."

Page 58A, Paragraph 3.14.1.5 MISCELLANEOUS INSTRUMENTS:

Add the following item to the miscellaneous instrument list:

"One computer (climb, cruise and approach system)"

Page 9 of 9

PROPOSAL "F" (Cont)

(Climb, Cruise and Approach System, and Safe Take-Off System)

Page A-7, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the first item in the description list as follows:

From: "1 Fuel Quantity Totalizer Indicator, Simmonds 393006-20585"

To: "1 Fuel Quantity Totalizer Indicator, Simmonds 393012-04642"

Page A-9, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the second item under "Kollsman Integrated Instrument System" as follows:

From: "2 Indicator, Airspeed Angle of Attack, Kollsman A29297-10 plus range markings"

To: "2 Indicator, Airspeed (incorporating Kollsman (P/N to be command speed index) supplied)

Add the following items under "Kollsman Integrated Instrument System":

"1 Altitude Transducer Oster AX-112-0000-000 1 Control Panel (climb, cruise Oster (P/N to be supplied) and approach system)

1 Computer Oster 9823-06"

Revise "PILOT AND COPILOT INSTRUMENT PANEL" and "PILOT AND COPILOT CONSOLE" illustrations to reflect the above changes.

Effect on Weight Empty: /51.0 pounds

Effect on Balance: /13,735 inch-pounds

Effect on Performance: None

CONVAIR

A DIVISION OF GENERAL DYNAMICS COMPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003	PEC NO.: ZD- 22-003		DATE:		
CUSTOMER: Delta Air Lines, Inc.		MCL 10,275 DTD			
CHANGE NO: 167A			MODEL: (880) 22-2		
TITLE: Transformer Rectifier for Buffets No. 2 and No. 3, Installation of					
ORIGIN: Convair in	Itiated.		1		
REASON FOR CHANGE: To make coffee makers functionable by providing means for designed voltage.					
FFFECT	ON WEIGHT *		EFFECT ON	BALANCE *	
GUAR, WT. EMPTY	OPER. WT. EMPTY				
45.0 lb	75.0 lb 75.0 lb		A,21	O INCH LB.	
EFFECT ON GUARANTEED	EFFECT ON GUARANTEED PERFORMANCE: * '				
	None				
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL **ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:					
PRIOR ACCEPTANCE OF	THE POLLOWING CHANGES.		ENGIN	EERING APPROVAL	
LATEST DATE OF ACCEPTANCE:		AIRPLA	NES AFFECTED:		
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED:		CONVA	R, A DIVISION OF GENERA	L DYNAMICS CORP.	
BY:					
DATE:				TONVAN (13.3 - 6-1217A	

Bransformer Rectifier for Buffets No. 2 and No. 3, Instal-Ticle:

lation of

Origin: Convair initiated.

Reason for Change: To make coffee makers functionable by providing

means for designed voltage.

Description of Change:

Page A-12, APPENDIX I-C, TURNISHINGS:

Change the bottom item in Description List as follows:

From: "2 Coffee Makers (115v ac) Nordskog Co. 50450"

"2 Coffee Makers (200-Volt, ...0

3-Phase, 400-Cycle,

A/C28v dc) Nordskog Co. 5045 "

Lifect on Weight Empty: #5.0 pounds Lifect on Balance: #4,210 inch-pounds

Effect on Performance: Hone

the following is not to appear in Specification language:

This change consists essentially of the installation of one 200v, 3-phase, 400-cycle input; 28v dc output transformer-silicon recti-fier in No. 2 and No. 3 buffets.

The transformer rectifier unit will be purchased by Convair as loose equipment with the coffee maker from Mordskog Co., Inc., and installed by Convair. This unit is Customer specified. All conditions of warranty for the unit shall be between Delta Air Lines and the Hordskog Co., because no environmental testing for the unit has been incorporated in this proposal per Delta request.

CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION . SAN DIE 50, CALIFORNIA

SPEC NO.: ZD- 22-003		DATE:	
CUSTOMER: Delta Air Lines Inc.		MCL 10,270 DTD 12 June 1958	
CHANGE NO: 166A		MODEL: (880) 22-2	
TITLE: Specification Administrative Change (Miscellaneous Specification Revisions)			
ORIGIN: Convair initiated, and request for revision to CCP No. 166 by Delta letter, File 142, dated 9 January 1959			
REASON FOR CHANGE: Se	e body of this CO	CP	
	ON WEIGHT *	EFFECT ON BALANCE *	
GUAR. WT. EMPTY	OPER. WT. EMPTY		
0	0	O INCH LB.	
EFFECT ON GUARANTEED PERFORMANCE: * None			
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF THE	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	DSAL DN	
		ENGINEERING APPROVAL	
LATEST DATE OF ACCEPTANCE:		AIRPLANES AFFECTED:	
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:	
		RECURRING: NON-RECURRING: TOTAL:	
ACCEPTED:		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.	
BY:			
DATE:		CONVAIR, 9, 0. 6.1217A	

CONVAIR: SD

Delta Air Lines Inc. Change No. 166A

Page 1 of 2

Title: Specification Administrative Change (Miscellaneous Specification

Revisions)

Origin: Convair initiated, and request for revision to CCP No. 166 by

Delta letter, File 142, dated 9 January 1959

Reason for Change: See body of this CCP

Description of Change:

Page 3, Paragraph 1.1 BASIC TYPE:

Revise the first sentence as follows:

From: "All hinges shall be attached with screws or bolts and anchor nuts, except that it will be permissible to use rivets to attach the wing leading edge section to the forward legs of the piano hinges."

To:

"All hinges shall be attached with screws or bolts and anchor nuts, except wing access door hinges may be installed without anchor nuts. It will be permissible to use rivets to attach the wing leading edge section to the forward legs of the piano hinges."

Page 31, Paragraph 3.7.1.6.3 PASSENGER COMPARTMENT FLOORING:

Delete the following words, starting in the 11th line:

"of underseat panels and those panels which need not be removed for servicing"

Page 55, Paragraph 3.12.9.13.1 UNDER-WING REFUELING:

In the eighth line, change "600 gpm" to "300 gpm"

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None

Page 2 of 2

The following shall <u>not</u> appear in the Specification language, and is presented as explanatory information for the proposed changes to the below items:

a. Page 3, Paragraph 1.1 BASIC TYPE:

All fasteners necessary to open doors for normal access shall be held with anchor nuts. Hinges on wing access doors will not require anchor nuts since hinges screws are not removed for normal access.

b. Page 31, Paragraph 3.7.1.6.3 PASSENGER COMPARTMENT FLOORING:

The second sentence applies specifically to "the floors in the heavy traffic areas (entry ways, galley, aisle and lavatories)", the words "of underseat panels and those panels which need not be removed for servicing" apply to panels other than those listed above. Since the last sentence of this paragraph includes the requirements for these "light duty" panels, deletion is proposed for the words "of underseat panels and those panels which need not be removed for servicing."

C. Page 55, Paragraph 3.12.9.13.1 UNDER-WING REFUELING:
The "600 gpm" capacity noted in paragraph is in error and should read "300 gpm".

A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-00	3	DATE:			
CUSTOMER: Delta A1	r Lines Inc.	MCL 10,260 Dtd 16 May 1958			
CHANGE NO: 165		MODEL: (880) 22-2			
TITLE: Landing	TITLE: Landing Flare Provisions, Deletion of				
ORIGIN: TWA req	uested and Convain	r proposed for Delta			
REASON FOR CHANGE: Flares not required for overland operation, to reduce weight and supplement to CCP No. 149.					
	EFFECT ON WEIGH	HT * . EFFECT ON BALANCE *			
Guar. Wt. Empty	Oper. Wt. Empty				
-13.0 lb	-13.0 1b	-13,863 Inch Lb.			
EFFECT ON GUARANTEED PERFORMANCE: * None None					
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal					
Acceptance of the Prior Acceptance	Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:				
		Engineering Approval			
LATEST DATE OF ACC	EPTANCE:	AIRPLANES AFFECTED:			
The second secon		EFFECT ON PRICE PER AIRPLANE:			
SPECTAL PROVISIONS		Recurring: Non-Recurring Total:			
SPECIAL PROVIDIOUS ACCIPTED:		Recurring: Non-Recurring			
ACCIPTED:		Recurring: Non-Recurring Total:			

CONVAIR: SD

Title: Landing Flare Provisions, Deletion of

Origin: TWA requested and Convair proposed for Delta

Reason for Change: Flares not required for overland operation, to reduce weight and supplement to CCP No. 149.

Description of Change:

Page 73, Paragraph 3.17.7.7.5 FLARE DISPENSERS:

After paragraph title add: "(Not Applicable)", and delete the remainder of the paragraph.

Page 108, Paragraph 3.19.2.7.1 FLARE DISPENSERS:

After paragraph title add: "(Not Applicable)", and delete the remainder of the paragraph.

Page A-16, APPENDIX I-C, PYROTECHNICS:

After "PYROTECHNICS" add "(not Applicable)"

Delete the following item from the Description List:

72 Flare Containers

13.0 lb"

Delete "5/19C" from the left hand margin of page.

Effect on Weight Empty: -13.0 pounds

Effect on Balance: -13,863 inch-pounds

Effect on Performance: None

C O N V A T d A Division of General Dynamics Corporation San Diego, California

COMMERCIAL CHANGE PROPOSAL COMMERCIAL CHANGE PROPOSAL

SPIC NO: ZD-22-00		DATE:	
CUSTOMER: Delta Air Lines Inc.		MCL 10,263 Ptd	
CHANGE NO: 164		MODEL: (880) 22-2	
TITER: Provisions for Ground Combustor Starting at No. 1 and No. 2 Nacelles, Installation of			
ORIGIN: Delta requested by Telecon, CONVAIR-DELTA on 12 May 1958.			
REASON FOR CHANGE:	Customer requested.	Rejeted	
		V o y	
	EFFECT ON WEIGHT	* EFFECT ON BALANCE *	
Guar. Wt. Empty	Oper. Wt. Empty		
+12.0 lbs	+12.0 lbs	+8,672 Inch Lb.	
EFFECT ON GUARANTEED PERFORMANCE: * None			
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal.			
Acceptance of th	is Change is Depende of the Following Ch	anges:	
Engineering Approval			
LATEST DATE ACCEPTANCE: AIRPLANES AFFECTED:			
Rec Hon		FECT ON PRICE PER ATTPLANT: courring: chalicatel:	
ACCLPEND:		TIVI.TR, A Div. of Gon. Dyn. Cor.	
BY:	IA .		

Page 1 of 2

Title: Provisions for Ground Combustor Starting at No. 1 and No. 2 Nacelles, Installation of

Origin: Delta requested by Telecon, CONVAIR-DELTA on 12 May 1958.

Reason for Change: Customer requested.

Description of Change:

CONVAIR: SD

Page 44. Paragraph 3.11.9 GROUND COMBUSTOR STARTER ACCESS:

Delete the first sentence and substitute the following:

"An access door shall be installed in each of the right hand doors of the four engine nacelles. The engines in No. 3 and No. 4 nacelles shall be provided with an electrical receptacle and duct connector for utilization of a ground in-line combustor for engine starting."

Page 47, Paragraph 3.12.4.4 STARTERS:

In the sixth line, change the word "line" to "in-line".

Insert the following sentence after the fourth sentence ending in the seventh line:

"The No. 3 and No. 4 engine assemblies having ground in-line combustor and electrical receptacles, are interchangeable with No. 1 and No. 2 engine assemblies to provide the capability of combustor cart starting at any two of the four engines."

Page 57, Paragraph 3.12.12 STARTING SYSTEM:

Revise the second sentence to read as follows:

"A control switch, to be energized from an outside source, shall be provided to permit selective ground line combustor starting from the cockpit."

Page A-3, APPENDIX I-C, ELECTRICAL EQUIPMENT:

Revise the second item from bottom of page as follows:

From: "2 Control Switches (For ground starter equipment)"

To: "1 Control Switch (For ground starter equipment)"

CONVAIR: SD

Delta Air Lines Inc. Change No. 164

Page 2 of 2

Figure 3.12-1 ENGINE STARTING SYSTEM:

Revise figure to show control switch and four-pole relay for ground starting.

Effect on Weight Empty: +12.0 pounds Effect on Balance: +8,672 inch-pounds Effect on Performance: None

The following is not to appear in the Specification language:

The reference to control switch in paragraph 3.12.12 and in APPENDIX I-C is a correction of existing language and not a part of the design change represented by this proposal.

CONVAIR

ENVISION OF GENTRAL DYNAMICS FORFORATION SAN DIE 10, CALIFORNIA

SPEC NO.: ZD- 22-003	DATE:	
CUSTOMER: Delta Air Lines Inc.	MCL 10,262 DTD	
CHANGE NO 162A	MODEL: (880) 22-2	
TITLE Kits for Conversion of 12-1	Place Club Area (Excluding Seats)	
ORIGIN. Convair initated.		
REASON FOR CHANGE: To provide for all	lternate seating arrangement.	
FFFECT ON WEIGHT	EFFECT ON BALANCE *	
GUAN, WT. EMPTY OPER, WT. EMPTY		
See weight statement at end of	CCP. INCH LB.	
EFFECT ON GUARANTEED PERFORMANCE: None		
A TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL A TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL A TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL PRIOR ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES: ENGINEERING APPROVAL		
LATEST DATE OF ACCEPTANCE.	AIRPLANES AFFECTED:	
SPECIAL PROVISIONS	RECURRING: NON-RECURRING: TOTAL.	
).CCEPTED!	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.	
BY		
CATE:	Maria Maria	

Title: Coach Kits for Conversion of 12-Place Club Area (Excluding

Seats)

Origin: Convair initated.

Reason for Change: To provide for alternate seating arrangement.

Description of Change:

Page 105, add the following new paragraph to the page, after paragraph 3.19.1.1.8:

"3.19.1.1.8.1 CONVERSION KITS: Kits for conversion of the 12place club area shall be provided as loose equipment for installation by the Buyer. The conversion
kits shall include the trim strip, left and right
hand hat racks for five-abreast seating, passenger
convenience pods, carpet, sidewall and floor seat
attach fittings. Alternate light fixtures shall be
provided in the club area overhead when the club
area is converted to first class or coach."

The following is not to appear in the Specification language:

The below is a detailed description of the items which are included in the conversion kit:

- A. A 2-inch wide trim strip, to be installed in the remaining trim gap when the partition between club area and main cabin is removed.
- B. Left and right hand hat racks for use with five-abreast seating arrangement including three passenger convenience pods for installation in each hat rack. The pods shall include the following:

RIGHT HAND:

LEFT HAUD:

3 Oxygen Masks 3 Oxygen Masks

2 Gaspers 3 Gaspers

2 Reading Lights and Switches 3 Reading Lights and Switches 1 Stewardess Call Button 1 Stewardess Call Button

- C. Carpet, to match main cabin carpet.
- D. Sidewall seat attach fittings and floor seat attach fittings identical to those in the main cabin.

WEIGHT SUMMARY: (for information only)

Total Weight of Items Removed: Dependent upon Customer

CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003			DATE:
CUSTOMER: Delta Air Lines Inc.			MCL 10,259 DTD 19 May 1958
CHANGE NO: 160			MODEL: (880) 22-2
TITLE: Specification Administrative Change			
ORIGIN: Convair initiated. REASON FOR CHANGE: To clarify the intent of the Specification			
TEASON FOR CHARGE.	o clarity the into	ent of	the Specification.
	ON WEIGHT *		EFFECT ON BALANCE *
GUAR, WT, EMPTY	OPER. WT. EMPTY O		O INCH LB.
EFFECT ON GUARANTEED PERFORMANCE: * None		None	
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		SAL N	ENGINEERING APPROVAL
LATEST DATE OF ACCEPTANCE:		AIRPL	ANES AFFECTED:
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:	
ACCEPTED: CO		CONVA	IR, A DIVISION OF GENERAL DYNAMICS CORP.
DATE			CONVAIR, S.D. 6-1217 A

CONVAIR: SD

Title: Specification Administrative Change

Origin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Page 111, Paragraph 3.19.3.3 TRIM:

Change the period at the end of second sentence to a comma and add the following:

"except for the panels above the cabin hat racks which shall not exceed approximately 15 feet."

See S.O. 880-2-30 for variation to this CCP

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

C O N V A T R A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-003	DATE:			
CUSTOMER: Delta Air Lines Inc.	MCL 10,258 Dtd			
CHANGE NO: 159	MODEL: (880) 22-2			
TITLE: Specification Adminis	trative Change			
ORIGIN: Convair initiated.				
REASON FOR CHANGE: To clarify the	intent of the Specification.			
EFFECT ON WEIG				
Guar. Wt. Empty Oper. Wt. Empt	Sy .			
0 0	Negligible Inch lb.			
EFFECT ON GUARANTEED PERFORMANCE:	None			
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal				
Acceptance of this Change is Depe Prior Acceptance of the Following				
	Engineering Approval			
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:	PEFECT ON PRICE PER ATTPLANE:			
	Non-Recurring Total:			
ACCEPTED:	CONVAIR, A Div. of Gen. Dyn. Corp.			
BY:				

Specification Administrative Change Title:

Origin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Page 112, Paragraph 3.19.4.3.2 WATER FIRE EXTINGUISHER:

Revise the paragraph to read as follows:

Three water fire extinguishers shall be provided; one located in the forward entrance area and two on the forward side of the partition immediately forward of the aft entrance way."

Delete peragraph: "3.19.4.3.3 PASSENGER COMPARTMENT FIRE EX-TINCUISHER

Effect on Weight Empty: Effect on Balance:

Negligible

Effect on Performance: None

A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-00	3	DATE:			
CUSTOMER: Delta Air Lines Inc.		MCL 10,251	Dtd 30 Apr	11 1958	
CHANGE NO: 158	HODEL: (880)	22-2			
TITLE: Low Energy Ignition System, Installation of					
ORIGIN: Convai	ORIGIN: Convair initiated.				
REASON FOR CHANGE: To increase service life of engine igniting plugs by installation of a low energy system for starting under less severe conditions.					
	EFFECT ON WEIGHT	* DFFE	CT ON BALA	NCE *	
Guar. Wt. Empty	Oper. Wt. Empty				
+35.0 lbs	+35.0 lbs	+26	,688 Incl	n lib.	
EFFECT ON GUARANTEED PERFORMANCE: * None					
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal					
	Acceptance of this Change is Dependent Upon Frior Acceptance of the Following Changes:				
		Eng	ineering A	ppreval	
LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:					
SPECIAL Providuous	FECT ON PRICE PR curring: n-Recurring tal:				
ACCHA TUD:	C()	IVI.ER, A Div. of	Gon. Tym.	Carry.	
3 ^V :					
onities:					

CONVAIR: SD

Title: Low Energy Ignition System, Installation of

Origin: Convair initiated.

Reason for Change: To increase service life of engine igniting plugs

by installation of a low energy system for starting

under less severe conditions.

Description of Change:

Page 78, add the following new paragraph to the page after paragraph 3.16.9:

"3.16.9.1 LOW ENERGY IGNITION SYSTEM: A secondary low energy type ignition system shall be installed, for engine starting under less sever conditions, to provide selective starting with the high energy system specified in Par. 3.16.9. Controls for ignition shall be located in the pilots' compartment."

Effect on Weight Empty: +35.0 pounds

Effect on Balance: +26,688 inch-pounds

Effect on Performance: None

A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-003		DATE:		
CUSTOMER: Delta Air Lines Inc.		MCL 10,256 Dtd 12 May 1958		
CHANGE NO: 157		MODEL: (880) 22- 2		
TITLE: Battery, C	TITLE: Battery, Change from 24 to 27.5 Volts			
ORIGIN: Convair initiated.				
REASON FOR CHANGE: To provide a storage battery of greater voltage to increase operating time of equipment, powered by battery, during an emergency.				
	EFFECT ON WEIG			
Guar. Wt. Empty	Oper. Wt. Empt	O Inch Lb.		
EFFECT ON GUARANTEED PERFORMANCE: * None				
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal				
Acceptance of the	is Change is Depe of the Following	ndent Upon Changes:		
		Engineering Approval		
LATEST DATE OF ACCE	SPTANCE:	AIRPLANES AFFECTED:		
SPECIAL PREVISIONS:		EFFECT ON PRICE PER ATTPLANE: Recurring: Non-Recurring Total:		
ACCEPTED:	necessaria de la composition della composition d	CONVAIR, A Div. of Gon. Dyn. Corp.		
BY:				
Dorna.				

CONVAIR: SD

Title: Battery, Change of from 24 to 27.5 Volts

Origin: Convair initiated.

Reason for Change: To provide a storage battery of greater voltage to

increase operating time of equipment, powered by

battery, during an emergency.

Description of Change: '

Page 66, Paragraph 3.16.2.3 BATTERIES:

Revise the first sentence to read as follows:

"Battery power of 27.5 volts shall be available."

Page 66, Paragraph 3.16.2.4.2 BATTERY COMPARTMENT FINISH:

Revise paragraph and paragraph title as follows:

"BATTERY AREA FINISH: The battery area shall be painted with electrolyte-resistant paint."

Page A-3, APPENDIX I-C, ELECTRICAL EQUIPMENT:

Change the 10th item in Description List as follows:

Prom: "1 Battery, Storage (24 volt or 2 - 12 volt) Nickel-Cadmium"

To: "1 Battery, Storage, Nickel-Cadmium, 27.5 volt"

Enclosure: (A) One copy of sketch showing in-flight and ground ventilation of battery area (for information only).

Effect on Weight Empty: O
Effect on Balance: O
Effect on Performance: None

The following is not to appear in the Specification language:

This change also relocates the battery from the main wheel well to the pressurized fuselage in the forward nose area. The advantages of this relocation are:

- A. To avoid excessive voltage drop due to low temperatures encountered in the unpressurized main wheel well.
- B. Battery would be subject to considerable less vibration in the proposed location thereby increasing its service life.

C O N V A T ? A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-003			DATE:			
CUSTOMER: Delta A	MC	10,249	_ Dtd_	28 April 1958		
CHANGE NO: 156			MODEL: (880) 22-2			
TITLE: Engine Vibration Indicators, Wiring Provisions for						
ORIGIN: TWA requested and Convair proposed for Delta						
REASON FOR CHANGE: To have wiring provisions available for the later installation of an engine vibration indicator system.						
	EFFECT ON WEI		- EF	FECT OF	N BALANCE *	
Guar. Wt. Empty	Oper. Wt. Emp	ty				
+51.0 lbs	+51.0 lbs	3			Inch Lb.	
EFFECT ON GUARANTEED PERFORMANCE: * None Rejected						
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal						
Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:						
			ŀ	lnginee	ring Approval	
LATHET DATE OF ACCEPTANCE: A			ES AFFECT	PED:		
SPECIAL PROVISIONS	EFFECT ON PRICE PER ALCPLANE: Recurring: Non-Recurring Total:					
ACCEPT 'D:		CONVAIR, A Div. of Gen. Dyn. Corp.				
BY:						
DATES:						

Delta Air Lines Inc. Change No. 156

CONVAIR: SD

Title: Engine Vibration Indicators, Wiring Provisions for

Origin: TWA requested and Convair proposed for Delta

Reason for Change: To have wiring provisions available for the later installation of an engine vibration indicator system.

Description of Change:

Page 43, add the following new paragraph to the page after paragraph 3.11.4:

"3.11.4.1 ENGINE VIBRATION INDICATORS (Wiring Provisions): Wiring provisions shall be installed between each of the four nacelles and the vicinity of the radio rack area to permit the future installation of an engine vibration indicator system. The wiring provisions shall consist of three each twisted double conductor shielded leads for the later installation of three vibration pickups on each engine. The terminations of the wiring at the nacelles and radio rack shall consist of coiling, taping and stowing a reasonable amount of wiring to facilitate the future installation.

Effect on Weight Empty: +51.0 pounds
Effect on Balance: +38,159 inch-pounds
Effect on Performance: None

SPEC NO: ZD-22-00	3	_	DATE:	
CUSTOMER: Delta Air Lines Inc.			MCL_ 10,257 Dta_	
CHANGE NO: 155			MODEL: (880) 22- 2	
TIPLE: Specifica	tion Administrati	ve Cha	ange	
ORIGIN: Convair in	nitiated.			
REASON FOR CHANGE:	To clarify the	intent	t of the Specification.	
2	EFFECT ON WEIG		EFFECT ON BALANCE *	
Guar. Wt. Empty	Oper. Wt. Empt	У		
0	0		O Inch lib.	
EFFECT ON GUARANTEI	ED PERFORMANCE: *	Non	ne e	
* Negligible Change Totals Reflected	es Will be Accumu in a Future Chan	lated ge Pro	and oposal	
Acceptance of thi				
			Engineering Approval	
LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:				
SPECIAL PROVISIONS:		Recur	er on PRICE PER ATEPLANE: rring: Recurring	
ACCI.PT-ID:		COIV/.	TR, A biv. of Gen. Dyn. Coup.	
With the second				
BEWIS:				

Delta Air Lines Inc. Change No. 155

CONVAIR: SD

Title: Specification Administrative Change

Origin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Page 23, Paragraph 3.6.3 ELEVATORS:

Delete the words "interconnected and" in the second line of the first sentence.

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

CUSTOMER: Delta Air Lines Inc. CHANGE NO: 154A		DATE:				
		MCL 10,247 DTD 24 April 19				1958
			MODEL: (880			
TITLE: Coach Seats	Provision of					
	greement No. 12 1 1956, and TWA/Del					of
REASON FOR CHANGE: To	provide for alte	ernate	seating arra	ingements,		
EFFECT	N WEIGHT .		EFFE	CT ON BALAN	ICE *	
GUAR, WT. EMPTY	SUAR. WT. EMPTY See weight data end of CCP.				INCH	LB.
EFFECT ON GUARANTEED P	None					
TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF TH	FUTURE CHANGE PROPO	SAL				
entire Hymphipianess telebrosine and			E	NGINEERING	APPROVA	L
LATEST DATE OF ACCEPTAN	ICE:	AIRPLA	ANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:				
DATE:		CONVAI	R, A DIVISION OF GEN	IERAL DYNAMI	CS CORP.	

Delta Air Lines Inc. Change No. 154A

Page 1 of 2

tle: Coach Seats, Provision of

Origin: Letter of Agreement No. 12 to Purchase Agreement, dated 13 September 1956 and TWA/Delta/Convair Seat Design Conference of 23 April 1958.

Reason for Change: To provide for alternate seating arrangements

Description of Change:

Page 104, Add the following new paragraph to the page, after paragraph 3.19.1.16.

- "3.19.1.1.6.2 COACH SEATS: Coach seats shall be furnished to permit an alternate or mixed seating arrangement in the main cabin area. The coach seats shall include the following:
 - A. Plug-in type food trays shall be provided for use on the seat row aft of movable coat dividers and/or aft of forward partition. All seats shall contain internal provisions for future installation of plug-in food tray receptacles.
 - B. Integral folding type food trays.
 - C. Removable plug-in type intermediate arm rests on triple and double coach seats. One intermediate removable arm rest for triple seat. Shall incorporate recline mechanism and ash receiver for center seat, styled similar to the remaining plug-in arm rests."
- Enclosures: (A) One copy of sketch "Convair 880 Double Coach Seat", Sheets 1 through 3 (for information only)
 - (B) One copy of sketch "Convair 880 Triple Coach Seat", Sheets 1 through 3 (for information only)

Weights (Alternate Load):

1.	One row five-abreast seats	72.00
2.	One double coach seat P u	TDB
2	One double coach seat, R.H.	lbs
II.	One double coach seat, L.H. (one per airplane)	lbs
	The very second property of the second proper	
5.	Five plus-in trays for front row of angel seets	108
6	Five plug-in trays for front row of coach seats	lbs
	Plug-in food tray sockets - loose equipment - per row	1.bs

Page 2 of 2

EXHIBIT "A"

The following shall not appear in the Specification language:

No mock-up or prototype coach seats have been included in this proposal.

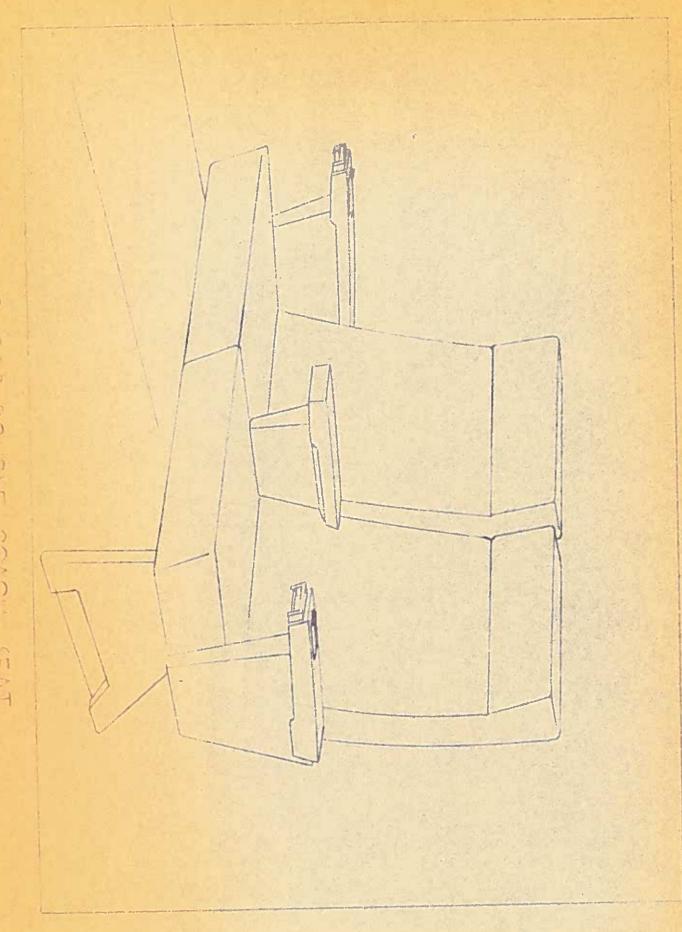
Equipment interchangeability for standard seats, as covered in CCP No. 138, is also applicable to coach seats.

Item "A" in paragraph 3.19.1.1.6.2 of this proposal, will provide five plug-in trays for first six rows of Customer order per airplane. Five additional plug-in trays will be provided for 18 or more coach seat rows per airplane. Receptacles for plug-in trays shall be provided as loose equipment on all coach seats not having receptacles installed.

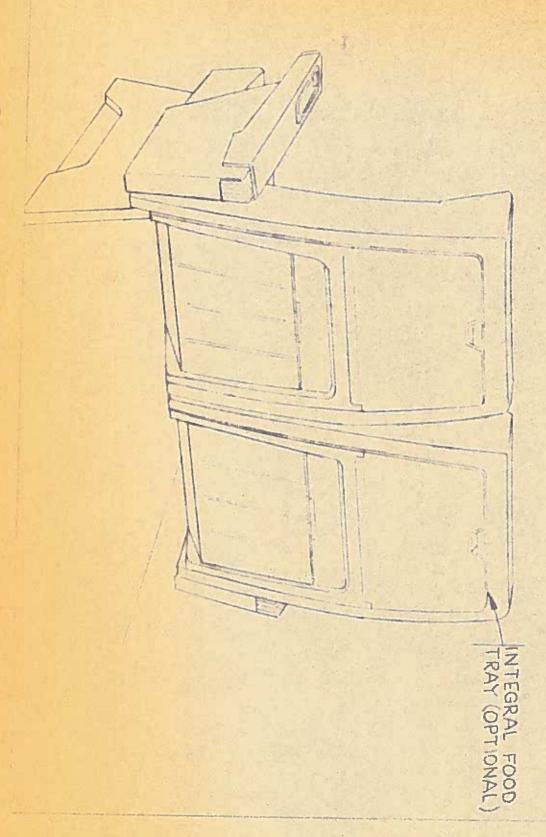
The following parameters shall be used in the coach seat design:

- A. No leg rest mounting provisions
- B. No foot rest.
- C. No stewardess step
- D. No crash energy absorber
- E. No muslin sub-covers
- F. No adjustable head rest
- G. Folding food trays will not be interchangeable with standard passenger seat food trays.
- H. 16 cz upholstery fabric
- I. Literature pockets installed
- J. Ash trays installed
- K. 38-degree maximum recline
- L. Entire construction essentially in accordance with the first class seat except for seat bottom, back and arm rest widths. Seat backs to be same height as standard seats
- M. Seat fittings to match existing standard seat airplane attach points
- N Triple seats to be approximately 64-1/2 inches wide overall and double seats to be approximately 44-1/2 inches wide overall.
- O. Stowage space under each seat shall have a minimum height of ten inches
- P. Hydraulic or mechanical recline locks and wide arms will be available. Weights herein reflect mechanical locks and standard arms.

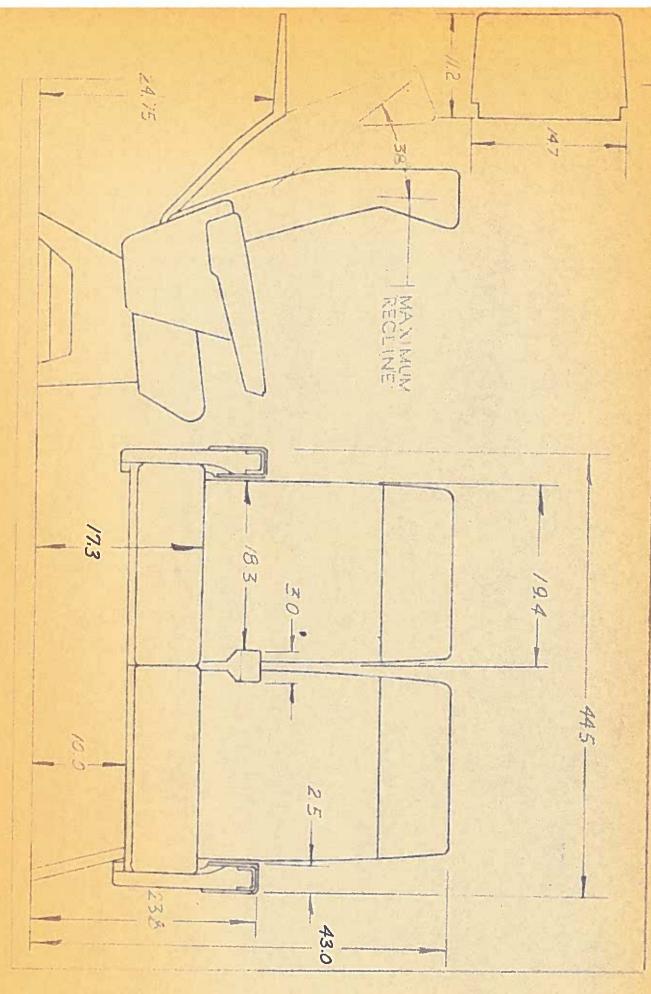
Kits for conversion of 12-place club area are covered in CCP No. 162A



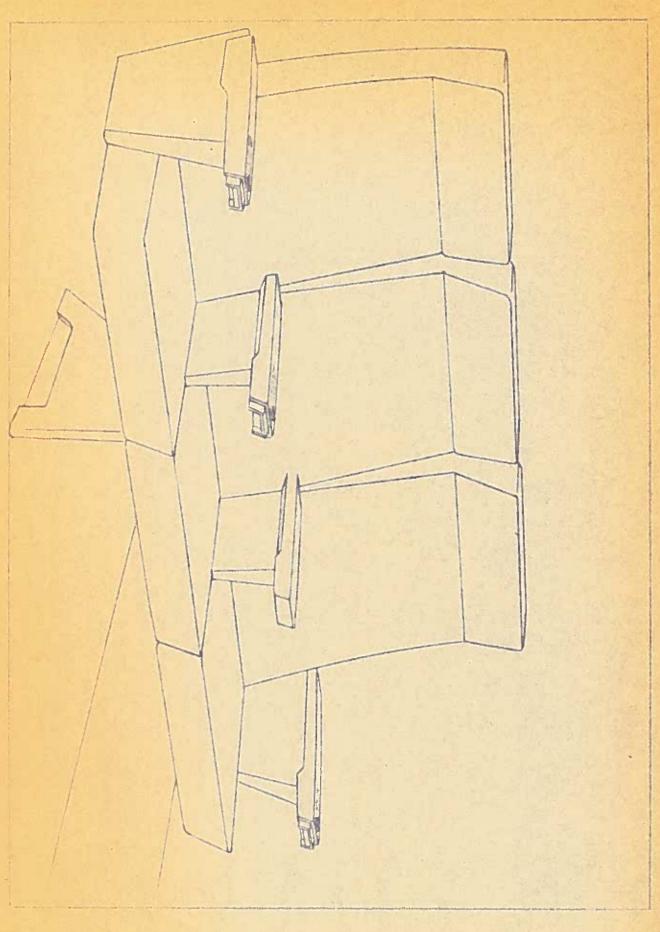
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CONVAR BEG DOWBLE COACH SEA

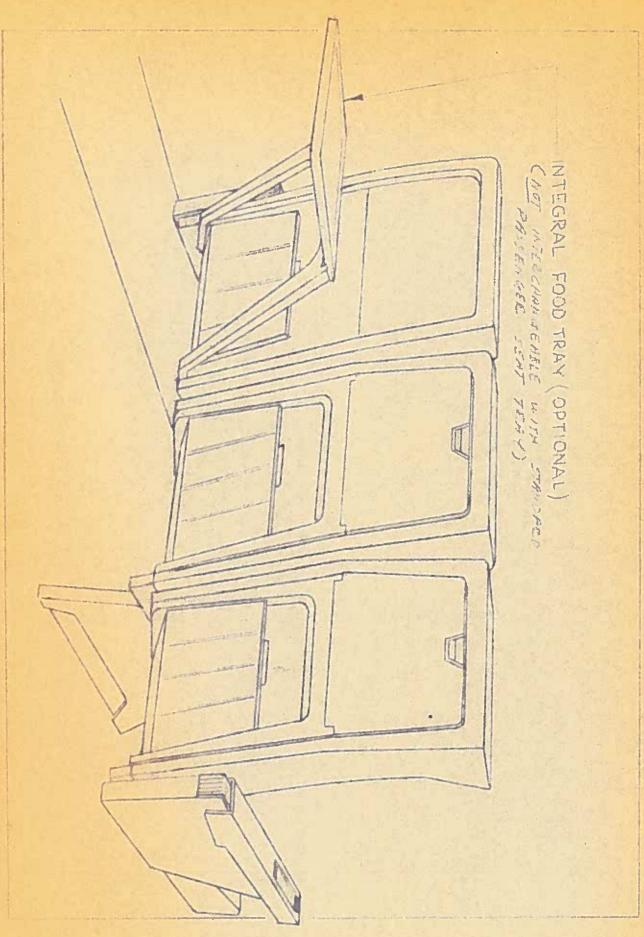


CONVAIR 650 DOUBLE COACH SEAT

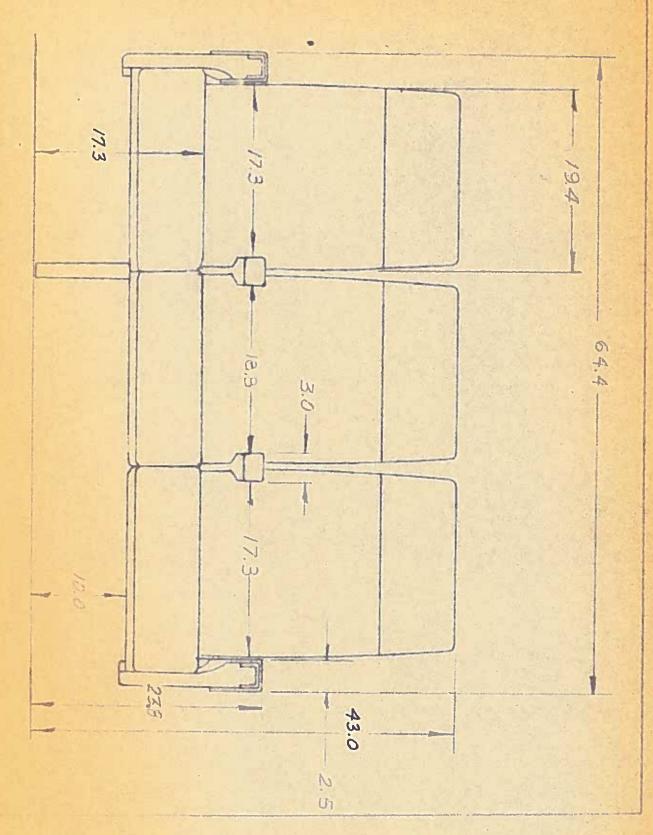


CONVAIR 880 TRIPLE COACH SEAT

HT 1 0F 3



CONVAIR 880 TRIPLE COACH SEAT



CONVAIR 880 TRIBLE COACH SEAT

SPEC NO: ZD-22-003		DATE:			
CUSTOMER: Delta Air Lines Inc.		MCL 10,250 Dtd_	MCL 10,250 Dtd 2 May 1958		
CHANGE NO: 153		MODEL: (880) 22-2			
FITHE: Furn and H	eank Indicator, Po	er Failure Warning of			
ORIGIN: Delta requirements on 2	ested during meet 3, 24 and 25 Apri	ng of Delta and Convai	r representa-		
REASON FOR CHANGE:	Customer requeste	· Cancelled			
	EFFECT ON WEIGH	* EFFECT ON	BALANCE *		
Guar. Wt. Empty	Oper. Wt. Empty				
+5.0 lbs	+5.0 lbs	+982	Inch Lb.		
EFFECT ON GUARANTEE	D PERFORMANCE: *	HR I I			
		one			
* Megligible Change lotals Reflected	es Will be Accumulin a Future Chang	ted and Proposal			
Acceptance of thi Prior Acceptance	s Change is Deper of the Following	lent Upon hanges:			
		Engineer	ing Approval		
LATUST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:					
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:			
ACCEPTED:		ONVAIR, A Div. of Gen.	Dyn. Corp.		
87: HA. 22:		BY: Chief of Contract Administration Commercial			

Delta Air Lines, Inc. Change No. 153

CUNVALUE SD

Title: Turn and Bank Indicator, Power Failure Warning of

Delta requested during meeting between Delta and Convair representatives on 23, 24 and 25 April 1958. Origin:

Reason for Change: Customer requested.

Description of Change:

Figure 3.14-1 PILOT AND CO-PILOT INSTRUMENT PANEL:

Revise above figure to show flag warning annunciators.

Effect on Weight Empty: +5.0 pounds

Effect on Balance: +982 inch-pounds

Effect on Performance: None

The following is not to appear in Specification language:

This change includes installation of two power failure relays in the nose section which are connected to the turn and bank indicators and operate flag warning annunciators on the pilots' panels. The annunciators to be similar to PENN KEYSTONE CORP. "SA 1202-266-RL3H except for the nomenclature display.

SPEC NO: ZD-22-00		DATE:		
CUSTOMER: Delta Air Lines Inc.		MCLDtd		
CHANGE NO: 152		MODEL: (880) 22-2		
TITLE: Speci:	fication Administrati	ve Change		
THASON FOR CHANGE: To make agreeable with approved cockpit mock-up.				
	EFFECT ON WEIGHT *	EFFECT ON BALANCE *		
Guar. Wt. Empty	Oper. Wt. Empty			
0	0	OInch Lb.		
EFFECT ON GUARANTEI		ione		
Totals Reflected	es Will be Accumulate in a Future Change P is Change is Dependen of the Following Cha	t Upon		
LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:				
APACIAD PROVIDIONO.		ECT ON PRICE PER AIRPLANE: urring: -Recurring al:		
ACCEPTED:	CON	WAIR, A Div. of Gen. Dyn. Corp.		
DATE:		Chief of Contract Administration Commercial		

Delta Air Lines Inc. Change No. 152

CONVAIR: SD

litle: Specification Administrative Change

Origin: Convair initiated.

Meason for Change: To make agreeable with approved cockpit mock-up.

Description of Change:

Page 109, Paragraph 3.19.2.8.2 CHECK-OFF LIST:

Delete the words "roll-type" in first line.

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

OPEC NO: ZD-22-00	3	DATE:
CUSTOMER: Delta Air	Lines Inc.	MCL 10,227 Dtd
CHANGE NO: 151		MODEL: (880) 22-2
Supplement	tary Fluorescent Li	ghting on Pilots' Panels
REASON FOR CHANGE:		ile 142, dated 22 April 1958
	EFFECT ON WEIGHT	* EFFECT ON BALANCE *
Guar. Wt. Empty	Oper. Wt. Empty	
+4.0 lbs	+4.0 lbs	Negligiblanch Lb.
FFFECT ON GUARANTEE	D PERFORMANCE: *	None
Acceptance of thi	es Will be Accumula in a Future Change s Change is Depend of the Following C	Proposal ent Upon
		Engineering Approval
DATEST DATE OF ACCE	PTANCE: A	IRPLANES AFFECTED:
Recu Non-		FFECT ON PRICE PER AIRPLANE: ecurring: on-Recurring otal:
ACCEPTED:	C	ONVAIR, A Div. of Gen. Dyn. Corp.
BY:BY:		Y: Chief of Contract Administration Commercial

CONVAIR: SD

Title: Supplementary Fluorescent Lighting on Pilots' Panels

Origin: Delta requested by letter, File 142, dated 22 April 1958

Reason for Change: Customer requested.

Description of Change:

Page 75, Paragraph 3.16.6.2.2 INSTRUMENT PANULS: (Lighting)

Add the following sentences to the end of paragraph:

"In addition, high intensity white fluorescent lighting shall be installed on the glare shield to provide increased lighting for the pilot's and co-pilot's panels. Control of these lights shall be by a switch within the white incandescent flood light rheostat, so arranged that when rheostat is turned to maximum position, the instant-starting fluorescent lights will illuminate and supplement the normal white incandescent lighting."

Lffect on Weight Empty: +4.0 pounds Lffect on Balance: Negligible

Effect on Performance: None

SPEC NO: ZD-22-003		DATE:
CUSTOMER: Delta Air	Lines Inc.	MCL 10,228 Dtd 18 April 1958
CHANGE NO: 150		MODEL: (880) 22-2
TITLE: Increase	f Maximum Landing V	Weight
ORIGIN: TWA reques	ted and Convair pro	oposed for Delta
REASON FOR CHANGE:	Product improvement	ent.
	EFFECT ON WEIGHT	* EFFECT ON BALANCE *
Guar. Wt. Empty	Oper. Wt. Empty	
+80.0 lbs	+80.0 lbs	+70,320 Inch Lb.
EFFECT ON GUARANTEE		
	As	noted.
* Negligible Change Totals Reflected	es Will be Accumula in a Future Change	ted and Proposal
Acceptance of the Prior Acceptance	is Change is Depend of the Following C	ent Upon hanges:
		Engineering Approval
LATEST DATE OF ACC	EPTANCE: A	IRPLANES AFFECTED:
SPECIAL PROVISIONS	F	EFFECT ON PRICE PER AIRPLANE: Recurring: Yon-Recurring Total:
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.
BY: DATE:		Chief of Contract Administration Commercial

CONVAIR: SD

Title: Increase of Maximum Landing Weight

Origin: TWA requested and Convair proposed for Delta

Reason for Change: Product improvement.

Description of Change:

Page 10, Paragraph 3.1.1.1 GUARANTEED PERFORMANCE:

Change the fifth item on the page to read as follows:

"CAA landing distance (intended destination) at sea level over 50 ft obstacle in accordance with CAR 4b with weight of 132,800 lb ft +5% 6,000"

Change the maximum allowable landing weight, of bottom item on page as follows:

From: "130,000 lbs"

To: "132,800 lbs"

Page 17, Paragraph 3.4.1 STRENGTH:

Change the design condition of the fourth item as follows:

From: "Maximum landing weight 130,000"

To: "Maximum landing weight (lb) 132,800"

Effect on Weight Empty: +80.0 pounds

Effect on Balance: +70,320 inch-pounds

Effect on Performance: None

SPEC NO: ZD-22-00	3	DATE:	
CUSTOMER: Delta Air Lines Inc.		MCL 10,224 Dtd 14 April 1958	
CHANGE NO: 149		MODEL: (880) 22- 2	
TITLE: Landing F.	lares, Deletion o	ſ	
ORIGIN: TWA reques	sted and Convair	proposed for Delta	
REASON FOR CHANGE:	As above		
	EFFECT ON WEIG	HT * EFFECT ON BALANCE *	
Guar. Wt. Empty	Oper. Wt. Empt	y a	
-37.0 lbs	-37.0 lbs	-57,572 Inch Lb.	
EFFECT ON GUARANTE	ED PERFORMANCE: *		
		None	
* Negligible Change Totals Reflected			
Acceptance of the Prior Acceptance			
		Engineering Approval	
LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:	
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.	
BY: DATE:		BY: Chief of Contract Administration Commercial	

Delta Air Lines Inc. Change No. 149

CONVAIR: SD

Title: Landing Flares, Deletion of

Origin: TWA requested and Convair proposed for Delta

Reason for Change: As above.

Description of Change:

Page A-16, APPENDIX I-C, PYROTECHNICS:

Delete the following item from the Description List:

"2 Flares

37.0 lb

Kilgore Mfg. Wiley Type SA8"

Effect on Weight Empty: -37.0 pounds
Effect on Balance: -57,572 inch-pounds

Effect on Performance: None

SPEC NO: ZD-22-003		DATE:		
CUSTOMER: Delta Air Lines Inc.		MCL 10,221 Dtd		
CHANGE NO: 148		MODEL: (880) 22-2		
TITLE: Increase of	Maximum Zero Fuel	Weight and Maximum Landing Weight		
ORIGIN: Convair ini	tieted.			
MEASON FOR CHANGE:	Product improveme	nt.		
	EFFECT ON WEIGHT	T * EFFECT ON BALANCE *		
Guar. Wt. Empty	Oper. Wt. Empty			
44:0.0 lbs	+480.0 lbs	+409,152 Inch Lb.		
EFFECT ON GUARANTE	ED PERFORMANCE: *			
		As noted.		
* Negligible Change Totals Reflected	es Will be Accumulation a Future Chang			
Acceptance of the Prior Acceptance	is Change is Depend of the Following (dent Upon Changes:		
		Engineering Approval		
LATEST DATE OF ACCI	CPTANCE:	AIRPLANES AFFECTED:		
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:		
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.		
BY:		BY: Chief of Contract Administration		
DATE:		Commercial Commercial		

Delta Air Lines Inc. Change No. 148

CONVAIR: SD

Title: Increase of Maximum Zero Fuel Weight and Maximum Landing Weight

origin: Convair initiated.

Reason for Change: Product improvement.

Description of Change:

Page 10, Paragraph 3.1.1.1 GUARANTEED PERFORMANCE:

Change the maximum allowable landing weight, of bottom item on page as follows:

From: "130,000 lbs"

To: "132,800 lbs"

Page 17, Paragraph 3.4.1 STRENGTH:

Change the design conditions of the third and fourth items as follows:

From: "Maximum zero fuel weight (1b) 113,000"

To: ""Maximum zero fuel weight (lb) 117,000"

From: "Maximum landing weight 130,000"

To: "Maximum landing weight (lb) 132,800"

Effect on Weight Empty: +480.0 pounds

Effect on Balance: +409,152 inch-pounds

ffect on Performance: As noted.

BFEC NO: ZD-22-00		DR115:				
CUSTOMER: Delta Air Lines		NCL 10,209 D	Rev. td 27 March 1958			
CHANGE NO: 147		MODEL: (880) 22-2				
TITLE: Turn and Bank	Indicator, Revision to					
ORIGIN: TWA requeste	d and Convair proposed for	Delta.				
	Fo provide a single phase indicator.	indicator in lieu	of a 3-phase			
	inarcator.	•				
	EFFECT ON WEIGHT *	EFFEC	F ON BALANCE *			
Guar. Wt. Empty	Oper. Wt. Empty					
-1.0 lb.	-1.0 lb.	negligible	Inch Lb.			
EFFECT ON GUARANTEE	DD PERFORMANCE: *					
	NONE					
* Negligible Change Totals Reflected	es Will be Accumulated in a Future Change Pr	oposal				
Acceptance of thi Prior Acceptance	s Change is Dependent of the Following Char	Upon nges:				
		Engi	neering Approval			
LATEST DATE OF ACCE	EPTANCE: AIRF	PLAMES AFFECTED:				
SPECIAL PROVISIONS	Reci Non-	CCT ON PRICE PER erring: Recurring				
ACCEPTED:	CON	/AIR, A Div. of	Gen. Dyn. Corp.			
BY:			t Administration rcial			

Delta Air Lines Inc.
Change No. 147

TITLE: Turn and Bank Indicator, Revision to

ORIGIN: TWA requested and Convair proposed for Delta.

REASON FOR CHANGE: To provide a single phase indicator in lieu of a 3-phase

indicator.

DESCRIPTION OF CHANGE:

Page A-7, Appendix I-C, Instruments and Related Equipment.

Change the 16th item as follows:

From: "2 Turn and Bank Indicators (electric)

Pioneer Central 3920-1AD-A1-1"

To: "2 Turn and Bank Indicators (electric)

Pioneer Central 3924-1AG-B-1"

Effect on Weight Empty: -1.0 lb. Effect on Balance: Negligible

Effect on Performance: None

SPEC NO: ZD-22-00.3		DATE:			
CUSTOMER: Delta Air Lines Inc.			MCL 10,199	Dtd 21	. February 19
CHANGE NO: 146			MODEL: (880) 22-2	
TITLE: High Inten	sity, Oscillating	g Posit	ion Lights,	Install	ation of
ORIGIN: Delta requ			.42, dated 18		ry 1958
	EFFECT ON WEIG		EFF	ECT ON	BALANCE *
Guar. Wt. Empty	Oper. Wt. Empt	у			
+35.0 lb	+35.0 lb		+4	7,174	Inch Lb.
EFFECT ON GUARANTE		one	10		
* Negligible Change Totals Reflected Acceptance of the Prior Acceptance	in a Future Char is Change is Depe	nge Pro endent	posal. Upon		
				gineeri	ng Approval
LATEST DATE OF ACCI	EPTANCE:	AIRPL	ANES AFFECTE	D:	
disapproves installation, another R proposal will be required to rein- N		EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:			
ACCEPTED:		CONVA	IR, A Div. o	f Gen.	Dyn: Corp.
BY:		BY:	Programme in Alexandria		
DATE:		Chief of Contract Administration Commercial			

Page 1 of 2

Title: High Intensity, Oscillating Position Lights, Installation of

Origin: Delta requested by Letter, File 142, dated 18 February 1958

Reason for Change: Customer requested.

Description of Change:

Page 61, Paragraph 3.14.3.4.1 FLUX VALVE:

Levise the first sentence to read as follows:

"Two flux valve units shall be installed just inboard of the right wing tip in the trailing edge area, with two access doors on the upper wing surface."

Page 74, Paragraph 3.16.8.1.1 POSITION LIGHTS:

Revise the paragraph to read as follows:

"Three high intensity, oscillating position lights shall be installed; one red light on the forward left wing tip, one green light on the forward right wing tip and one white light on the aft end of the fuselage. Those lights shall be designed to oscillate between an arc of approximately 126 degrees in plan view and to embrace an arc of approximately 80 degrees in the vertical plane."

Page 74, Pargraph 3.16.8.1.4 AFFI-COLLISION LIGHTS:

Add the following after the paragraph title: "(Not applicable)", and delete this paragraph.

Page 75, Paragraph 3.16.8.1.4.1 CONTROL:

Delete the paragraph including paragraph number and title.

Page 85, Paragraph 3.16.17 ESSENTIAL POWER:

Under "Lighting", delete the sixth item "(Anti-Collision Lights)", and renumber the remaining items 1 through 10.

Page 2 of 2

Page A-2, APPENDIX I-C, ELECTRICAL EQUIPMENT:

Under "EXTERIOR LIGHTS", change the first two items as follows:

From: "1 Wing Tip Lamp Assembly (left hand)"

To: "1 Position Light Assembly, Wing, Oscillating, Red (left hand)"

From: "1 Wing Tip Lamp Assembly (right hand)"

Position Light Assembly, Wing, Oscillating, Green (right hand)"

Delete the fourth item "(2 Position Lamps)"

Delete the fifth item "(1 Tail Light, Red)"

Change the sixth item as follows:

From: "1 Tail Light White"

To: "1 Position Light Assembly, Aft, Oscillating, White"

Delete the ninth item "(2 Rotating Anti-Collision Light Assemblies)"

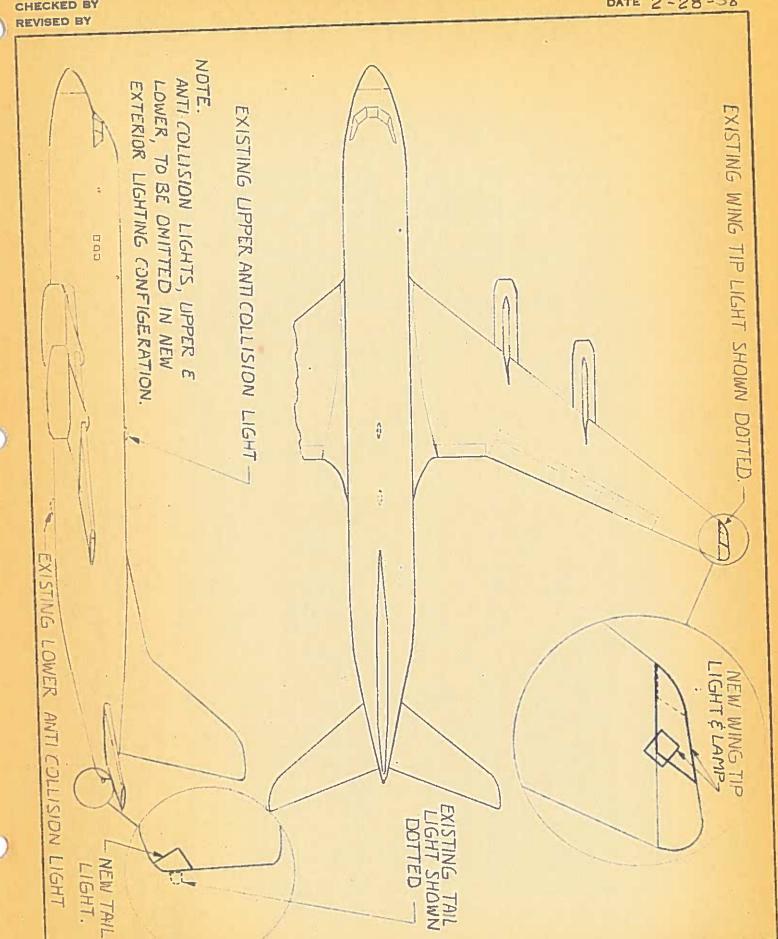
Enclosure: (A) One (1) copy of sketch on proposed position light locations (for information only).

10

C O N V A A DIVISION OF GENERAL DYNAMICS CORPORATION BAN DIEGO

PAGE REPORT NO.

DATE 2-28-58



SPEC NO: ZD-22-003		DATE	:	
CUSTOMER: Delta Air Lines Inc.		MCL_	Dtd	
CHANGE NO: 145A		MODE	L: (880) 22- 2	
TITLE: Specificat	ion Administrati	ve Change		
ORIGIN: Convair in	itiated.			
REASON FOR CHANGE:	To clarify the	intent of t	the Specification.	
			Cancellad	
	EFFECT ON WEI		EFFECT ON BALANCE *	
Guar. Wt. Empty	Oper. Wt. Emp	cy		
0	0		O Inch Lb.	
EFFECT ON GUARANTEI		*		
		None		
* Negligible Change Totals Reflected	es Will be Accum in a Future Cha	lated and nge Proposa	1	
Acceptance of the Prior Acceptance	is Change is Dep of the Followin	endent Upon Changes:		
			Engineering Approval	
LATEST DATE OF ACCE	EPTANCE:	AIRPLANES	AFFECTED:	
			PRICE PER AIRPLANE:	
supersedes CCP No. 64.		Non-Recur	ecurring: on-Recurring otal:	
ACCEPTED:		CONVAIR,	A Div. of Gen. Dyn. Corp.	
BY:		BY:	of Contract Administration	
DATE:		Chief	of Contract Administration Commercial	

Delta Air Lines Inc. Change No. 145A

CONVAIR: SD

Title: Specification Administrative Change

Grigin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Page 80, Paragraph 3.16.11.1 LANDING GEAR WARNING HORN:

add (To CeP 64) And to the top to read as follows:

"An electrical interconnect between flaps, landing gear scissors switch and throttles shall be provided. This interconnect will sound the warning horn when any two throttles are advanced to take-off position and the flaps and elevator trim are not in the take-off regime."

Effect on Weight Empty: O Effect on Balance: O Effect on Performance: None

Delta Air Lines Inc. Change No. 145A

CONVAIR: SD

EXHIBIT "A"

The following shall not appear in the Specification language:

"The language contained in this Change was originally proposed in CCP No. 64 as a revision to paragraph 3.10.2.1. Paragraph 3.10.2.1 was revised by CCP No. 1 which deleted reference to the subject of this CCP. Therefore, to incorporate this subject into the Specification the requirement is being covered by this Change under Paragraph 3.16.11.1."

SPEC NO: ZD-22-003		DATE:		
CUSTOMER: Delta Air Lines Inc.		MCL 10.222 Dtd		
CHANGE NO: 144		MODEL: (880) 22- 2		
TITLE: Passenger Cabin Oxygen Cylinders, Relication of				
ORIGIN: TWA requested and Convair proposed for Delta				
REASON FOR CHANGE: To maintain uniform stowage location of portable oxygen bottles on Convair 880 airplanes.				
	EFFECT ON WEIG	HT * EFFECT ON BALANCE *		
Guar. Wt. Empty	Oper. Wt. Empt	y		
0	0	Negligible Inch Lb.		
EFFECT ON GUARANTEED PERFORMANCE: *				
None				
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal				
Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:				
		Engineering Approval		
LATEST DATE OF ACCE	PTANCE:	AIRPLANES AFFECTED:		
SPECIAL PROVISIONS: Requires acceptance TWA and Delta.	by	EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:		
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.		
BY:		BY:		
DATE:		Chief of Contract Administration Commercial		

CONVAIR: SD Delta Air Lines Inc.
Change No. 144

Title: Passenger Cabin Oxygen Cylinders, Relocation of

Origin: TWA requested and Convair proposed for Delta

Reason for Change: To maintain uniform stowage location of portable

oxygen bottles on Convair 880 airplanes.

Description of Change:

Page 114, Paragraph 3.19.5.6 PORTABLE BOTTLES:

Add the following sentence to the paragraph:

"The bottles shall be located as follows: one in the forward left hand hatrack stowage bin, one in the left hand mid-cabin stowage bin and one in the aft left hand stowage bin."

Effect on Weight Empty: 0

Effect on Balance: Negligible

Effect on Performance: None

C () N V A I R A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-003		DATE:		
CUSTOMER: Delta Air Lines Inc.		MCL 10,211 Dtd		
CHANGE NO: 143		MODEL: (880) 22-2		
TITLE: Portable Oxygen Cylinder Assemblies, Revision to				
ORIGIN: Convair initiated.				
REASON FOR CHANGE: To provide versatile portable oxygen cylinder assemblies in the passenger cabin area.				
EFFECT ON WEIGHT * EFFECT ON BALANCE *				
Guar. Wt. Empty	Oper. Wt. Empty			
0	-12.0 lb	-11,296 Inch Lb.		
EFFECT ON GUARANTEED PERFORMANCE: *				
None				
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal				
Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:				
		Engineering Approval		
LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:				
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:		
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.		
BY: DATE:		BY: Chief of Contract Administration Commercial		

CONVAIR: SD

Title: Portable Oxygen Cylinder Assemblies, Revision to

Origin: Convair initiated.

Reason for Change: To provide versatile portable oxygen cylinder

assemblies in the passenger cabin area.

Description of Change:

Page 114, Paragraph 3.19.5.6 PORTABLE BOTTLES:

Revise the third sentence to read as follows:

"Three 310-liter portable oxygen bottles, each with continuous flow regulators and two Puritan continuous flow mask assemblies, shall be provided in the passenger compartment."

Page A-15, APPENDIX I-C, OXYGEN EQUIPMENT:

Change the fifth item in the Description List as follows:

From: "2 Air-Pac-Portable (Crew Compartment)

10: "1 Portable 02 Cylinder Assembly Scott 6000B1-0 (310-Liter) (Crew Compartment)

Change the sixth item in the Description List as follows:

From:	ušt	Portable High Pressure Oxygen Bottles (310-liter)	Scott Mfg. Company	Air Pack Assy. No. 6000Bl-0
To:	"3	Portable O2 Cylinder Assemblies (310-Liter)	Scott	5500-3BB"

Change the eighth item as follows:

From: "2 pr. Asbestos Gloves"

To: "l pr. Asbestos Gloves"

696-141	V-V-V-V-V-V-V-V-V-V-V-V-V-V-V-V-V-V-V-	Weight (1b)	Moment (inch-lb)
Lilect	on Weight Empty:	0.0	0.0
Effect	on Fixed Useful Loa	d: -12.0	-11,296
Lifect	on Operating Weight	Empty: -12.0	-11,296
Effect	on Performance:	None	

NOTE: The effects of CCP No. 30 are included in this Change.

A Division of General Dynamics Corporation San Diego, California

		DATE:		
CUSTOMER: Delta Air	Lines Inc.	MCL 10,196 Dtd 17 February 19		
CHANGE NO: 142		MODEL: (880) 22- 2		
TITLE: Lavatory To	owel Dispenser, Add:	ition of		
ORIGIN: FWA request	ed and Convair prop	posed for Delta		
REASON FOR CHANGE: To provide an additional towel dispensing facility in each lavatory.				
	EFFECT ON WEIGHT	* EFFECT ON BALANCE *		
Guar. Wt. Empty	Oper. Wt. Empty			
Aegligible	Negligible	Negligible Inch Lb.		
EFFECT ON GUARANTEED PERFORMANCE: *				
None				
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal				
Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:				
Prior Acceptance	of the Pollowing o.			
Prior Acceptance	of the following of	Engineering Approval		
Prior Acceptance				
	EPTANCE: A by both R N	Engineering Approval		
LATEST DATE OF ACCIONS SPECIAL PROVISIONS Requires acceptance TWA and Delta. ACCEPTED:	EPTANCE: A by both R N T	Engineering Approval IRPLANES AFFECTED: FFECT ON PRICE PER AIRPLANE: ecurring: on-Recurring		
SPECIAL PROVISIONS Requires acceptance TWA and Delta.	EPTANCE: A E by both R N T	Engineering Approval IRPLANES AFFECTED: FFECT ON PRICE PER AIRPLANE: ecurring: on-Recurring otal:		

Title: Lavatory Towel Dispenser, Addition of

Origin: TWA requested and Convair proposed for Delta

Reason for Change: To provide an additional towel dispensing facility

in each lavatory.

Description of Change:

Page 107, Paragraph 3.19.2.2.3 MISCELLANEOUS LAVATORY EQUIPMENT:

Change the eighth item in the equipment list as follows:

From: "Two towel dispensers (one for linen and one for paper)"

To: "One towel dispenser unit (consisting of three dispensers; two universal for either linen or paper, and one for paper

only)"

Page A-12, APPENDIX I-C, FURNISHINGS:

Change the 17th item in the Description List as follows:

From: "3 Linen Towel Dispensers"

To "3 Towel Dispenser Units (each consisting of three dispensers; two universal for either linen or paper, and one for paper only)"

Effect on Weight Empty: Negligible Effect on Balance: Negligible Effect on Performance: None

A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-00	3	DATE:			
CUSTOMER: Delta A	ir Lines Inc.	MCI, 10,210 Dtd 26 March 1958			
CHANGE NO: 141		MODEL: (880) 22-2			
TITLE: Lavatory W. Resistant		nter Top, Change to Corrosion			
ORIGIN:	ORIGIN:				
REASON FOR CHANGE:		apad			
		Cancelled			
	EFFECT ON WEIG				
Guar. Wt. Empty	Oper. Wt. Empt	y and the second se			
+14.0 16	+14.0 lb	+14,000 Inch Lb.			
EFFECT ON GUARANTEED PERFORMANCE: *					
None					
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal					
Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:					
Engineering Approval					
LATEST DATE OF ACCI	EPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS		EFFECT ON PRICE PER AIRPLANE: Recurring:			
		Non-Recurring			
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.			
BY:		BY: Chief of Contract Administration			
DATE:		Commercial			

CONVAIR: SD

Delta Air Lines Inc. Change No. 141

Title: Lavatory Wash Basin and Counter Top, Change to Corrosion Resistant Steel

Origin:

Reason for Change:

Description of Change: .

Page 107, Paragraph 3.19.2.2 LAVATORIES:

Revise the third sentence, starting in the seventh line, to read as follows:

"A .032 corrosion resistant steel wash basin, with 3/4 inch diameter drain, a manually-operated drain stopper with easily replaceable seal, one each hot water and cold water household type springloaded faucets so designed that passengers may wash their hands with running water, shall be provided in each lavatory. The adjacent counter top shall be .025 corrosion resistant steel."

Effect on Weight Empty: +14.0 pounds

Effect on Balance: +14,000 inch-pounds

Effect on Performance: None

The following is not to appear in the Specification language:

The effects of CCP No. 115 (Lavatory Hot Water Provisions) have been included in the change to the above paragraph.

CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DILGO, CALIFORNIA

SPEC NO.: ZD- 22-003	DATE:
CUSTOMER: Delta Air Lines Inc.	MCL 10,207 DTD 24 February 19
CHANGE NO: 140A	MODEL: (880) 22-1
TITLE: Stabilizer Trim Control, poration of	Electric Drive Capability Incor-
ORIGIN: Delta verbal request of 1	.9 February 1959.
REASON FOR CHANGE: Customer request,	and revision to CCP No. 140.
EFFECT ON WEIGHT *	EFFECT ON BALANCE *
GUAR. WT. EMPTY OPER. WT. EMPTY	
/20.0 lb /20.0 lb	,4+,586 INCH LB.
EFFECT ON GUARANTEED PERFORMANCE: " Non	ne
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AT TOTALS REFLECTED IN A FUTURE CHANGE PROPACE OF THIS CHANGE IS DEPENDENT UP PRIOR ACCEPTANCE OF THE FOLLOWING CHANGE	POSAL
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:
	RECURRING: NON-RECURRING: TOTAL:
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.
BY:	
DATE:	TOWVALL, SHIPETA

C O N V A I R A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-00	5	DATE:		
CUSTOMER: Delta Ai	r Lines Inc.	MCL 10,193 Dtd 11 February 1958		
CHANGE NO: 139		MODEL: (880) 22-2		
TITLE: Approach S	peed Indicator, Instal	lation of		
ORIGIN: TWA requested and Convair proposed for Delta REASON FOR CHANGE: As above Rejected				
	EFFECT ON WEIGHT *	EFFECT ON BALANCE *		
Guar. Wt. Empty	Oper. Wt. Empty			
+17.0 lb	+17.0 lb	+8,994 Inch Lb.		
EFFECT ON GUARANTEED PERFORMANCE: * None				
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal				
Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:				
Engineering Approval				
LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:				
SPECIAL PROVISIONS: EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:				
ACCEPTED:	CON	AIR, A Div. of Gen. Dyn. Corp.		
BY:DATE:		hief of Contract Administration Commercial		

CONVAIA: SD

Title: Approach Speed Indicator, Installation of

Origin: TWA requested and Convair proposed for Delta

Reason for Change: As above.

Description of Change:

Page 62. Paragraph 3.14 INSTRUMENTS AND NAVIGATION EQUIPMENT:

Add the following new paragraph to the page:

"3.14.3.5.1 APPROACH SPEED INDICATING SYSTEM: An approach speed indicating system shall be installed. The system shall consist of a lift transducer assembly in the wing leading edge, a flap position potentiometer in the flap gear box, a lift computer in the electrical rack, and two speed control indicators; one on the pilot's glare shield and one on the copilot's glare shield."

Effect on Weight Empty: +17.0 pounds

Effect on Balance: +8,994 inch-pounds

Effect on Performance: None

The following is not to appear in the Specification language:

"The Approach Speed Indicating System shall be fabricated by SAFE FLIGHT INSTRUMENT CORPORATION."

C O N V A I R A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-003		DATE:		
CUSTOMER: Delta Air	Lines Inc.	MCI, 10,206 Dtd 17 March	1958	
CHANGE NO: 138		MODEL: (880) 22-2		
TIPLE: Cabin Interior Modification and Installation of a 12-Place Club Area.				
ORIGIN: References: Letter of Agreement No. 21, Delta Letter, File 142, dated 11 March 1957, and Interior Mock-up of 11 June 1957 To furnish Customer with desired cabin interior and to provide a 12-place club area.				
	EFFECT ON WEIGHT *	EFFECT ON BALANCE	*	
Guar. Wt. Empty	Oper. Wt. Empty			
+351.8 lbs	+351.8 lbs	+236,831 Inch L	b.	
EFFECT ON GUARANTEED PERFORMANCE: * None				
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal				
Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:				
		Engineering Appro	oval	
LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:				
SPECIAL PROVISIONS:	Recu Non-	CT ON PRICE PER AIRPLANE: rring: Recurring		
ACCEPTED:	CON	VAIR, A Div. of Gen. Dyn. Co.	rp.	
BY: DATE:		hief of Contract Administra Commercial	tion	

Page 1 of 5

Title: Cabin Interior Modification and Installation of a 12-Place Club Area.

Origin: References: Letter of Agreement No. 21, Delta Letter, File 142, dated 11 March 1957, and Interior Mock-up of 11

June 1957.

Reason for Change: To furnish Customer with desired cabin interior and

to provide a 12-place club area.

Description of Change:

Page v INTRODUCTION:

In eighth line of first paragraph, change "80 to 108" to "84 to 107".

Page 1, Paragraph 1.1 BASIC TYPE:

In fifth line of first paragraph, change "80" to "84", and in seventh line, change "108" to "107".

Page 12, Paragraph 3.1.2.2 LOADING SUMMARY:

Under "Payload", change "(80 passengers plus 4 lounge)" to "(84 passengers including 12-place club area)".

Page 12, Paragraph 3.1.2.2.1 PAYLOAD: Coach Configuration:

Change "(108 passengers)

26,450 lb"

To: "(107 passengers)

26,285 1b"

Page 78, Paragraph 3.16 ELECTRICAL SYSTEM:

Add the following new paragraph after paragraph 3.16.8.6:

"3.16.8.7 SIDE WINDOW LICHTING: Valance type fluorescent lighting, including a dimming feature, shall be installed at the cabin side windows. A switch shall be included to separate this lighting from the overhead cove lights."

Page 81, Paragraph 3.16.11.4 CALL SYSTEM:

Revise the third sentence to read as follows:

"Call lights to indicate calls as follows: One from the club area (or forward passenger area) and one from the aft passenger area, one from each lavatory, and one from the flight deck shall be installed and shall be visible from the passenger compartment and the cabin attendants' stations."

Page 2 of 5

Page 83, Paragraph 3.16.11.4.1 PASSENGER CALL SWITCHES:

Revise the first sentence to read as follows:

"Call switches shall be installed; two for each row of seats within reach of passengers, one in each lavatory and six in the club area."

Page 103, Paragraph 3.19.1.1 INTERIOR:

Revise the first sentence to read as follows:

"The interior shall include furnishings and equipment for 84 passengers, including 12 club area passengers, plus crew accommodations."

Page 104, Paragraph 3.19.1.1.5 LOUNGE SEATS:

Delete the paragraph title and the first sentence, and substitute the following:

"CLUB AREA SEATS: A 12-place club area shall be provided aft of the forward main entrance door, with seat assemblies arranged as follows:

- (a) Left hand double seat facing aft(b) Right hand double seat facing aft
- (c) Right hand: double seat facing forward
- (d) Left hand quadruple seat assembly
 (e) Right hand double seat facing inboard

Page 104, Paragraph 3.19.1.1.6 PASSENGER SEATS:

Change the words "lounge type" appearing in the third line, and "lounge" appearing in the ninth and the eleventh lines, to "club area".

Fage 105, Paragraph 3.19.1.1.8 CONVERTIBILITY:

Revise the paragraph to read as follows:

"The standard 84-passenger configuration (including 12 club area passengers) shall include full provisions to permit quick conversion to a full range of mixed class interiors utilizing the two coat compartments shown in Figure 1-3 as dividers. These coat compartments shall be designed to utilize the standard floor seat attach

Page 3 of 5

points and thus may be placed in any of the rows between rows six and fifteen inclusive, excluding emergency hatch areas, in the main cabin section. Each of the 22 seat rows shall incorporate standard cabin windows, and each row aft of the club area shall have reading lights and individual air outlets for five abreast seating as shown in the coach configuration in Figure 1-3. All rows shall permit the installation of standard seats or coach seats. Provisions shall be made for installation of a left and right hand hat rack in the club area for use with four and five abreast seating arrangements."

Page 107, Paragraph 3.19.2.3 MISCELLANEOUS LAVATORY EQUIPMENT:

Change the first item in the equipment list to read as follows:

"Three shatterproof mirrors (one mirror in each lavatory)"

Page 109, Paragraph 3.19.2.8.8 MAGAZINE HACKS:

Revise the paragraph to read as follows:

"Four magazine racks of the self-cleaning type shall be installed; one in aft part of the forward entrance area, two in the aft part of aft entrance area and one in the club area."

Page 110, Paragraph 3.19.3.1.1 PASSENGER COMPARTMENT:

Revise the first sentence to read as follows:

"The passenger compartments provide seating accommodations for 84 passengers, including 12 club area passengers".

Add the following to the last sentence in paragraph:

".... except in the areas of lowered ceiling".

Page 111, Paragraph 3.19.3.3 TRIM:

Add the following sentence to the end of paragraph:

"Carpeting shall be provided on side walls of cabin interior extending approximately 10 inches up the walls."

Page 111, Paragraph 3.19.3.5 CURTAINS:

Hevise the first sentence to read as follows:

CONVAIR: SD

Page 4 of 5

"All passenger compartment windows shall be provided with glare control."

Delete the last sentence which read as follows:

"Window curtains, wherever practicable, shall be interchangeable."

Page 11, Paragraph 3.19.3.6 HAT RACKS:

Revise the paragraph to read as follows:

"Overhead hat racks shall be installed extending the full length of the passenger compartment except in the area of the stowage bins. They shall be sufficiently rigid to support passenger walking in the aisle in rough weather. The racks shall be designed for a stowage load of one pound per lineal inch and an additional load of 170 pounds applied at each third row of seats. Stowage bins, to break the continuity of the hat racks, shall be installed in four locations on each side of the aisle. A door, or doors, shall be provided for each bin."

Page 125. Paragraph 3.23.1 EQUIPMENT INTERCHANGEABILITY:

Add under Item 2 - Interchangeability Without Alteration

Double asterisk before **Passenger Seat Assembly LH and RH.

Add the following note to the bottom of the page:

**NOTE: Passenger seats immediately aft of the escape hatches shall have special outboard arm rests, however, they will be interchangeable in all other respects.

Page A2, APPENDIX I-C, ELECTRICAL SYSTEM:

Under "INTERIOR LIGHTS", change the tenth item as follows:

From: "2 Fasten Seat Belts - No Smoking Sign"

To: "4 Fasten Seat Belts - No Smoking Signs"

Page A-12, APPENDIX I-C, FURNISHINGS:

Change the fifth item in Description List as follows:

From: "40 Passenger Seats (Double)"

To: "36 Passenger Seats (Double)"

Page 5 of 5

Change the sixth item as follows:

From: "2 Lounge Seats (Double)"

"4 Club Area Seats (Double)" To:

Add the following item:

"1 Club Area Seat (Quadruple)"

Change the 22nd item as follows:

"3 Wash Basin Mirrors" From:

"3 Lavatory Mirrors (Shatterproof)" To:

Figure 1-2 - INBOARD PROFILE AND PLAN VIEW

Figure 1-3 - ALTERNATE SEATING ARRANGEMENT

Figure 3.7-1 - EMERGENCY EXIT FLOW

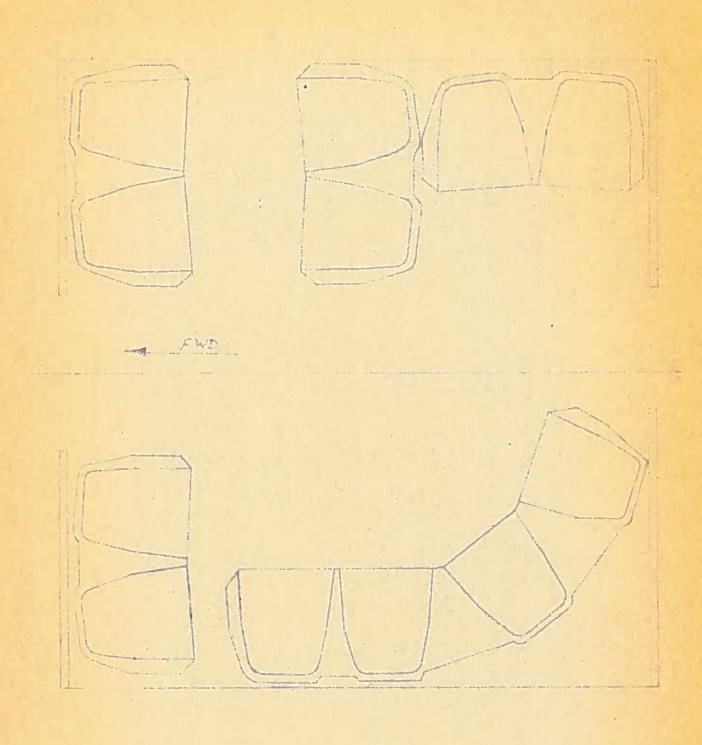
Figure 3.19-2 - AFT WATER SYSTEM SCHEMATIC
Figure 3.19-3 - LAVATORY TOTLET WATER SCHEMATIC
Figure 3.19-4 - HIGH PRESSURE OXYGEN GAS SYSTEM SCHEMATIC

Revise above figures as required to reflect this change.

Enclosure: (A) One (1) copy of sketch of proposed Club Area Arrangement.

Effect on Weight Empty: +351.8 pounds Effect on Balance: +236,831 inch-pounds

Effect on Performance: None



12-PLACE CLUB AREA

C O N V A I R A Division of General Dynamics Corporation San Diego, California

SPEC NO: <u>ZD-22-003</u>			DATE:		
CUSTOMER: Delta Air Lines Inc.			MCL 10,200	Dtd 24	February 195
CHANGE NO: 137			MODEL: (880) 22-	
	ire Extinguisher E ified, Installatio				
ORIGIN: Convair in		C	ncell	ed	
REASON FOR CHANGE:	REASON FOR CHANGE: To delete the Accessory Products solenoid actuated valve fire extinguishing system and install an Accessory Products system utilizing a cartridge type method of puncturing a frangible disc.				
	EFFECT ON WEIGH		EFF	ECT ON 1	BALANCE *
Guar. Wt. Empty	Oper. Wt. Empty				
-2.0 lb	-2.0 lb			-2,338	Inch Lb.
EFFECT ON GUARANTEED PERFORMANCE: *					
	None				
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal					
	is Change is Depen of the Following				
	Engineering Approval				
LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:					
SPECIAL PROVISIONS:			T ON PRICE Pring: Recurring		
ACCEPTED:		CONVA	IR, A Div. o	f Gen. 1	Dyn. Corp.
BY:			m		
DATE:		Ch	ief of Contr	act Adm: mercial	inistration
	100				

CONVAIR: SD

Nacelle Fire Extinguisher Equipment, Accessory Products Com-Title:

pany, Modified, Installation of

Convair initiated. Origin:

To delete the Accessory Products solenoid actuated Reason for Change:

valve fire extinguishing system and install an Accessory Products system utilizing a cartridge type method of puncturing a frangible disc.

Description of Change:

This change has no effect on Specification language.

Effect on Weight Empty:

-2.0 pounds -2,338 inch-pounds Effect on Balance:

None Effect on Performance:

C O N V A I R A Division of General Dynamics Corporation San Diego, California

CUSTOMER: Delta Air Lines Inc. MCL 10,198 Dtd 24 February 1958 CHANGE NO: 136 MODEL: (880) 22-2 TITLE: Nacelle Fire Extinguisher Equipment, Walter Kidde, Installation of ORIGIN: TWA requested and Convair proposed for Delta. REASON FOR CHANGE: To install a Walter Kidde fire extinguishing system in lieu of an Accessory Products Company fire extinguishing system. EFFECT ON WEIGHT * EFFECT ON BALANCE * Guar. Wt. Empty Oper. Wt. Empty +7.0 lb +7.0 lb +5,618 Inch Lb. EFFECT ON GUARANTEED PERFORMANCE: * None * Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:
Nacelle Fire Extinguisher Equipment, Walter Kidde, Installation of ORIGIN: TWA requested and Convair proposed for Delta. REASON FOR CHANGE: To install a Walter Kidde fire extinguishing system in lieu of an Accessory Products Company fire extinguishing system. EFFECT ON WEIGHT * EFFECT ON BALANCE * Guar. Wt. Empty Oper. Wt. Empty +7.0 lb +7.0 lb +5,618 Inch Lb. EFFECT ON GUARANTEED PERFORMANCE: * None * Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal Acceptance of this Change is Dependent Upon
ORIGIN: TWA requested and Convair proposed for Delta. REASON FOR CHANGE: To install a Walter Kidde fire extinguishing system in lieu of an Accessory Products Company fire extinguishing system. EFFECT ON WEIGHT * EFFECT ON BALANCE * Guar. Wt. Empty
REASON FOR CHANGE: To install a Walter Kidde fire extinguishing system in lieu of an Accessory Products Company fire extinguishing system. EFFECT ON WEIGHT * EFFECT ON BALANCE * Guar. Wt. Empty Oper. Wt. Empty +7.0 lb +7.0 lb +5,618 Inch Lb. EFFECT ON GUARANTEED PERFORMANCE: * None * Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal Acceptance of this Change is Dependent Upon
in lieu of an Accessory Products Company fire extinguishing system. EFFECT ON WEIGHT * EFFECT ON BALANCE * Guar. Wt. Empty
Guar. Wt. Empty Oper. Wt. Empty +7.0 lb +7.0 lb +5,618 Inch Lb. EFFECT ON GUARANTEED PERFORMANCE: * None * Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal Acceptance of this Change is Dependent Upon
+7.0 lb +7.0 lb +5,618 Inch Lb. EFFECT ON GUARANTEED PERFORMANCE: * None * Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal Acceptance of this Change is Dependent Upon
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal Acceptance of this Change is Dependent Upon
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal Acceptance of this Change is Dependent Upon
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal Acceptance of this Change is Dependent Upon
Totals Reflected in a Future Change Proposal Acceptance of this Change is Dependent Upon
Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:
Engineering Approval
LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:
SPECIAL PROVISIONS: EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:
ACCEPTED: CONVAIR, A Div. of Gen. Dyn. Corp.
BY:BY:
DATE: Commercial Commercial

CONVAIR: SD

Title: Nacelle Fire Extinguisher Equipment, Walter Kidde, Installa-

tion of

Origin: TWA requested and Convair proposed for Delta.

Reason for Change: To install a Walter Kidde fire extinguishing system

in lieu of an Accessory Products Company fire extin-

guishing system.

Description of Change:

This change has no effect on Specification language.

Effect on Weight Empty: +7.0 pounds

Effect on Balance: +5,618 inch-pounds

Effect on Performance: None

The following is not to appear in Specification language:

This change proposes installation of a "Walter Kidde" fire extinguishing system utlizing a cartridge type method of puncturing a frangible disc (single squib) to release the fire extinguishing agent, in lieu of an "Accessory Products Company" fire extinguishing system using a solenoid actuated valve to release the fire extinguishing agent.

On Customer acceptance of this change, Appendix I-C will be revised accordingly to show manufacturer and equipment items covered herein.

C O M V A I R A Division of General Dynamics Corporation San Diego, California

SPMC NO: ZD-22-003		DATE:		
CUSTOMER: Delta Air Lines Inc. MCL 10,203 Dtd				
CHANGE NO: 135		MODEL: (880) 22-2		
TITLE: Drip Sticks	for Fuel Quantity In	ndication, Installa	ition of	
ORIGIN: Convair ini	tiated			
REASON FOR CHANGE: To provide most suitable underwing fuel level indication for Convair 880 airplanes.				
	EFFECT ON WEIGHT *	EFFECT O	N BALANCE *	
Guar. Wt. Empty	Oper. Wt. Empty O	0	Inch Lb.	
EFFECT ON GUARANTEED PERFORMANCE: * None				
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal				
Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:				
Engineering Approval				
LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:				
SPECIAL PROVISIONS: EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:				
ACCEPTED:	CONV	AIR, A Div. of Gen	. Dyn. Corp.	
BY: DATE:	C	hief of Contract A Commerci	dministration	

CONVAIR: SD

a B. 19

Title: Drip Sticks for Fuel Quantity Indication, Installation of

Origin: Convair initiated

Reason for Change: To provide most suitable underwing fuel level indication for Convair 880 airplanes.

Description of Change:

Page 52, Paragraph 3.12.9.9.2 AUXILIARY FUEL GAGES:

Revise the paragraph to read as follows:

"A total of 16 graduated, tube-type drip sticks shall be installed in the underwing side of the fuel tanks to provide for underwing fuel level indication. The drip sticks shall be installed as follows:

2 in each main fuel tank 2 in each replenishing fuel tank

The drip sticks shall be located in such a manner as to reduce the unmeasurable fuel to a minimum."

Effect on Weight Empty: O
Effect on Balance: O
Effect on Performance: None

C O N V A I R A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-00		DATE:		
CUSTOMER: Delta Ai	r Lines Inc.	MCL 10,201 Dtd 6 March 1958		
CHANGE NO: 134		MODEL: (880) 22- 2		
TITLE: Passenger	r Window Glare Contro	ol, Deletion of		
	quested during meetings on 13 and 14 Febru			
REASON FOR CHANGE:	Customer requested.	a tak		
		Rejected		
	EFFECT ON WEIGHT	EFFECT ON BALANCE *		
Guar. Wt. Empty	Oper. Wt. Empty			
-60.0 lb	-60.0 1ъ	-49,980 Inch Lb.		
EFFECT ON GUARANTEED PERFORMANCE: * None				
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:				
CCP No. 138 Engineering Approval				
LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:				
SPECIAL PROVISIONS	TECT ON PRICE PER AIRPLANE: curring: n-Recurring tal:			
ACCEPTED: CONVAIR, A Div. of Gen. Dyn. Corp.				
BY:DATE:		Chief of Contract Administration Commercial		

CONVAIR: SD

Title: Passenger Window Glare Control, Deletion of

Origin: Delta requested during meeting of Delta and Convair repre-

sentatives on 13 and 14 February 1958.

Reason for Change: Customer requested.

Description of Change:

Page 111, Paragraph 3.19.3.5 CURTAINS:

"Each passenger coat compartment opening shall be curtained. Curtain material shall be as specified in the Finish Specification. Provisions shall be made for the installation of passenger window roller type glare control. Rubber extrusions shall be provided to cover curtain slide on each side of passenger window frames."

Effect on Weight Empty:

Effect on Balance:

-60.0 pounds

-49,980 inch-pounds

Effect on Performance: None

A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-003	- 8	DATE:		
CUSTOMER: Delta Air	Lines Inc.	MCLDtd		
CHANGE NO: 133	Till Control on the Control of the C	MODEL: (880) 22- 2		
TITLE: Specificati	on Administrativ	e Change		
Convair initiated, and agreed to by Delta during meeting of ORIGIN: Delta and Convair representatives on 13, 14 February 1958. REASON FOR CHANGE: To clarify the intent of the Specification.				
NA S	EFFECT ON WEIG	HT * EFFECT ON BALANCE *		
Guar. Wt. Empty	Oper. Wt. Empt	У		
0	0	O Inch Lb.		
EFFECT ON GUARANTEED PERFORMANCE: *				
		None		
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:				
		Engineering Approval		
LATEST DATE OF ACC	EPTANCE:	AIRPLANES AFFECTED:		
SPECIAL PROVISIONS		EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:		
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.		
		BY: Chief of Contract Administration Commercial		

CONVAIR: SD

Title: Specification Administrative Change

Origin: Convair initiated, and agreed to by Delta during meeting of Delta and Convair representatives on 13, 14 February 1958.

Meason for Change: To clarify the intent of the Specification.

Description of Change:

Page 80, Paragraph 3.16.11.1 LANDING GEAR WAFNING HORN:

Revise the second sentence to read as follows:

"When any one of the landing gear units is not fully extended and locked, the horn will operate if any one of the power levers is closed or the wing flap selector is not in the range of an approach to land."

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

A Division of General Lynamics Corporation San Diego, California

COMPLICIAL CHANGE PROPOSAL

SFEC NO: ZD-22-003		DATE:					
CUSTOMER: Delta Ai		MCL 10,189 Dtd					
CHANGE NO: 132		MODEL: (880) 22- 2					
Shoulder Harness Retractor, Stewardess Seats; Installation of							
GRIGIN: TWA request	ted and Convair prop	oosed for Delta.					
harness and safety belt away from seat occupant to a position clear of the seat.							
	EFFECT ON WYIGHT	* EFFECT ON BALANCE *					
Guar. Mt. Empty	Oper. Wt. Empty						
+2.0 lbs	+2.0 lbs	+1,914 Inch Lb.					
+2.0 lbs +2.0 lbs +1,914 Inch ho. EMPLOT ON GUARANTEED PERFORMANCE: * None None							
+ Regitable Changes Will be Accumulated and Totals Reflected in a Future Change Proposal Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:							
		Engineering Approval					
LATING DATE OF ACCEPTANCE: ALRPLANES AFFECTED:							
S. ECIAL PROVISIONS		REFRECT ON PRICE PER AIRPLANE: Recurring: Hon-Recurring Total:					
ACCERTIO:	C	CONVAIR, A Div. of Gen. Dyn. Corp.					
PY:		Chief of Contract Administration Commercial					

CONVAL .: SD

Title: Shoulder Harness Retractor, Stewardess Seats; Installation of

Origin: TWA requested and Convair proposed for Delta.

meason for Change: To simultaneously retract stewardess' shoulder

harness and safety belt away from seat occupant to

a position clear of the seat.

Description of Change:

Page 105, Paragraph 3.19.1.1.7 SAFETY BELTS:

Add the following sentence to the end of paragraph:

"Each of the three cabin attendant seats shall be provided with a shoulder harness-safety belt retracting device to withdraw the harness and belt halves to a flat position against the seat backs."

Effect on Weight Empty: +2.0 pounds

Effect on Balance: +1,914 inch-pounds

Effect on Performance: None

C () N V A I R A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-00	3		DATE:_				
CUSTOMER: Delta Air Lines Duc.			MCL_	-	_Dtd	***	
CHANGE NO: 131			MODEL:	(880) 22-2		
TITLE: Specification Administrative Change.							
ORIGIN: Convair initiated. REASON FOR CHANGE: To clarify the intent of the Specification.							
	EFFECT ON WEI	and the second	M.	EF	FECT ON	BALANCE *	
Guar. Wt. Empty	Oper. Wt. Emp	ty			0	Inch Lb.	
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal							
Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes: Engineering Approval							
LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:							
BFEGIAL PROVIDIONS			EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:				
ACCEPTED: . CO			CONVAIR, A Div. of Gen. Dyn. Corp.				
BY: DATE:			BY: Chief of Contract Administration Commercial				

CONVAIR: SD

Title: Specification Administrative Change.

Origin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Figure 1-2 INBOARD PROFILE AND PLAN VIEW:

Delete forward and aft cargo capacity figures from illustration.

Effect on Weight Empty: O
Effect on Balance: O
Effect on Performance: None

The following is not to appear in Specification language:

"The correct forward and aft cargo capacities are given in paragraph 3.1.3.3; therefore deletion from Figure 1-2 is proposed to avoid duplication thereof."

CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-003			DATE:				
CUSTOMER: Delta Air Lines Inc.			MCL 10,194 DTD 12 February 19				
CHANGE NO: 1300			MODEL: (880) 22-2				
TITLE: Passenger	Seat Fabric Weight	s, Rev	rision to				
REASON FOR CHANGE: TO		рава	month of October langer seat fabric an				
EFFECT	ON WEIGHT *		EFFECT ON BA	LANCE *			
GUAR, WT. EMPTY +76.0 lb	OPER. WT. EMPTY		+66,576	INCH LB.			
ACCEPTANCE OF THIS CH		SAL.	ENGINEER	ING APPROVAL			
LATEST DATE OF ACCEPTANCE:		AIRPL	ANES AFFECTED:				
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:					
BY:		CONVA	MR, A DIVISION OF GENERAL DY				
DATE:				CONVENTAGE OFFICE			

CONVAIR: SD

Delta Air Lines Inc. Change No. 1300

Page 1 of 2

Title: Passenger Seat Fabric Weights, Revision to

Origin: Customer request to Convair during month of October 1958

Reason for Change: To adjust weight of passenger seat fabric and

supplement to CCP No. 130A

Description of Change:

This change has no effect on Specification language, except as noted in Exhibit "A".

Effect on Weight Empty: +75.0 lb Effect on Balance: +66,576 inch-lb Effect on Performance: None

Page 2 of 2

EXHIBIT "A"

The following shall not appear in Specification language:

The below listed item shows revised weight for seat fabric selected by Customer:

"Seat Upholstery (36-double seats)

16.0 02"

NOTE: This CCP supplements CCP No. 130A and cancels CCP No. 130B.
The weight increase represented herein is for the additional
4.5 oz of fabric weight above the Convair basic weight of
11.5 oz.

A Division of General Lynamics Corporation San Diego, California

SPEC NO: ZD-22-00 3			DATE:				
CUSTOMER: Delta Air	Lines Inc.			Dtd			
CHANGE NO:129		HO	DEL:_	(880) 22- 2			
TITLE: Stewardess Seats and Handsets, Relocation of							
Convair initiated and Delta request by Letter, File 142, dated ORIGIN: 13 January 1958							
TEASON FOR CHANGE: To incorporate relocation of stewardess seats and handsets into the Specification as defined in CCP 35B, Proposal 2.							
	EFFECT ON WEIG	W. San	25.00	EFFECT ON BALANCE *			
Guar. Wt. Empty	Oper. Wt. Emp	t.y					
0	0			O Inch Lb.			
EFFECT ON GUARANTEED PERFORMANCE: *							
		None	100				
	* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal						
Acceptance of thi Prior Acceptance							
				Engineering Approval			
LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:							
SPECIAL PROVISIONS:		Recurri Non-Rec	ng: urrin	RICE PER AIRPLANE:			
ACCEPTED:		CONVAIR	, A [Div. of Gen. Dyn. Corp.			
BY:	BY: Chie	Chief of Contract Administration Commercial					

CONVAIR: SD

Delta Air Lines Inc. Change No. 129

Page 1 of 2

Title: Stewardess Seats and Handsets, Relocation of

Origin: Convair initiated and Delta request by Letter, File 142,

dated 13 January 1958.

Reason for Change: To incorporate relocation of stewardess seats and

handsets into the Specification as defined in

CCP 35B, Proposal 2.

Description of Change:

Page 99, Paragraph 3.17.2.3.5 SERVICE INTERPHONE:

Change second line from top of page as follows:

From: "Both Buffet Areas"

To: "Forward Stewardess Seat"

Aft Stewardess Seat"

In second paragraph from top of page, revise the second and third sentences as follows:

"A single stroke chime and call light shall be installed at the flight deck, forward stewardess station and aft stewardess station for use with these handsets, and shall be activated by momentary contact call switches labled "Stewardess" and "Cockpit". An additional contact call switch and signal light shall be installed at the forward stewardess station and aft stewardess station for interstation signals."

Page 10d Paragraph 3.19.2.1.3 CONTROL PANEL:

Revise the paragraph to read as follows:

"A cabin attendant's panel shall be provided over each service door on right hand side of cabin (one forward and one aft). These panels shall contain controls for the following:

Entrance, passenger and buffet lighting Passenger and crew call system

A separate control panel for the service interphone and public address systems, and a handset shall be provided, one each at forward and aft stewardess stations."

Page 2 of 2

Page 104, Paragraph 3.19.1.1.4 CABIN ATTENDANT SEATS:

Revise the first sentence to read as follows:

"Three upholstered seats and back-rests shall be provided for cabin attendants; one single aft facing seat on aft face of forward left hand coat compartment, one aft facing seat on left hand cabin aft bulkhead and one forward facing seat on inboard face of aft left hand coat compartment."

Figure 1-2 INBOARD PROFILE AND PLAN VIEW Figure 1-3 ALTERNATE SEATING ARRANGEMENT

CONVAIR: SD

Revise above figures to show relocation of stewardess seats as applicable.

- Enclosure: (A) One copy of Convair Drawing No. SK-10751-1 STEWARDESS SEAT LOCATION (for information only)
 - (B) One copy of Convair Drawing No. SK-10757-2 STEWARDESS SEAT & P.A. PANEL, AFT ENTRANCE AREA (for information only)
 - (C) One copy of Convair Drawing No. SK-10757-3 STEWARDESS SEAT & P.A. PANEL, FWD. ENTRANCE AREA (for information only)
 - (D) One copy of Convair Drawing No. SK-10757-4 STEWARDESS SWITCH PANEL (TYP FWD & AFT SERVICE DOOR) (for information only)

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

C O N V A T A A Division of General Dynamics Corporation San Diego, California

	SPEC NO: ZD-22-00	3	_ DAT	E:	-			
	CUSTOMER: Delta Air	Lines Inc.	MCI	10,187	Dtd			
CHANGE NO: 128			MODEL: (880) 22- 2					
	ORIGIN: TWA requested and Convair proposed for Delta REASON FOR CHANGE: To make shoulder harness detachment independent of the safety belt.							
and the latest designation of the latest des								
-								
Canal Della		EFFECT ON WEIGH	HT *	EFFI	ECT ON	BALANCE *		
-	Guar. Wt. Empty	Oper. Wt. Empt	У					
And in case of the last	+1.0.1b	+1.0 lb		+	250	Inch Lb.		
Section of the last	EFFECT ON GUARANTEED PERFORMANCE: * None							
	* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes: Engineering Approval							
*	LATEST DATE OF ACCEPTANCE: AIRPLANES AFFECTED:							
	SPECIAL PROVISIONS	EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:						
1	ACCEPTED:		CONVAIR	, A Div. of	Gen.	Dyn. Corp.		
	BY:		BY:	A STATE OF THE RESIDENCE	-	and the same and an overland the same		
	DATE:		Chief		act Adı nercia	ministration l		

CONVAIR: SD

Delta Air Lines Inc. Change No. 128

Title: Crew Safety Belt Buckle, Design of

Origin: TWA requested and Convair proposed for Delta

Reason for Change: To make shoulder harness detachment independent of

the safety belt.

Description of Change:

Page 105, Paragraph 3.19.1.1.7 SAFETY BELTS:

Add the following after the second sentence ending in the fourth line:

"The design of the safety belt buckle, for the pilot, copilot, flight engineer and observer, shall be such that the shoulder harness and crotch strap can be unfastened simultaneously without unfastening the safety belt."

Page A-12, APPENDIX I-C, FURNISHINGS:

After seventh, eighth and eleventh items in the Description List, "Pilots' Safety Belt and Shoulder Harness", "Flight Engineer's Safety Belt and Shoulder Harness" and "Observer's Safety Belt and Shoulder Harness", under Manufacturer, add:

"Pacific Scientific"

Effect on Weight Empty: +1.0 pound

Effect on Balance: +250 inch-pound

Effect on Performance: None

SPEC NO: ZD-22-003		DA'FE:
CUSTOMER: Delta Air	r Lines Inc.	MCL 10,175 Dtd 3 January 1958
CHANGE NO: 127A		MODEL: (880) 22- 2
TITLE: Customer Fr	urnished Passenge	er Seats
ORIGIN: TWA Request	and Convair Pro	oposed for DAI.
REASON FOR CHANGE:	To provide for isenger seats in revision to CCP	installation of Buyer furnished pas- lieu of Convair furnished seats, and No. 127.
	EFFECT ON WEIGH	
Guar. Wt. Empty	Oper. Wt. Empty	Negligible Inch Lb.
EFFECT ON GUARANTEEI		one Cancelled
* Negligible Changes Totals Reflected	Will be Accumulation a Future Chang	lated and ge Proposal
Acceptance of this Prior Acceptance		
20.0 00.0 00.0 230		Engineering Approval
LATEST DATE OF ACCES	PTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.
BY: DATE:		BY: Chief of Contract Administration Commercial

Delta Air Lines Inc. Change No. 127A

Page 1 of 2

Title: Customer Furnished Passenger Seats

Origin: TWA Request and Convair Proposed for DAL

Reason for Change: To provide for installation of Buyer Furnished passenger seats in lieu of Convair Furnished seats.

Description of Change:

CONVAIR: SD

Page 5, Paragraph 1.3 WEIGHT INCREASE:

Revise the last sentence in the paragraph to read as follows:

"Refer to Appendix I-B for list of Buyer vended and Buyer furnished items and their associated weights."

Page 103, Paragraph 3.19.1.1.1 SEATS:

Revise first sentence to read as follows:

"All seats, except Buyer Furnished passenger seats, shall be equipped with"

Page 104, Paragraph 3.19.1.1.6 PASSENGER SEATS:

Delete the entire paragraph and replace with the following:

"Provisions consisting of attach fittings only shall be made for the installation of 36 double Buyer furnished passenger seats. These seats shall be interchangeable with Convair seats in respect to attach point locations and attach point configuration. Passenger seat arrangement shall be as shown on Figure 1-2."

Page 104, Paragraph 3.19.1.1.6.1 SEAT BACK MOVEMENT:

Delete the entire paragraph.

Page 105. Paragraph 3.19.1.1.7 SAFETY BELTS:

Revise first sentence to read as follows:

"Buyer approved commercial type safety belts shall be provided on all crew and club area seat accommodations."

Add the following to the end of paragraph:

"Buyer furnished passenger seats shall include passenger safety belts."

Delta Air Lines Inc. Change No. 127A

Page 2 of 2

Add new page ---

"Al - APPENDIX I-B - CUSTOMER FURNISHED - CONVAIR INSTALLED EQUIPMENT:

No. Description

No. Description

Bouble passenger seats (including fabric and safety belts and excluding attach fittings)

Passenger trays (integral folding food trays located in seat backs)

Weight

1,764 lb

Page Al2 - APPENDIX I-C - FURNISHINGS:

Delete the fifth item in the equipment list:

"36 Passenger Seats (Double)"

Change the twelfth item in the equipment list:

From: "84 Passenger Safety Belts"

To: "12 Passenger Safety Belts"

Effect on Weight Empty: Effect on Balance:

Negligible

Effect on Performance: None

C () N V A I R A Division of General Dynamics Corporation San Diego, California

SPEC NO: ZD-22-00	3	DATE:	
CUSTOMER: Delta Ai:	r Lines Inc.	MCL_10,188	Otd 7 February 1958
CHANGE NO: 126		MODEL: (880)	22-2
TITLE: Self-Sealing	g Couplings in Mai	n Wheel Brake Line	s, Installation of
ORIGIN: TWA request	ed and Convair pro	posed for Delta	
REASON FOR CHANGE:	To retain uniformi	ty in equipment of	Model 22 Airplane
	EFFECT ON WEIGH	T * EFFE	CT ON BALANCE *
Guar. Wt. Empty	Oper. Wt. Empty		
+7.0 lbs	+7.0 lbs	+6,1	47 Inch Lb.
EFFECT ON GUARANTEE	ED PERFORMANCE: *		
	1	lone	
* Negligible Change Totals Reflected	es Will be Accumul in a Future Chang	ated and e Proposal	
	s Change is Depen of the Following		
		Eng	ineering Approval
LATEST DATE OF ACCE	EPTANCE:	AIRPLANES AFFECTED	
SPECIAL PROVISIONS:		EFFECT ON PRICE PER Recurring: Non-Recurring Total:	
ACCEPTED:		CONVAIR, A Div. of	Gen. Dyn. Corp.
BY:		BY:	a hand a state of the same of the same
DATE:			ct Administration ercial

Delta Air Lines Inc. Change No. 126

Title: Self-Sealing Couplings in Main Wheel Brake Lines; Installation

of

Origin: TWA requested and Convair proposed for Delta.

Reason for Change: To retain uniformity in equipment of Model 22 Airplane Description of Change:

Page 63, Paragraph 3.15.1.4 LINES:

Add the following after the third sentence ending in the ninth line:
"Self-sealing couplings shall be installed in the hydraulic brake lines at each main wheel brake."

Effect on Weight Empty: +7.0 pounds

Effect on Balance: +6,147 inch-pounds

Effect on Performance: None

The following not to appear in Specification language:

"A total of eight (8) self-sealing couplings, of special design, are included in the equipment items covered in this Change."

SPEC NO: ZD-22-00		DATE:	watersaan hand.
CUSTOMER: Delta Air	Lines Inc.	MCL 10,186 Dta	28 January 195
CHANGE NO: 125A		MODEL: (880) 22-2	
TITLE: Main Land:	ing Gear Piston Head B	earing; Change in N	aterial of
and Delta	nested by Letter, File TWX dated 10 June 195 Customer requested.	142, dated 24 Jam	ery 1958;
Character 114 12 - Land	EFFECT ON WEIGHT *	. EFFECT OF	BALANCE *
Guan. Wt. Empty	Oper. Wt. Empty	/ 10,536	Inch Lb.
EFFECT ON GUARANTE		ne	
	es Will be Accumulated in a Future Change Pr		
Acceptance of thi Frior Acceptance	s Change is Dependent of the Following Chan	Upon ges:	
		Engineer	eing Approval
LATEST DATE OF ACCE	PTANCE: AIRP	LANES AFFECTED:	
SPECTAL PONTULONO	Recu Hon-	CT ON PRICE PER ATS Becurring	
ACCEPTED:		MIR, A Div. of Gen.	, Dyn. Corp.
BY:	To the residence to the second of the second		
DATE:			

Delta Air Lines Inc. Change No. 125A

CONVAIR: SD

Main Landing Gear Piston Head Bearing; Change in Material of Title:

Delta requested by Letter, File 142, dated 24 January 1958; Origin:

and Delta TWX dated 10 June 1958.

Reason for Change: Customer requested.

Description of Change:

This change has no effect on Specification language.

Effect on Weight Empty: /12.0 pounds Effect on Balance: /10,536 inch-pounds Effect on Balance:

Effect on Performance: None

The following is not to appear in the Specification language:

"This change consists of a material change of the main landing gear piston head bearing from 7075 aluminum alloy to aluminum bronze."

SPEC NO: ZD-22-00	3	DATE:
CUSTOMER: Delta Air	Lines Inc.	MCL 10,185 Dta
CHANGE NO: 124		MODEL: (880) 22-2
TITLE: Titanium Co	owling, Door Ski Aluminum Alloy	ns and Frames, Where Heat Permits;
ORIGIN: 2. Convain	quested and Conv	air proposed for Delta.
REASON FOR CHANGE:	1. To eliminat fires. 2. To reduce w	e friction spark ignition of crash eight.
	EFFECT ON WEIG	GHT * EFFECT ON BALANCE *
Guar. Wt. Empty	Oper. Wt. Emp	ty
-73.0 lbs	-73.0 lbs	-58,400 Inch Lb.
EFFECT ON GUARANTEE	D PERFORMANCE:	
		None
* Negligible Change Totals Reflected		
Acceptance of thi Prior Acceptance	s Change is Depo of the Following	endent Upon g Changes:
		Engineering Approval
LATEST DATE OF ACCE	PTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.
BY:		BY: Chief of Contract Administration

Title: Titanjum Cowling, Door Skins and Frames, Where Heat Permits;

Change to Aluminum Alloy

Origin: 1. TWA requested and Convair Proposed for Delta.

2. Convair initiated.

2.

Reason for Change: 1. To eliminate friction spark ignition of crash

fires. To reduce weight.

Description of Change:

Page 44, Paragraph 3.11.6 COWLING:

Revise the second sentence to read as follows:

"The cowling shall be of aluminum alloy, stainless steel or titanium; or a combination thereof."

Delete the third sentence:

"(The cowling frames and skin, from the engine inlet aft shall be of stainless steel or titanium.)"

Enclosure: (A) One copy of Convair Drawing No. PP-22-05+ STUDY -MODEL 880 CONTROLLED WHEELS UP LANDING ATTITUDES - GROUND RELATIONSHIPS (for information only).

Effect on Weight Empty: -73,0 pounds Effect on Balance: -58,400 inch-pounds

Effect on Performance: None

SPEC NO: ZD-22-003		DATE:
CUSTOMER: Delta Air	Lines Inc.	MCL 10,180 Dtd 10 January 1958
CHANGE NO: 123		MODEL: (880) 22-2
TITLE: 1000-Watt	Landing Lights, Inst	allation of
ORIGIN: TWA reques	ted and Convair prope	osed for Delta.
REASON FOR CHANGE:	To provide landing allow for condition	ns of limited visibility.
	EFFECT ON WEIGHT *	EFFECT ON BALANCE *
Guar. Wt. Empty	Oper. Wt. Empty	
+5.0 lbs	+5.0 lbs	+3,250 Inch Lb.
EFFECT ON GUARANTEE	D PERFORMANCE: * None	Rejected
	s Will be Accumulate in a Future Change P	
Acceptance of thi Prior Acceptance	s Change is Dependen of the Following Cha	t Upon nges:
		Engineering Approval
LATEST DATE OF ACCE	PTANCE: AIR	PLANES AFFECTED:
SPECIAL PROVISIONS:	Rec Non	ECT ON PRICE PER AIRPLANE: arring: -Recurring al:
ACCEPTED:		VAIR, A Div. of Gen. Dyn. Corp.
BY: DATE:	BY:	Chief of Contract Administration Commercial

Delta Air Lines Inc. Change No. 123

CONVAIR: SD

Title: 1000-Watt Landing Lights, Installation of

Origin: TWA requested and Convair proposed for Delta.

Reason for Change: To provide landing lights of increased power to

allow for conditions of limited visibility.

Description of Change:

Page 74. Paragraph 3.16.8.1.2 LANDING LIGHTS:

Ravise the first sentence to read as follows:

"Two 1000-watt landing lights shall be provided and located so as not to create reflections into the pilots' compartment as a result of light beam interference with structure."

Effect on Weight Empty: +5.0 pounds Effect on Balance: +3,250 inch-pounds

Effect on Performance: None

C C N V A T d A Division of Ceneral Dynamics Corporation San Diego, California

SPEC NO: ZD-22-003		DATE:
CUSTOMER: Delta Air	Lines Inc.	MCL 10,184 Dtd_
CHANGE NO: 122		MODEL: (880) 22-2
TIPLE: Foot Rest	, Wedge-Shaped, H	assock Type; Provision of
CRIGIN: TWA reque	sted and Convair	proposed for Delta.
REASON FOR CHANGE:	To provide add	itional passenger comfort.
	EFFECT ON WEIGH	
Guar. Wt. Empty	Oper. Wt. Empty	
+ 105.0 los	+ 105.0 lbs	+85,050 Inch l.b.
METECT ON GUARANTE	ED PERFORMANCE: *	None
* Megligible Change Totals Reflected	es Will be Accumul in a Future Chang	ated and ge Proposal
Acceptance of the Prior Acceptance	is Change is Deper of the Following	dent Upon Changes:
		Engineering Approval
LATEST DATE OF ACC	EPTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS		EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.
BY:DATE:		BY: Chief of Contract Administration Commercial

CONVAIR: SD .

Delta Air Lines Inc. Change No. 122

Title: Foot Rest, Wedge-Shaped, Hassock Type; Provision of

Origin: TWA requested and Convair proposed for Delta.

Meason for Change: To provide additional passenger comfort.

Description of Change:

Page 108, add the following new paragraph to the page, after paragraph 3.19.2.3:

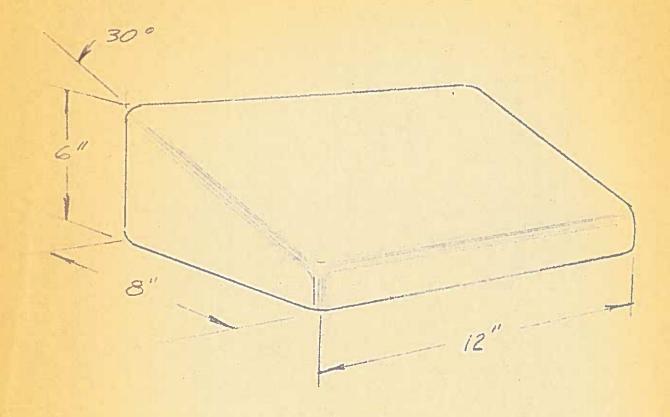
"3.19.2.3.1 PASSENGER SEAT FOOT HEST: A wedge-shaped, hassock type foot rest shall be provided, as loose equipment, for each passenger seat. The foot rest shall measure approximately 12.0" long, 8.0" wide and 6.0" high with a 30-degree slope on the 8.0" width. The foot rest shall consist of the following, or equivalent, materials: styrofoam core covered with a layer of fibreglass, padding of 1/2" polyether foam and outer cover of 20-oz/sq yard material, including two zippers."

Enclosure: (A) One copy of sketch - PROPOSED FOOT REST

Effect on Weight Empty: +105.0 pounds

Effect on Balance: +85,050 inch-pounds

Effect on Performance: None



SPEC NO: ZD-22-00;	3	DATE:
CUSTOMER: Delta Ai	r Lines Inc.	MCL 10,182 Dtd
CHANGE NO: 121		MODEL: (880) 22-2
TITLE: Rivets in	n Wing Leading Ed	ges, Use of
ORIGIN: Convair	initiated.	
REASON FOR CHANGE:	To reduce weigh	t.
	EFFECT ON WEIGH	HT * EFFECT ON BALANCE *
Guar. Wt. Empty	Oper. Wt. Empt	
-11.0 lbs	-11.0 lbs	-8,470 Inch Lb.
EFFECT ON GUARANTEI	ED PERFORMANCE: *	
		None
* Negligible Change Totals Reflected	es Will be Accumu in a Future Chan	lated and ge Proposal
Acceptance of the Prior Acceptance	is Change is Depe of the Following	ndent Upon Changes:
		Engineering Approval
LATEST DATE OF ACC	EPTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS		EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.
BY: DATE:		BY: Chief of Contract Administration Commercial

Delta Air Lines Inc. Change No. 121

CONVAIR: SD

Title: Rivets in Wing Leading Edges, Use of

Origin: Convair initiated.

Reason for Change: To reduce weight.

Description of Change:

Page 3, Paragraph 1.1 BASIC TYPE:

In second line of second paragraph from top of page, change the period after "nuts" to a comma and add the following:

"except that it will be permissible to use rivets to attach the wing leading edge section to the forward legs of the piano hinges."

Effect on Weight Empty: -11.0 pounds
Effect on Balance: -8,470 inch-pounds
Effect on Performance: None

SPIC NO: ZD-22-003		_ DATE:
CUSTOMER: Delta Air	Lines Inc.	MCL 10,195 Dtd_
CHANGE NO: 120A		MODEL: (880) 22-2
TITHE: Single Loop	, A-C Fenwal Fire	e Detector System; Installation of
		Delta for a single loop system. ted; and revision to CCP No. 120.
	EFFECT ON WEIG	GHT * EFFECT ON BALANCE *
Guar. Wt. Empty	Oper. Wt. Empt	-18,564 Inch Lb.
EFFICT ON GUARANTEE		* None
* Negligible Change Totals Reflected Acceptance of thi Prior Acceptance	in a Future Char te Change is Dept	endent Upon ig Changes:
		Engineering Approval
LATEST DATE OF ACCU	EPTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS	•	EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.
BY:		BY: Chief of Contract Administration Commercial

fitle: Single Loop, A-C Fenwal Fire Detector System, Installation of

Origin: Verbal request of TWA and Delta for a single loop system.

Meason for Change: Customer requested; and revision to CCP No. 120.

Description of Change:

Page 87, Paragraph 3.16.17.1 EMERGENCY POWEH:

Under "AC Emergency", add the following new item after "Engine Instruments"

"Lngine Macelle Fire Detector"

Page 89, Paragraph 3.16.17.1 EMERGENCY POWER:

Under "DC Emergency", delete the first item under "Warning" (Engine Fire Detector) and renumber the remaining items 1 through 7.

Page 112, Paragraph 3.19.4.2.1 GENERAL:

nevise the first sentence to read as follows:

"Single loop, discrete sensing, continuous type fire detectors shall be installed in the nacelles, including the engine compressoraccessory section and the engine burner and turbine section."

Figure 3.12-5 FIRE DETECTOR SYSTEM, ENGINE NACELLES

Replace above figure in Specification with Enclosure (A)

English NACELLES (Revised 1-23-58).

Effect on Weight Empty: -23.0 pounds

Effect on Balance: -15,564 inch-pounds

Lifect on Performance: None

REPORT NO. 20-22-003 MODEL 4.2 - 2 DATE

1-23-58

FIRE DETECTOR SYSTEM ENGINE NACELLES

WARNING LIGHTS

BLINKING LIGHTS - TURBINE COMPARTMENT

STEADY LIGHTS - ACCESSORY COMPARTMENT

REFER TO FIRE WARNING ELECTRICAL SCHEMATIC 22-80381 FOR WIRING CONTROL UNITS FIRE AND-OVERHEAT TEST SWITCH ACCESSORY COMP. --FIREWALL

NACELLES

- CONTINUOUS FIRE DETECTOR CABLE

-TURBINE COMPARTMENT FIGURE 3.12-5

SPEC NO: ZD-22-00			DATE:		
CUSTOMER: Delta Ain	Lines Inc.		MCL	- Dtd_	
CHANGE NO: 119			MODEL:_	(880) 22-	2
TITLE: Specificati	on Administrat	tive Chang	ge		
ORIGIN: Convair ini	tiated.				
REASON FOR CHANGE:	To clarify th	ne intent	of the	Specifics	tion.
	EFFECT ON WI			EFFECT (ON BALANCE *
Guar. Wt. Empty	Oper. Wt. En	mpty			
0	0			C	Inch Lb.
EFFECT ON GUARANTEI	ED PERFORMANCE		A Comment		
		None		Miles I	
* Negligible Change Totals Reflected	es Will be Acci in a Future Cl	umulated hange Pro	and posal		
Acceptance of the Prior Acceptance	ls Change is Do of the Follow:	ependent ing Chang	Upon es:		
				Engine	ering Approval
LATEST DATE OF ACCE	EPTANCE:	AIRPL	ANES AF	FECTED:	
SPECIAL PROVISIONS		Recur Non-R	ring:	5	
		Total			
ACCEPTED:		CONVA	IR, A D	iv. of Ge	n. Dyn. Corp.
BY:		BY:			
27.2		City.	100 00	Contract	Administration

Delta Air Lines Inc. Change No. 119

CONVAIR: SD

Title: Specification Administrative Change

Origin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Page 116. Paragraph 3.20.1.2 AIR CONDITIONING CONTROLS

Delete the last sentence in the first paragraph and substitute the following:

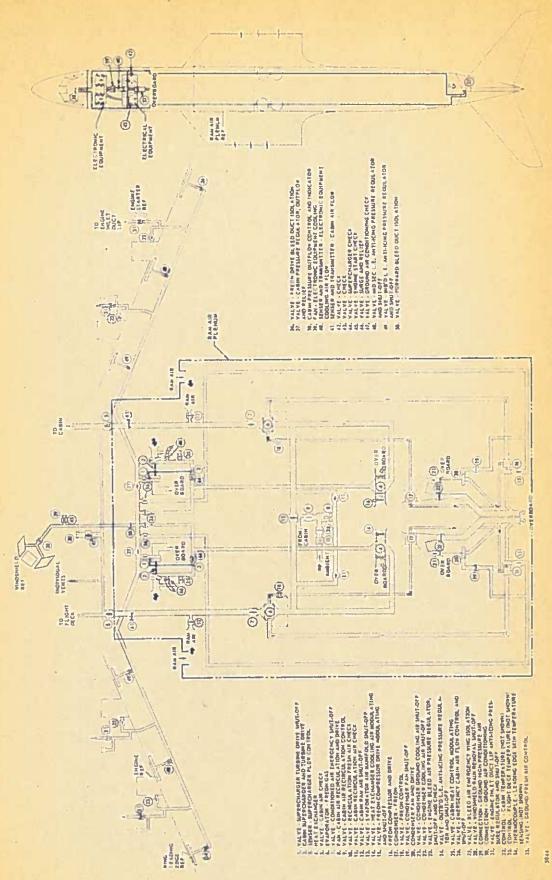
"A dual airflow indicator shall be installed to indicate the cabin (RH) and flight deck (LH) compressor airflows. The indicator shall be located on the flight deck. A bearing temperature indicator and a RPM indicator shall be installed on the flight deck for each turbo-compressor."

Figure 3.20-1 AIR CONDITIONING SCHEMATIC

Replace above figure in Specification with Enclosure: (A)

Enclosure: (A) One (1) copy of Figure 3.20-1 AIR CONDITIONING PRES-SURIZATION SYSTEM

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None



A Division of Honoral Primains Corporation San More, India raia

1188° NO: 20-22-003	Agentina agentina agentina antina agentina antina agentina agentina agentina agentina agentina agentina agenti	DAUE:
cush way: Delta Air I	ines Inc.	MC15 10,171 Dtd
OHAMAR NO: 118		:moer: (880) 22- 2
Water Tank Qu	antity Gage, Install	ation of
WA requested	and Convair propose	d for Delta
OFFASON FOR CHANGE: To	provide means for w	ater tank quantity indication g station.
	REFLOR OF MURGED *	EFFECT ON BALANCE *
Amer. Mr. Smpty	oper. Wh. Smity	
+6.0 lbs	+6.0 lbs	+3,603 Inch l.b.
HET WE OH GUARANTERED	PERFORMANCE: * None	Cancelled
Medigible Changes Totals Reflected is	Will be Accumulated n a Future Change Pro	and oposal
Acceptance of this trior Acceptance of	Change is lependent f the Following Chang	Upon rest
		Engineering Approval
DATEST DATE OF ACCEP	CAMCL: AIRF	LANIIS AFFECTED:
JACTAL PROVISIONS:	lecu Ton-	OT (N PRICE PER AURPLANE: rring:
ACCEPTIO:	CONV.	AIR, A Div. of Gen. Dyn. Corp.
DV.		
	Superintended in a paper form. Very a management and	hief of Contract Administration Commercial

Delta Air Lines Inc. Change No. 118

Title: Water Tank Quantity Gage, Installation of

Origin: TWA requested and Convair proposed for Delta

Reason for Change: To provide means for water tank quantity indication at the ground servicing station.

Description of Change:

Page 106, Paragraph 3.19.2.1.4 WATER SYSTEM:

Add the following after the sixth sentence ending in the tenth line:

"A remote reading water tank quantity gage, actuated by a transmitter, shall be installed on the fuselage exterior tank servicing panel, which shall be readily visible to the ground servicing personnel."

Effect on Weight Empty: +6.0 pounds

Effect on Balance: +3,603 inch-pounds

Effect on Performance: None

SPEC NO: ZD-22-00	3	DATE:_		
CUSTOMER: Delta Air	Lines Inc.	MCI	Dtd	tre top tota
CHANGE NO: 116		MODEL:	(880) 22-	2
TITLE: Specificat:	ion Administrative	Change		
GRIGIN: Convair in:		ntent of the	Specificati	on.
	EFFLOT ON WHIGH	II. *	EFFECT ON	BALANCE *
Guar. Wt. Empty O	Oper. Wt. Empty		0	Inch Lb.
EFFECT ON GUARANTEI	D PERFORMANCE: *	None		
Acceptance of thi	in a Future Change is Change is Dependent of the Following	ge Proposal	Engineer	ring Approval
LATEST DATE OF ACCI	PTANCE:	AIRPLANES AF	FFECTED:	
SPICIAL PROVISIONS		EFFECT ON PI Recurring: Non-Recurring Total:	ng	
ACCEPTED:		CONVAIR, A	of Gen.	. Dyn. Corp.
BY: DATE:		BY: Chief of	Contract Ad Commercia	lministration

CONVAIR: SD

Delta Air Lines Inc. Change No. 116

Title: Specification Administrative Change

Origin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Page 29, Paragraph 3.7.1.5.1 MAIN ENTRANCE DOORS:

Revise the paragraph to read as follows:

"Two main entrance doors, with an average width of approximately 30 inches and a height of 74 inches, shall be provided on the left side of the passenger compartment; one forward of the wing and one aft of the wing."

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

EXHIBIT "A"

The following is not to appear in Specification language:

The attached CCP indicates an average door width of not less than 30 inches. This average width is the total opening area divided by the height of 74 inches.

The following minimum dimensions are given for information to further define the size of the main entrance door openings:

a.	Width at top of door neglecting corner radii	36.2"
b.	Width at bottom of door neglecting corner radii	26.0"
c.	Width of flat door sill considering corner radii (at floor level)	20.0"
d.	Clear opening width at heighth of 24.5" above floor	29.5"

This dimension is clear opening between door tracks at 24.5" above floor and increases above this point.

SPEC NO: ZD-22-00	3	DATE:			
CUSTOMER: Delta Ai	r Lines Inc.	MCL 10,173	Dtd 17 December 195		
CHANGE NO: 115		MODEL: (880) 22-2			
TITLE: Lavatory Hot Water Provisions, Installation of					
ORIGIN: TWA reques			h lavatory.		
	EFFECT ON WEIG		FECT ON BALANCE *		
Guar. Wt. Empty +31.0 lbs	Oper. Wt. Empt	+26, +33,	565 Inch Lb 589 Inch Lb.		
EFFECT ON GUARANTE	D PERFORMANCE: *	None			
* Negligible Change Totals Reflected					
Acceptance of thi Prior Acceptance					
		E	ngineering Approval		
LATEST DATE OF ACCE	EPTANCE:	AIRPLANES AFFECT	Eu:		
SPECIAL PROVISIONS:		EFFECT ON PRICE I Recurring: Non-Recurring Total:			
ACCEPTED:		CONVAIR, A Div.	of Gen. Dyn. Corp.		
BY:			ract Administration		

Delta Air Lines Inc. Change No. 115

Mitle: Lavatory Hot Water Provisions, Installation of

Origin: TWA requested and Convair proposed for Delta.

Reason for Change: To make hot water available in each lavatory.

Description of Change:

Page 107, Paragraph 3.19.2.2 LAVATORIES:

Revise the third sentence to read as follows:

"A wash basin, with a 3/4-inch diameter drain, a manually-operated drain stopper with easily replaceable seal, one each hot water and cold water household type spring-loaded faucets so designed that passengers may wash their hands with running water, shall be provided in each lavatory."

Add the following after the above sentence:

"A 2-quart capacity hot water tank, equipped with electrical heating elements, shall be installed, one in the forward lavatory and one aft to supply hot water to each lavatory wash basin."

Figure 3.19-2 AFT WATER SYSTEM SCHEMATIC

Mevise figure as required to show above change.

Effect on Weight Empty: +31.0 lbs
Effect on Useful Load: +8.0 lbs
Effect on Oper. Weight Empty: +39.0 lbs
Effect on Performance: None

+26,565 inch-pounds +7,024 inch-pounds +33,589 inch-pounds

apic no: <u>ZD-22-003</u>			DATE:		
CUSTOMER: Delta Air Lines Inc.			MCL 10,179 Dtd 13 January 195		
CHANGE NO: 114			MODEL: (880) 22- 2		
TITLE: D-C Driven Hydraulic Pump, Installation of					
ORIGIN: TWA request	ed, and Convai:	r propos	sed to Delta.		
REASON FOR CHANGE:	To provide mea	ans for draulic	ground recharging of main wheel system.		
	EFFECT ON WE		EFFECT ON BALANCE *		
Proposal "A" +129.0 Proposal "B" + 59.0	Oper. Wt. Em +129.0 1b + 59.0 1b		Proposal "A" +107,465 inch-lb Proposal "B" + 43,203 inch-lb		
EFFECT ON GUARANTEE	D PERFORMANCE:	* None	9		
* Negligible Change Totals Reflected Acceptance of thi Prior Acceptance	in a Future Change is De	ange Pro	Upon .		
LATEST DATE OF ACCE	PTANCE:	AIRPL	LANES AFFECTED:		
		Recur Non-R	EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:		
ACCEPTED:		CONVA	AIR, A Div. of Gen. Dyn. Corp.		
BY:		BY:			
DATE:		Ch	nief of Contract Administration Commercial		

Delta Air Lines.Inc. Change No. 114

CONVAIR: SD

Title: D-C Driven Hydraulic Pump, Installation of

Origin: TWA requested and Convair proposed for Delta.

Reason for Change: To provide means for ground recharging of main

wheel brake hydraulic system.

Description of Change:

PROPOSAL "A"

Page 63. Paragraph 3.15.1 DESCRIPTION AND COMPONENTS:

Add the following to the end of paragraph:

"A D-C electrically driven hydraulic pump, of approximately 1/2 GPM capacity, shall be installed in the hydraulic compartment to recharge the main wheel brake system on the ground. The pump inlet shall be connected to No. 2 reservoir."

Page 66,

Add the following new paragraph to the page under paragraph 3.16.2.3:

"3.16.2.3.1 HYDRAULIC PUMP MOTOR BATTERY: One 60 ampere-hour storage battery shall be provided to supply direct current to the electrically driven hydraulic pump described in paragraph 3.15.1. A pump control relay shall be installed near the battery, with the necessary wiring, and control shall be through a switch on the flight engineer's hydraulic panel."

Figure 3.15-1 HYDRAULIC SYSTEM

Revise figure as required to reflect this change.

Effect on Weight Empty: +129.0 pounds

Effect on Balance: +107,465 inch-pounds

Effect on Performance: None

PROPOSAL "B"

Page 63, Paragraph 3.15.1 DESCRIPTION AND COMPONENTS:

Add the following to the end of paragraph:

"A D-C electrically driven hydraulic pump, of approximately 1/2 GPM capacity, shall be installed in the hydraulic compartment to recharge the main wheel brake system on the ground. The pump inlet shall be connected to No. 2 reservoir.

Page 79.

Add the following new paragraph to the page, under paragraph 3.16.10.4:

"3.16.10.4.1 EXTERNAL D-C POWER RECEPTACLE: A D-C power receptacle shall be installed in the nose wheel well to provide direct current from a ground source to the electrically driven hydraulic pump described in paragraph 3.15.1. A pump control relay shall be installed adjacent to the power receptacle, with the necessary wiring, and control shall be through a switch on the flight engineer's panel. A bracket shall be installed in the nose wheel well side wall to accommodate the D-C power receptacle."

Figure 3.15-1 HYDRAULIC SYSTEM

Revise figure as required to reflect this change.

Effect on Weight Empty:

Effect on Balance:

+59.0 pounds +43,203 inch-pounds

Effect on Performance:

None

EXHIBIT "A"

The following is not to appear in the Specification language:

The below listed items are included in the equipment covered in this change:

D-C Driven Hydraulic Pump, PESCO, P/N 1110+6-0+0 (modified for use with Skydrol)

60 Ampere-Hour Battery, SONOTONE, TYPE 19-60H120

		DATE:			
CUSTOMER: Delta Air Lines Inc.		MCL 10.161 Dtd 6 November	1957		
CHANGE NO: 113		MODEL: (880) 22-2			
	Wheel Braking, Insta				
TWA requestion: TREASON FOR CHANGE:	Sept	Letter No. 880-323 dated 20 sember 1957, and Convair proposelta	sed		
To prevent the rotating wheels from flinging matter into the wheel wells and causing possible damage to equipment or structure.					
Guar. Wt. Empty	EFFECT ON WRIGHT *	EFFECT ON BALANCE	*		
+ 15.0 lbs	Oper. Wt. Empty + 15.0 lbs	+ 13,818 Inch Lt			
EFFECT ON GUARANTER	ED PERFORMANCE: * None	Rejected			
	es Will be Accumulate in a Future Change P				
		A. YY			
	is Change is Dependen of the Following Cha				
			val		
	of the Following Cha	nges:	val		
Prior Acceptance	Of the Following Cha	nges: Engineering Appro			
Prior Acceptance	of the Following Cha EPTANCE: AIR EFF Rec Non Tot	Engineering Appropriate PLANES AFFECTED: ECT ON PRICE PER AIRPLANE: urring: -Recurring			
Prior Acceptance LATEST DATE OF ACCE SPECIAL PROVISIONS:	Of the Following Cha EPTANCE: AIR EFF Rec Non Tot CON	Engineering Appropriate PLANES AFFECTED: ECT ON PRICE PER AIRPLANE: urring: -Recurring al:	·p.		

litle: Automatic Wheel Braking, Installation of

Origin: TWA requested, Reference: TWA Letter No. 880-323 dated 20 September 1957, and Convair pro-

posed for Delta.

Reason for Change: To prevent the rotating wheels from flinging matter

into the wheel wells and causing possible damage to

equipment or structure.

Description of Change:

Page 35, Paragraph 3.8.2.2 WHEELS, BRAKES AND BRAKE CONTROL SYSTEMS:

Add the following after the sentence ending in the tenth line from top of page:

"Means shall be installed for automatic braking of all main landing gear wheels to a complete stop prior to retraction into the wheel wells. Braking shall be accomplished by actuating the main wheel brake metering valves by means of hydraulically actuated cylinders, spring loaded to the off position. The brakes shall be applied after the main gear has retracted approximately 15 degrees."

Page 37, Paragraph 3.8.4.2 WHEELS AND BRAKES:

Add the following after the sentence ending in the sixth line from the top of the page:

"Means for automatic braking of the nose landing gear wheels shall be installed. The metering valve actuating cylinders, described in paragraph 3.8.2.2, shall actuate the nose wheel brake metering valve through the normal brake linkage."

Figure 3.8-2 HYD WHEEL BRAKE SCHEMATIC Figure 3.15-1 HYDRAULIC SYSTEM

Revise the above figures as required to reflect changes.

SPEC NO: ZD-22-003		DATE:	
CUSTOMER: Delta Ai	r Lines Inc.	MCI, 10,162	ntd
CHANGE NO: 112			80) 22- 2
TITLE: Retractabl	e Device for Pilo	t's and Copilot'	s Seat Belts
GRIGHN: TWA reques		TWA Letter 880-3 and Convair prop	28 dated 24 September
REASON FOR CHANGE:	To prevent the b	elt buckle from	dropping and causing or mounted instruments.
	EFFECT ON WEIGH		FFECT ON BALANCE *
Guar. Wt. Empty +0.5 lb	Oper. Wt. Empty +0.5 1b		+118 Inch Lb.
EFFECT ON GUARANTEE	D PERFORMANCE: *	None Reje	tel
* Negligible Change Totals Reflected	s Will be Accumul in a Future Chang	lated and ge Proposal	
Acceptance of thi Prior Acceptance	s Change is Deper of the Following	ndent Upon . Changes:	
			Engineering Approval
LATEST DATE OF ACCE	PTANCE:	AIRPLANES AFFEC	TED:
SPECIAL PROVISIONS:		EFFECT ON PRICE Recurring: Non-Recurring Total:	
ACCEPTED:		CONVAIR, A Dav.	of Gen. Dyn. Corp.
BY:		Chief of Con	tract Administration ommercial

CONVAIR: Delta Air Lines Inc. SD Change No. 112

Title: Retractable Device for Pilot's and Copilot's Seat Belts

TWA requested, Reference: TWA Letter 880-328 dated 24 September 1957 and Convair proposed for Delta. Origin:

Reason for Change: To prevent the belt buckle from dropping and causing possible damage to the console or mounted instruments.

Description of Change:

Page 105. Paragraph 3.19.1.1.7 SAFETY BELTS:

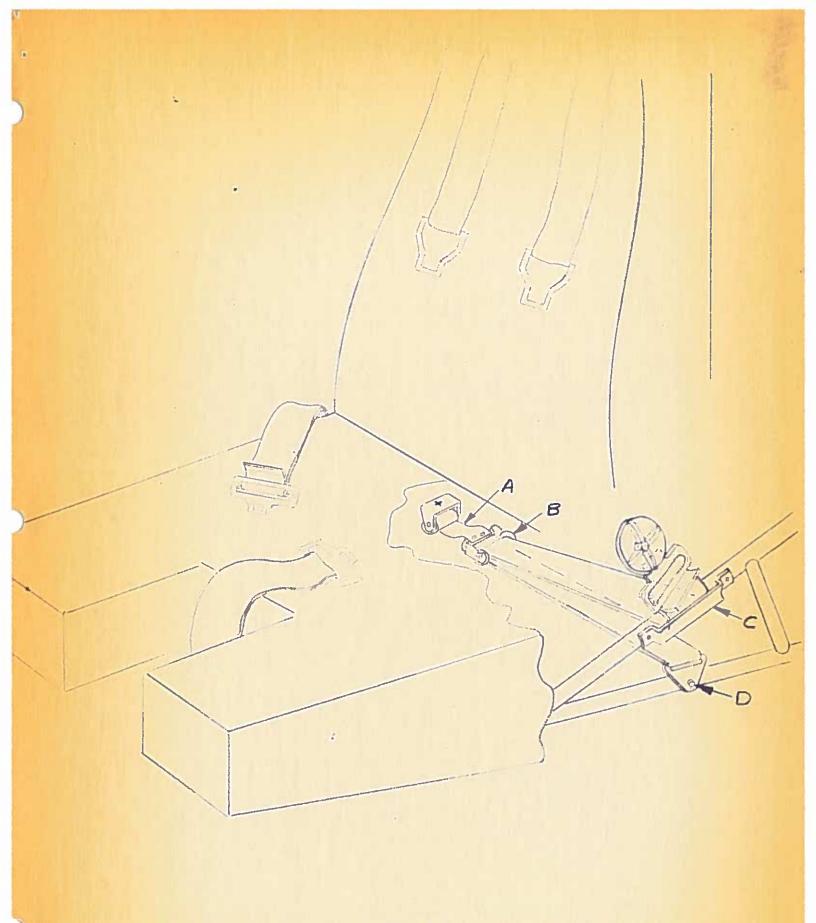
Add the following new sentence to the end of the paragraph:

"In addition, a device shall be installed to retract the left hand side of the pilot's and copilot's safety belts."

Enclosure: (A) One (1) copy sketch - Seat Belt Retractor Installation, with copy of description of installation and function.

Effect on Weight Empty: +0.5 pounds Effect on Balance: +118 inch-pounds

Effect on Performance: None



SEAT BELT RETRACTOR INSTL.

Per attached sketch, attach a spool mounted flat spring (A) to underside of seat bottom. Through self energizing action, spring winds up on spool when no unwinding load is applied. The free end of the spring is attached to a roller (B) of width to accommodate safety belt width.

When installing, belt is passed through guide (C), around roller (B) and then anchored to seat structure at point (D). Retracting action of spring (A) holds belt buckle at guide (C). When the two belt halves are joined around a seat occupant, the belt forms a straight line between (C) and (D). Therefore no belt loads are taken out by the retraction mechanism.

This concept requires that belt adjustment for length be accomplished at the buckle as shown, rather than near the belt anchor point as presently accomplished. However, adjusting at the buckle location should be more convenient.

SPEC NO: ZD-22-00	3	DATE:			
CUSTCHER: Delta Air Lines Inc.		MCLDtd_			
CHANGE NO: 111		MODEL: (880) 22-2			
FITHE: Deletion of	f Appendix I-C I	tems			
ORIGIN: Convair in	itiated.				
EEASON FOR CHANGE:	To delete inco	onsequential items from Appendix I-C.			
	DISTROM ON HUT	OHE & THEREOF ON DATAMON &			
Guar. Wt. Empty	EFFECT ON WEIG				
O O	0	0 Inch Lb.			
EFFECT ON GUARANTE	DFFECT ON GUARANTEED PERFORMANCE: * None				
* Negligible Change Totals Reflected Acceptance of the Prior Acceptance	in a Future Char is Change is Dep	endent Upon			
LATEST DATE OF ACC	EPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS		EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:			
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.			
BY:		BY: Chief of Contract Administration Commercial			

CONVAIR: SD

Delta Air Lines Inc. Change No. 111

Page 1 of 5

Title: Deletion of Appendix I-C Items

Origin: Convair initiated.

Reason for Change: To delete inconsequential items from Appendix I-C.

Description of Change:

Delete the following items from the APPENDIX I-C Section:

Page Al, Power Plant Equipment

Under: Fuel System Equipment

Delete: 4 Fuel Tank Caps

Page A2, Electrical Equipment

Under: Interior Lights

Delete: AR Belly Cargo Dome Lights

2 Service Door Lights

1 Buffet Compartment Light

3 Lavatory Compartment Lights AR Ceiling and Aisle Lights

AR Passenger Reading Lights, L.H.

AR Passenger Reading Lights, R.H. AR Cockpit Lights

2 Fasten Seat Belts - No Smoking Sign

Return to Cappage Sign Lavatory Occupied Sign Compartment 1

AR Accessory Compartment Light

AR Coat Compartment Light

Add: AR Passenger Reading Lights

NOTE: Not to be part of Specification language:

> All interior lights have been reviewed by Convair Engineering and it has been determined that available off-the-shelf lights would not be suitable, in Convair's opinion, for the Model 22 airplane. Therefore all interior lights will be designed to Convair's specifications and bear a Convair part number. Convair has incorporated these items on the deletion list since interchangeability with other aircraft will not be achieved.

Delta Air Lines Inc. Change No. 111

Page 2 of 5

Page A3. Electrical Equipment

Delete the following items:

1 Utility Receptacle

1 Flap Take-Off Warning Horn Interrupter

1 Position Light Flasher 2 Flare Release Solenoids 6 Shaver Outlet Receptacles

1 Electrical Receptacle (7-pin connector) AiResearch (For Ground Starter Equipment)

Add the following item:

1 Electrical Receptacle (For Ground Bendix 10-107014-28 Starter Equipment)

Page A6, Electronic Equipment

Under: HF Communication Antenna

Convair P/N to be supplied Delete: 1 Antenna

Under: VHF Communication Antenna

37R-1 Delete: 1 Antenna (Flush Type) Collins

Under: VHF Navigational Antenna

Delete: 1 Omni-Directional Antenna Convair

Page A7. Instruments and Related Equipment

Delete the following items:

AR Anti-Icing Air Temperature Indicator AR Structural Temperature Overheat Indicator

4 Reverse Thrust Indicating Lights 4 Oil Pressure Low Indicator Lights

Page A9. Instruments and Related Equipment

Under: Kollsman Integrated Instrument System

Kollsman P/N to be supplied 1 Switch Computer Selector

Convair 1 Warning Light

Page 3 of 5

Page AlO, Hydraulic Equipment

Delete the following items:

AR Metering Check Valves

AR Check Valves

6 Spoiler Control Valves

1 Temperature Control Valve

2 Temperature Bulb 4 Hydraulic Fluid Quantity Transmitter

Reservoir Drain Valves

Pressure Reducers
Landing Gear Unlatching Cylinders

8 Brake Shuttle Valves

AR Selector Valves

Pressure Switches

2 Yaw Damper Cylinders

Yaw Damper Control Valve

Page Al2, Furnishings

Delete the following items:

Pilots' Seats

Flight Engineer's Seat

Stewardess Seats

1 Observer Seat

40 Passenger Seats (Double)

Lounge Seats (Double)

Wash Basins, Valves and Hardware

Toilet Tissue Dispensers

Soap Dispensers

Sanitary Napkin Dispensers

Linen Towel Dispensers

Cleansing Tissue Dispensers

Waste Containers

Ash Trays (Lavatory)

Potable Water Tank (50-Gal. capacity, pressurized)

Wash Basin Mirrors

Toilets, Disposal Tanks, and Hardware

Stewardess Switch Panel

Stewardess Call Chimes Electro Switch P/N to be supplied

AR Lavatory Assist Handles

Lavatory Coat Hooks (Flush Type)

Lavatory Signs "No Smoking - Return to Cabin"

Lavatory "Occupied" Signs

"No Smoking - Fasten Seat Belts" Sign

AR Buffet Water Filters

Delete the following note from bottom of page:

Delta Air Lines Inc. Change No. 111

Page 4 of 5

Page Al2, Furnishings (Cont)

Add the following items:

2	Pilots' Seats	Convair	22-91502
1	Flight Engineer's Seat	Convair	22-91500
3	Stewardess Seats	Convair	
1	Observer Seat	Convair	22-91501
40	Passenger Seats (Double)	Convair	
2	Lounge Seats (Double)	Convair	
1	Stewardess Switch Panel	Convair	
1	Potable Water Tank (50-	Convair	
	Gal. capacity, pres-		
	surized)		

Page Al3, Furnishings

Delete the following items:

l Airplane Check-Off List

1 Spare Lamp Box

2 Entrance Compartment Switch Panels

1 Airplane Log Book Holder

l Flight Manual (CAA Approved)

l Maintenace and Operating Manual

2 Pitot Heads

6 Static Pressure Ports

1 Static Pressure Port (Alternate)

6 Emergency Lights

2 Cabin Temperature Thermometers Taylor Instr. P/N to be (Mercury) assigned

Page Al5, Oxygen Equipment

Delete the following items:

*1 Build-up and Vent Valve

AR Cutlets

*1 Oxygen Converter

*1 Quantity Gage *1 Filler Valve

2 pr. Asbestos Gloves

1 Check Valve

l Line Valve

*NOTE: The following not to be a part of Specification language:

"*Items which have been deleted by Page Rev. 11-15-57."

CONVAIR: 5D

Delta Air Lines Inc. Change No. 111

Page 5 of 5

Page Aló, Pyrotechnics

Change the following item:

From: 2 Flare Containers 13.0 lb

To: 2 Flare Dispensers 13.0 1b Convair 22-98301

Page Al8, Fire Extinguishing Equipment

Delete the following equipment:

AR Double Check Tees

AR Directional Valves

Portable Water Bottles
Fire and Overheat Detector Panels

An Fire Detectors

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

SPEC NO: ZD-22-003		DATE:			
CUSTOMER: Delta Air Lines Inc.		MCLDtd			
CHANGE NO: 110		MODEL: (880) 22-2			
TITLE: Specification	n Administrative	Change			
ORIGIN: Convair ini		ntent of the Specification.			
	100				
	AFFECT ON WEIGH	HT * EFFECT ON BALANCE *			
Guar. Wt. Empty	Oper. Wt. Empt	y			
0	0	O Inch Lb.			
EFFECT ON GUARANTE	EFFECT ON GUARANTEED PERFORMANCE: *				
		None			
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal					
Acceptance of th Prior Acceptance	is Change is Depe of the Following	ndent Upon Changes:			
		Engineering Approval			
LATEST DATE OF ACC	EPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS	:	EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:			
ACCEPTED:		CONVAI , A Div. of Gen. Dyn. Corp.			

CONVAIR: SD

Title: Specification Administrative Change

Origin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Page 80. Paragraph 3.16.11.2 FIRE WARNING BELL:

In the second line, place a comma after "engines" and delete the last three words in the line "(or wheel wells,)"

Figure 3.19-9 SCHEMATIC - WHEEL WELL FIRE DETECTION:

Delete the above figure from the Specification.

Figure 3.19-10 MOD 22 FIRE EXTINGUISHING SYSTEM SCHEMATIC

Revise the above figure to delete pylon fire extinguishing from the diagram.

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

SPEC NO: ZD-22-003	DATE:
CUSTOMER: Delta Air Lines, Inc.	MCL 10,170 Dtd17 December 1957
CHANGE NO: 109A	MODEL: (880) 22-1
CAA Flight Data Recorder	
ORIGIN: Customer request.	
WASON FOR CHANGE: To comply with CAA	requirements. Ref.: CAR Amend- ment 40-6
PEFFECT ON WRIGHT	* EFFECT ON BALANCE *
Proposal No.2 +53.8 Oper. Wt. Empty +22.0 lbs +53.8 lbs	+16,946 Inch-Lbs. +48,110 Inch lb.
	oposal No. 1 None oposal No. 2 None
* Negligible Changes Will be Accumulat Totals Reflected in a Future Change	ed and Proposal
Acceptance of this Change is Depender Prior Acceptance of the Following Change	nt Upon anges:
	Engineering Approval
DATEST DATE OF ACCEPTANCE: AT	RPLAIRIS AFFECTED:
He No.	FICT ON PRICE PER ATRPLANE: curring: n-Recurring tal:
ACCEPTED: CO	NVAIR, A Div. of Gen. Dyn. Corp.
BY:	Chief of Contract Administration Commercial

CONVAIR: SD Delta Air Lines Inc. Change No. 109A

Title: CAA Flight Data Recorder

Origin: Customer request.

Reason for Change: To comply with CAA requirements. Ref.: CAR Amend-ment 40-6

PROPOSAL No. 1

Description of Change:

Page 62, Paragraph 3.14 INSTRUMENTS AND NAVIGATIONAL EQUIPMENT:

Add the following new paragraph:

"3.14.3. FLIGHT DATA RECORDING PROVISIONS: Mounting provisions consisting of a raised platform, support legs, sway braces, brackets, tubing and wiring shall be made in the hydraulic compartment for the later installation of a General Mills No. 601000C recorder and a No. 600207 amplifier. Aluminum tubing shall be installed from the co-pilot's static and pitot lines to the recorder location and shall be capped. Wiring shall include installation of one circuit breaker."

Effect on Weight Empty: +22.0 pounds

Effect on Balance: +16,946 inch-pounds

Effect on Performance: None

PROPOSAL No. 2

Page 62, Paragraph 3.14 INSTRUMENTS AND NAVIGATIONAL EQUIPMENT:

Add the following new paragraph:

"3.14.3. **SFLIGHT DATA RECORDER: A General Mills No. 601000C recorder and a No. 600207 amplifier shall be installed in the hydraulic compartment. Aluminum tubing shall be installed from the co-pilot's static and pitot lines to the recorder. Wiring shall include the installation of one circuit breaker."

Page A9, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Add the following to the equipment list:

Flight Data Recorder

l Recorder l Amplifier General Mills
General Mills

601000C 600207

Effect on Weight Empty: -53.8 pounds Effect on Balance: -48,110 inch-pounds

Effect on Performance: None

SPEC NO: ZD-22-003		DATE:
CUSTOMER: Delta Aiz	Lines Inc.	MCLDtd_
CHANGE NO: 108		MODEL: (880) 22-2
TITLE: Main Landi	ng Gear Wheels, Incre	ased Reight of Wheel Flange
ORIGIN: Convair in	itiated.	
REASON FOR CHANGE:	To provide main gear Tire and Rim Associa	wheels as recommended by the tion.
	EFFECT ON WEIGHT *	EFFECT ON BALANCE *
Guar. Wt. Empty	Oper. Wt. Empty	
+ 10.0 lb	+ 10.0 lb	+8,780 Inch Lb.
EFFECT ON GUARANTEE	D PERFORMANCE: * None	
Totals Reflected Acceptance of thi	es Will be Accumulate in a Future Change P is Change is Dependen of the Following Cha	roposal t Upon
LATEST DATE OF ACCE	EPTANCE: AIR	PLANES AFFECTED:
SPECIAL PROVISIONS	Rec Non	ECT ON PRICE PER AIRPLANE: urring: -Recurring al:
ACCEPTED:	CON	VAIR, A Div. of Gen. Dyn. Corp.
BY: DATE:		Chief of Contract Administration Commercial

Delta Air Lines Inc. Change No. 108

Title: Main Landing Gear Wheels, Increased Height of Wheel Flange.

Origin: Convair initiated.

Reason for Change: To provide main gear wheels as recommended by the Tire and Rim Association.

Description of Change:

Page All, APPENDIX I-C, LANDING GEAR EQUIPMENT:

Change the first item under "MAIN GEAR" as follows:

To: "8 Wheels @ 66.45 lb ea 531.6 lb"

Effect on Weight Empty: +10.0 pounds Effect on Balance: +8,780 inch-pounds Effect on Performance: None

SPEC NO: ZD-22-003	DATE:			
CUSTOMER: Delta Air Lines Inc.	MCLDtd			
CHANGE NO: 107	MODEL: (880) 22- 2			
TITLE: Specification Administrati	ve Change			
ORIGIN: Convair initiated.				
REASON FOR CHANGE: To clarify the	intent of the Specification.			
Can ok				
EFFECT ON WEIG				
Guar. Wt. Empty Oper. Wt. Empt	o Inch Lb.			
EFFECT ON GUARAUTEED PERFORMANCE: * None				
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal				
Acceptance of this Change is Depe Prior Acceptance of the Following	changes:			
	Engineering Approval			
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:	EFFECT ON PRICE PER ATRPLANE: Recurring: Non-Recurring Total:			
ACCEPTED:	CONVAIR, A Div. of Gen. Dyn. Corp.			
BY: DATE:	BY: Chief of Contract Administration Commercial			

Page 1 of 2

Title: Specification Administrative Change

Origin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Page 46, Paragraph 3.12.4.2.1 GENERATOR DRIVE OIL SYSTEM:

Change paragraph title to "CONSTANT SPEED DRIVE OIL SYSTEM"

In third line, delete the word "generator"

Revise the third sentence as follows:

"The constant speed drive oil system shall be independent of the engine oil system."

Add the following new sentence to the end of the paragraph:
"The constant speed drive oil system shall be designed for use with MIL-L-7808 oil."

- Page 46, Paragraph 3.12.4.2.2 GENERATOR DISCONNECTS:

 Change the paragraph title to "CONSTANT SPEED DRIVE"

 Delete the word "generator" in the first sentence.
- Page 46, Paragraph 3.12.4.2.3 GENERATOR MALFUNCTION DETECTOR:

 Change the paragraph title to "CONSTANT SPEED DRIVE MALFUNCTION DETECTOR"

 Delete the word "generator" in the first sentence.
- Page 47, Paragraph 3.12.5.1 AIR INLET:

 Change the paragraph title to "AIR INLET ANTI-ICING"
- Page 47, Paragraph 3.12.6 EXHAUST SYSTEM:

 Delete the paragraph and substitute the following:

 "A reverse thrust unit and sound suppressor shall be provided in place of an exhaust tailpipe."

CONVAIR: SD

Delta Air Lines Inc. Change No. 107

Page 2 of 2

Page 48, Paragraph 3.12.8.2 OIL LOW PRESSURE AND TEMPERATURE INDICATION WARNING:

Revise paragraph title to read: "OIL LOW PRESSURE AND TEMPERA-. TURE INDICATORS"

Page 55, Paragraph 3.12.9.13.3 REFUELING CONTROLS:

Revise the second sentence to read as follows:

"The panels shall contain fuel quantity gages, shut-off controls for refueling and defueling and jacks for service interphone system."

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

SPMC NO: ZD-22-00	3	DATE:
CUSTOMER: Delta Air	Lines Inc.	MCL 10,168 Rev. 17 January 19
CHANGE NO: 106A		MODEL: (880) 22- 2
TITIM: Cargo Comp	partment Web Gates,	Revision of
ONIGIN: Meeting of sentatives	25 November 1957 h	between Convair and TWA Repre-
REASON FOR CHANGE:	TWA request, Convai to CCP No. 106.	ir proposed for DAL, and revision
	FFFECT ON WEIGHT	* EFFECT ON BALANCE *
Guar. Wt. Empty	Oper. Wt. Empty	
+33.0 lbs	+33.0 lbs	+26,338 Inch Lb.
EFECT ON GUARANTE		Tone Rejected
* Negligible Chang Totals Reflected	es Will be Accumula in a Future Change	ted and Proposal
Acceptance of the Prior Acceptance	is Change is Depend of the Following C	ent Upon hanges:
		Engineering Approval
LATEST DATE OF ACC	EPTANCE: A	IRPLANES AFFECTED:
Special Provisions	7.	FFECT ON PRICE PER AIRPLANE: ecurring: on-Recurring otal:
ACCEPTAD:	C	ONVAIR, A Div. of Gen. Dyn. Corp.
107: (Dr. 188):		Y: Chief of Contract Administration Commercial

CONVAIR: SD

Title: Cargo Compartment Web Gates, Revision of

Origin: Meeting of 25 November 1957 between Convair and TWA Repre-

sentatives.

Reason for Change: TWA request; Convair proposed for DAL and Revision

to CCP No. 106.

Description of Change:

Page 108, Paragraph 3.19.2.5.1 WEB GATES:

Revise the entire paragraph to read as follows:

"Two transverse web gates with center opening sections shall be installed, one forward and one aft of the door in both the forward and aft cargo compartment."

Enclosure: (A) (ne (1) copy of Convair Drawing No. 22-09914, Sheet 1, dated 3 December 1957, revised 16 January 1958.

Effect on Weight Empty: +33.0 pounds

Effect on Balance: +26,338 inch-pounds

Effect on Performance: None

NOTE: The following not to appear in Specification language:

The space between the web gates will not be usable, and will result in a 53 cubic feet reduction of cargo area in the forward cargo compartment and 50 cubic feet reduction in the aft cargo compartment.

CONTERCTAL CHANGE PROPOSAL

SPEC NO: ZD-22-003	DATE:			
CUSTOMER: Delta Air Lines Inc.	MCI, Dtd			
CHANGE NO: 105				
TATER: Specification Administrati	ve Change			
ORIGIN: Convair initiated.				
REASON FOR CHANGE: To clarify the	intent of the Specification.			
EFFECT ON WEI				
Guar. Wt. Empty Oper. Wt. Emp	o Inch Ib.			
DEFECT ON GUARANTEED PERFORMANCE: * None				
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal				
Acceptance of this Change is Dep Prior Acceptance of the Followin	endent Upon g Changes:			
	Engineering Approval			
LATIST DATE OF ACCEPTANCE:	AIRPLANUS AFFECTED:			
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:			
ACCEPTED:	CONVAIR, A Div. of Gen. Dyn. Corp.			
BY: DATH:	BY: Chief of Contract Administration Commercial			

Delta Air Lines Inc. Change No. 105

CONVAIR: SD

Title: Specification Administrative Change

Origin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Page 108, Paragraph 3.19.2.5 LUGGAGE AND CARGO COMPARTMENTS:

Delete the second sentence and substitute the following:

"Flooring and forward bulkhead in each cargo compartment shall be of .0+5 aluminum alloy or equivalent."

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

EXHIBIT "A"

(Not a part of Specification language.)

The Specification language contained in the second paragraph of 3.7.1.6.5, with the requirement that permanent cargo flooring indentation shall not be greater than .050-inch, can be complied with and remains unchanged.

Enclosure: (A) One copy of Convair sketch, Figure 1 - FWD AND AFT CARGO COMPARTMENTS (for information only).

. CONVATA: SD

Title: Passenger Seat, Bottom Articulation of

Origin: Delta request during Delta-Convair meeting of 8 April 1958.

meason for Change: To provide seat bottom movement coincidental to seat back movement; and revision to CCP No. 104.

Description of Change:

Page 104, Paragraph 3.19.1.1.6 PASSENGER SEATS:

Add the following to the end of the paragraph:

"A bottom support pan for each bottom cushion, to be mounted independently of seat bottom structure, shall be provided. Cushion support pans shall be linked to seat backs in such manner that as seat backs are reclined, the seat bottoms move forward and inclination from the horizontal increases."

Liffect on Weight Empty: +205.0 pounds

ffect on Balance: +179,580 inch-pounds

affect on Performance: None

SPEC NO: ZD-22-00	}	DATE:			
CUSTOMER: Delta Air Lines Inc.		MCL_		Dtd	
CHANGE NO: 103		MODEL	(880	0) 22- 2	
THTLE: Specificati	on Administrative	Change			
ORIGIN: Convair in	tiated.	0 1			
REASON FOR CHANGE:	To clarify the	intent of	the Spe	ecifica	tion.
	EFFECT ON WEIGH	T *	EFI	FECT ON	BALANCE *
Guar. Wt. Empty	Oper. Wt. Empty				
0	0			0	Inch Lb.
EFFECT ON GUARANTEED PERFORMANCE: *					
	None				
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal					
Acceptance of this Change is Dependent Upon Prior Acceptance of the Following Changes:					
			Er	ngineeri	ing Approval
LATEST DATE OF ACCE	PTANCE:	AIRPLANES A	AFFECTI	ED:	
SPECIAL PROVISIONS:		EFFECT ON I Recurring: Non-Recurr Total:	ing		
ACCEPTED:		CONVAIR, A	Div.	of Gen.	Dyn. Corp.
BY:		BY:			
DATE:		Chief of		ract Adm mercial	ministration L

CONVAIR: SD

Title: Specification Administrative Change

Origin: Convair initiated.

Reason for Change: To clarify the intent of the Specification.

Description of Change:

Page 58, Paragraph 3.14.1.1 PILOT'S INSTRUMENTS:

In third line, change the following:

From: "Figure 3.14-3"

To: "Figure 3.14-1"

Paragraph 3.14.1.2 COPILOT'S INSTRUMENTS:

In third line, change the following:

From: "Figure 3.14-3"

To: "Figure 3.14-1"

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None

C () V A T & A Division of General Pynamics Corporation San Diego, California

COLDERCTAL CHANGE PROPOSAL

EPEC HO: ZD-22-003		DATE:		
CUSCOMER: Delta Air Lines Inc.		NCL 10,164	Dtd_26	November 1957
CHANGE NO: 102		MODEL: (880) 22-2		
Nose Land: from No. 2	ng Gear; Steering, B to No. 1 Hydraulic	raking and Ad System	ctuation,	Change
GRIGHM: Convair in	itiated.			
TOASON FOR CHANGE:	To retain steering failure in either o	capability in fine the hydrau	n event o	fa ms.
	EFFECT ON WEIGHT *	EF	FECT ON E	BALANCE *
Guar. Wt. Ampty	Oper. Wt. Empty			
-10.0 lbs	-10.0 lbs		+1,755	Inch Lb.
EFFECT ON GUARANTEE	PERFORMANCE: * None	V		
Accentance of this	Will be Accumulated in a Future Change Pr Change is Dependent of the Following Char	Upon ges:		an and a second
		E	ngineeri	ng Approval
LA EST DATE OF ACCEN	PTANCE: AIRE	LANES AFFECI	ED:	
SPECIAL PROVISIONS:	Reci Non-	CCT ON PRICE irring: Recurring		
ACCEPTED:	COM	MAIR, A Div.	of Gen.	Dyn. Corp.
B7:	BY:	Chief of Cont		inistration
DATE:	And the second s	Co	ommercial	

CONVAIR: SD

Delta Air Lines Inc. Change No. 102

Page 1 of 2

Nose Landing Gear; Steering, Braking and Actuation, Change from No. to No. 1 Hydraulic System Title:

Origin: Convair initiated.

Reason For Change: To retain steering capability in event of a failure

in either of the hydraulic systems.

Description of Change:

Page 58, Paragraph 3.14.1.2 COPILOT'S INSTRUMENT PANEL:

Add the following item to the instrument list:

"One hydraulic brake pressure indicator (2")"

Page 63, Paragraph 3.15.1 DESCRIPTION AND COMPONENTS:

Delete the second and third sentences and substitute the following:

"System No. 1 shall actuate the spoilers, horizontal stabilizer, flaps, nose landing gear, nose gear steering and nose gear brakes. System No. 2 shall actuate the spoilers, flaps, main landing gear and the main gear brakes. An electrically driven standby pump shall be provided which will supply hydraulic power to system No. l and No. 2."

Revise the fourth sentence to read as follows:

"Both systems shall normally be in continuous operation."

Page 64, Paragraph 3.15.1.11 BRAKE VALVE:

Revise the second sentence to read as follows:

"The brakes on the main wheels shall be operated by hydraulic system No. 2 and the nose wheel brakes shall be operated by hydraulic system No. 1."

Delta Air Lines Inc. Change No. 102

Page 2 of 2

Page Alo, APPENDIX I-C, HYDRAULIC EQUIPMENT:

Change the second item as follows:

"6 Accumulators, Piston Type" From:

"5 Accumulators, Piston Type" To:

Change the third item as follows:

"4 Gage, Accumulator Air". From:

"3 Gage, Accumulator Air" To:

Figure 3.8-1 HYD. LANDING GEAR WITH ACCUMULATORS
Figure 3.8-2 WHEEL BRAKE SCHEMATIC
Figure 3.14-1 PILOT AND CO-PILOT INSTRUMENT PANEL
Figure 3.15-1 HYDRAULIC SYSTEM

Revise above figures as required to reflect these changes.

Page A-10, APPENDIX I-C. HYDRAULIC EQUIPMENT:

Delete the following two items from the Description List:

"2 Yaw Damper Cylinders"

"1 Yaw Damper Control Valve"

-10.0 pounds Effect on Weight Empty:

+1,755 inch-pounds Effect on Balance:

None Effect on Performance:

SPEC NO: ZD-22-003		DA'TE:		
CUSTOMER: Delta Air	Lines Inc.	MCH. 10,160 Dtd 6 Nov. 1957		
CHANGE NO: 101		MODEL: (880) 22- 2		
		voir Gage, Addition of		
ORIGIN: September 1957; and Convair proposed for Delta, where applicable. REASON FOR CHANGE: To provide individual fluid quantity indication for each hydraulic system.				
	EFFECT ON WEI	IGHT * EFFECT ON BALANCE *		
Guar. Wt. Empty	Oper. Wt. Emp	oty		
+3.0 lbs	+8.0 lbs	+5,599 Inch Lb.		
DEFECT ON GUARANTE	ED PERFORMANCE:	*		
None				
* Negligible Changes Will be Accumulated and Totals Reflected in a Future Change Proposal				
Acceptance of the Prior Acceptance CCP No. 3A, Revis	is Change is Dep of the Followin sed 28 October 1	pendent Upon ng Changes:		
		Engineering Approval		
LATEST DATE OF ACC	SPTANCE:	AIRPLANES AFFECTED:		
SPECIAL PROVISIONS		EFFECT ON PRICE PER AIRPLANE: Recurring: Non-Recurring Total:		
ACCEPTED:		CONVAIR, A Div. of Gen. Dyn. Corp.		
BY: DATE:		BY: Chief of Contract Administration Commercial		

Delta Air Lines Inc. Change No. 101

Page 1 of 2

Title: Second hydraulic heserveir Cage, Addition of

Origin: TWA requested, Reference: TWA Letter No. 380-323, dated 20

September 1917; and Sonvair proposed

for Delta, where applicable.

meaton for Change: To provide individual fluid quantity indication for

each hydraulic system.

Description of Change:

age 55, Paragraph 3.14.1.4 FILIGHT ENGINEER'S INSTRUMENTS:

Add the following item to the instrument list:

"Che remote quantity gage, hydraulic reservoir (dual)"

Gragoson 3.14.1.5 MISCHLANBOUS INSTRUMENTS:

Delete the tests item from instrument list:

"(Two mydraulic fluid quantity indicators)"

Tage 54, Paragraph 3.15.1.7 RESERVOIRS:

Delete the fourth and fifth sentences from the paragraph and aubstitute the following:

"A qual, remote reading quantity gage shall be installed on the rlight engineer's panel to indicate the fluid level of each reservoir system. Each hydraulic system reservoir shall be provided with a transmitter, and a two-channel amplifier installed in the electrical compartment, to actuate the dual quantity gage."

and Alu, APA NDIX I-C, HYDRAULIC EQUIPMENT:

Change the 21st item in Description List as follows:

From: "I hydraulic Fluid Quantity Indicator"

"1 Hydraulic Fluid Quantity Indicator (Dual)"

Delta Ale Line: Inc. Change 1.0. 131

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e ald, Alladix I-C, Andraulic aquimint: (Cont)

Change the 24th item in Description List as follows:

rom: "1 Hydraulic Fluid Quantity Transmitter"

To: "2 Hydraulic Fluid Quantity Transmitter"

Mi ure 3.15-1 HYDRAULIC SYSTEM

herise above figure as required to reflect this change.

Affect on Weight Ampty: +5.0 pounds frect on Balance:

+5,599 inch-pounds

Effect on Performance: Hone